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**A COMPUTER CODE FOR FULLY-COUPLED ROCKET NOZZLE FLOWS
(FULLNOZ)**

H. S. Pergament, et al

AeroChem Research Laboratories, Incorporated

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Princeton, New Jersey 08540**

April 1975

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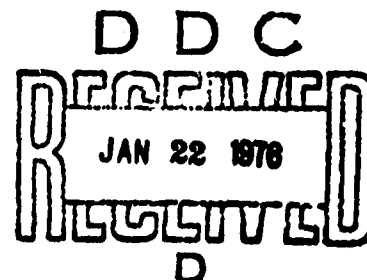
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21. ABSTRACT (Continue on reverse side if necessary and identify by block number) A comprehensive computer code (FULLNOZ) has been developed to perform detailed calculations of rocket nozzle flows downstream of the sonic line. The code uses the streamtube method to integrate the hyperbolic governing equations of steady supersonic flow. The program represents a significant advance in nozzle flow predictions through the coupling of gas/particle nonequilibrium effects, non-equilibrium chemistry, turbulent boundary layers (to determine the effects of wall heat transfer and shear stress) and turbulent mixing across streamtubes.		

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This report describes the analytical and numerical techniques employed by the code, presents results of a sample calculation for the Minuteman Stage 2 nozzle and gives complete instructions on the preparation of input data and a full FORTRAN listing.

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SUMMARY

This report describes the analytical and numerical techniques utilized in the development of a fully-coupled rocket nozzle flow computer code (FULLNOZ). The code uses the streamtube method to integrate the governing equations of steady supersonic flow. The elliptic Navier-Stokes equations are reduced to hyperbolic form by assuming diffusional effects along streamlines are small compared to those across streamlines. Finite difference techniques are then used to solve the hyperbolic equations along and perpendicular to streamlines.

FULLNOZ represents a significant advance in nozzle flow calculations by coupling the effects of nonequilibrium chemistry, gas/particle thermal and dynamic nonequilibrium, turbulent mixing across streamtubes and turbulent boundary layers. Turbulent mixing is treated via a phenomenological eddy viscosity model, while the turbulent boundary layer analysis utilizes the experimental data of Keener and Hopkins (which relates the compressible skin friction coefficient to measured velocity/temperature profiles in flows with favorable pressure gradients), the Van Driest transformations and the momentum integral equation. The operation of FULLNOZ requires the specification of initial gas and particle properties just downstream of the sonic line, the nozzle wall contour and temperature, and a chemical reaction mechanism and rate coefficients. The marching scheme proceeds downstream computing flow properties and composition along surfaces orthogonal to a specified number of streamtubes. A mixed explicit/implicit differencing scheme is used to obtain the most favorable integration step size.

Sample calculations are presented for the Minuteman Stage 2 nozzle. In addition, this report describes the preparation of input data and gives a full FORTRAN listing of the program. Also included is an analysis of heterogeneous electron/ion recombination on particles, although it has not as yet been incorporated into the code.

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PREFACE

This is a report on the work performed on Contract F44620-74-C-0006 covering the period 1 September 1973 to 30 April 1975. It is basically a program user's manual for FULLNOZ, although some additional results obtained during this period, e.g. a preliminary study of how to incorporate heterogeneous electron-ion recombination into the code, have also been included.

The authors would like to thank J. T. Kelly for his initial work in the development of FULLNOZ and Capt. L. R. Lawrence, AFOSR Program Manager, for his interest, encouragement and support during the development of FULLNOZ.

This code is available for public use and may be obtained by forwarding a request and a tape to Lt. Robert Sperlein, DYSP, at the Air Force Rocket Propulsion Laboratory, Edwards Air Force Base, CA 93523.

This scientific report has been reviewed and is approved.

LIST OF SYMBOLS

A	streamtube area; also pre-exponential term in rate coefficient equations
$a_{i,jj}$	enthalpy-temperature polynomial coefficients, see Eq. (20)
B	activation energy
C_D	particle drag coefficient
\bar{C}_D	normalized drag coefficient, see Eq. (34)
C_{DFM}	free molecular drag coefficient
C_{DI}	high Reynolds number drag coefficient, see Eq. (35)
C_i	mass fraction of i th species
C_F	skin friction coefficient, see Eqs. (54) and (59)
C_p	gas specific heat
C_s	particle specific heat
D	diffusion flux, defined by Eq. (23)
\bar{D}	diffusion coefficient
F_i	defined as C_i/M_i
f_p	ratio of actual particle drag coefficient to the drag coefficient for Stokes flow
g_p	ratio of actual particle heat transfer coefficient to heat transfer coefficient for Stokes flow
H	gas stagnation enthalpy
H_{12}	shape factor, defined by Eq. (69)
H_f^0	heat of formation at 298°K
h	gas static enthalpy
h'	particle/gas heat transfer coefficient
I	total number of gas species
J	total number of particle groups
K_p	equilibrium constant
k	thermal (eddy) conductivity
k_f	forward reaction rate coefficient
k_g	molecular thermal conductivity in particle/gas interaction terms

L	total number of species on left or right side of reaction
L₁-L₃	specific heat polynomial coefficients
L₆	enthalpy constant of integration, see Eq. (31)
L₇	entropy constant of integration, see Eq. (32)
Le	turbulent Lewis number, $Le = \frac{C \bar{D}_p}{\frac{p}{k}}$
M	Mach number
M_i	molecular weight of ith species
m	streamtube mass flow, defined by Eq. (13)
N	number of reactions
Nu	Nusselt number, $Nu = \frac{h' r_p}{k_g}$
n	distance normal to streamline, also exponent in Eqs. (57) and (58)
Pr	turbulent Prandtl number, $Pr = \frac{\mu C}{k} \frac{p}{k}$
Pr_g	laminar Prandtl number
p	static pressure
q	heat flux, defined by Eq. (22)
q_w	wall heat transfer, defined by Eq. (62)
R	universal gas constant, also value of nozzle radius at each x
Re	Reynolds number based on streamtube width
Re_p	particle Reynolds number
Re_x	Reynolds number based on distance along nozzle wall
Re_θ	Reynolds number based on boundary layer momentum thickness
r	radial distance from axis; also recovery factor, Eq. (49)
r_p	particle radius
s	distance along streamline
S	entropy
S_t	Stanton number, defined by Eq. (62)
T	static temperature
T_r	recovery temperature

u	velocity along streamline
U	velocity in boundary layer
U_T	friction velocity, defined by Eq. (63)
V_p	particle velocity normal to gas streamline
\dot{W}_i	production rate of i th species
x	axial distance from nozzle starting line
y	distance from nozzle wall
Y	molar concentration

Greek Letters

α	numerical stability coefficient for marching scheme, see Eq. (29)
γ	ratio of specific heats
ΔG	Gibbs free energy
$\Delta_k(\Phi)$	change in Φ across streamtube
δ	boundary layer thickness
δ^*	displacement thickness
δn	finite difference mesh spacing in n direction
δs	finite difference mesh spacing in s direction
Θ	angle between streamline and plume axis; also boundary layer momentum thickness
$\Theta_s, \Theta_s', \Theta_s''$	respectively; final, initial and intermediate streamline radii of curvature
Φ	$T/1000$, used in Eq. (30)
ν	kinematic viscosity
ν'	stoichiometric coefficient on left hand side of reaction
ν''	stoichiometric coefficient on right hand side of reaction
μ	eddy viscosity
μ_g	molecular viscosity in particle/gas interaction terms
ρ	gas density
ρ_p	particle cloud density
ρ_s	density of liquid or solid particle
τ	shear stress, defined by Eq. (21) and Eq. (60)
\cdot	multiplication

Σ summation

Subscripts

aw	adiabatic wall
c	compressible
e	wall streamtube
FM	free molecular
g	gas
i	ith species; also, incompressible
j	particle group identification index
k	streamtube index
l	orthogonal surface index; also refers to species on left or right hand side of reaction in Eq. (6)
m	mth reaction
n	differentiation in direction normal to streamlines
o	stagnation value
p	particle
s	differentiation in streamline direction
t	total
w	wall

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I. INTRODUCTION

This report describes a new fully-coupled rocket nozzle code (FULLNOZ) which treats, simultaneously, gas/particle nonequilibrium, nonequilibrium chemistry, diffusion across streamlines, and turbulent boundary layers within axisymmetric and two-dimensional nozzles. Because of its fully-coupled capability FULLNOZ can be applied to a wider range of problems than such existing rocket nozzle codes,[†] as TDK¹ (which does not account for gas/particle nonequilibrium effects), the nozzle portion of the CONTAM² code, the original constant γ gas/particle nonequilibrium code of Nickerson and Kliegel³ and the recently-developed gas/particle and chemical nonequilibrium code developed at Lockheed/Huntsville.⁴ The emphasis in the present code has been on an accurate calculation of nozzle exit plane gas and particle properties (particularly major and minor neutral and charged species concentrations), rather than on the determination of specific impulse, although FULLNOZ is well equipped to calculate I_{sp} .

[†] e.g. none of these codes calculate turbulent boundary layers.

1. "ICRPG Two-Dimensional Kinetic (TDK) Nozzle Analysis Computer Program," Dynamic Science Corp., December 1973 (revised version).
2. Hoffman, R.J., English, W.D., Oeding, R.G., and Webber, W.T., "Plume Contamination Effects Prediction: The CONTAM Computer Program," Final Report, Air Force Rocket Propulsion Laboratory, AFRPL-TR-71-109, December 1971.
3. Nickerson, G.R. and Kliegel, J.R., "Axisymmetric Two-Phase Perfect Gas Performance Program," TRW Systems Report No. 02874-6006-R000, Vols. I and II, April 1967.
4. Penny, M.M. and Smith, S.D., "Supersonic Gas-Particle Flows, Including Reacting Chemistry," JANNAF 8th Plume Technology Meeting, Colorado Springs, CO, July 1974.

FULLNOZ is based on the **MULTITUBE** code developed by Boynton,⁵ which incorporates the streamtube method† (described by Boynton and Thomson⁶) to integrate the hyperbolic governing equations of steady supersonic flow. The major routines incorporated into **FULLNOZ** which are not in **MULTITUBE** include (1) particle/gas nonequilibrium, (2) nonequilibrium chemistry and (3) turbulent wall boundary layers. Briefly, in the streamtube method the elliptic Navier-Stokes equations‡ are reduced to hyperbolic form by assuming that diffusional effects along streamlines are small compared to diffusion across streamlines. This assumption is very good for rocket nozzle (and plume) flows and enables one to solve an initial value problem (where a marching procedure can be used) rather than the more difficult boundary value problem. The gas flow equations, in finite-difference form, are solved along and perpendicular to streamlines while a full continuum particle cloud system of equations is incorporated for the condensed phase.

The advantages of using the streamtube method over the method of characteristics in calculating rocket nozzle (and plume) flows are:

- Species diffusion, shear, and heat transfer normal to streamlines are easily included.
- Chemical reactions or internal relaxations are easily incorporated since the calculation follows streamlines.
- Bounding surfaces and gradients normal to streamlines are treated without difficulty; and
- A wide variety of boundary conditions including mass transfer, shear and heat transfer can readily be incorporated.

† In contrast to the abovementioned codes,¹⁻⁴ which all use the method of characteristics.

‡ In using the Navier-Stokes equations as a base the technique can readily be extended, with the proper boundary conditions, to low density nozzle flows.

5. Boynton, F.P., "The **MULTITUBE** Supersonic Flow Computer Code," General Dynamics/Convair GDC-DBB 67-003, February 1967.
6. Boynton, F.P. and Thomson, A., "Numerical Computation of Steady, Supersonic, Two-Dimensional Gas Flow in Natural Coordinates," J. Computational Phys. 3, 379-398 (1969).

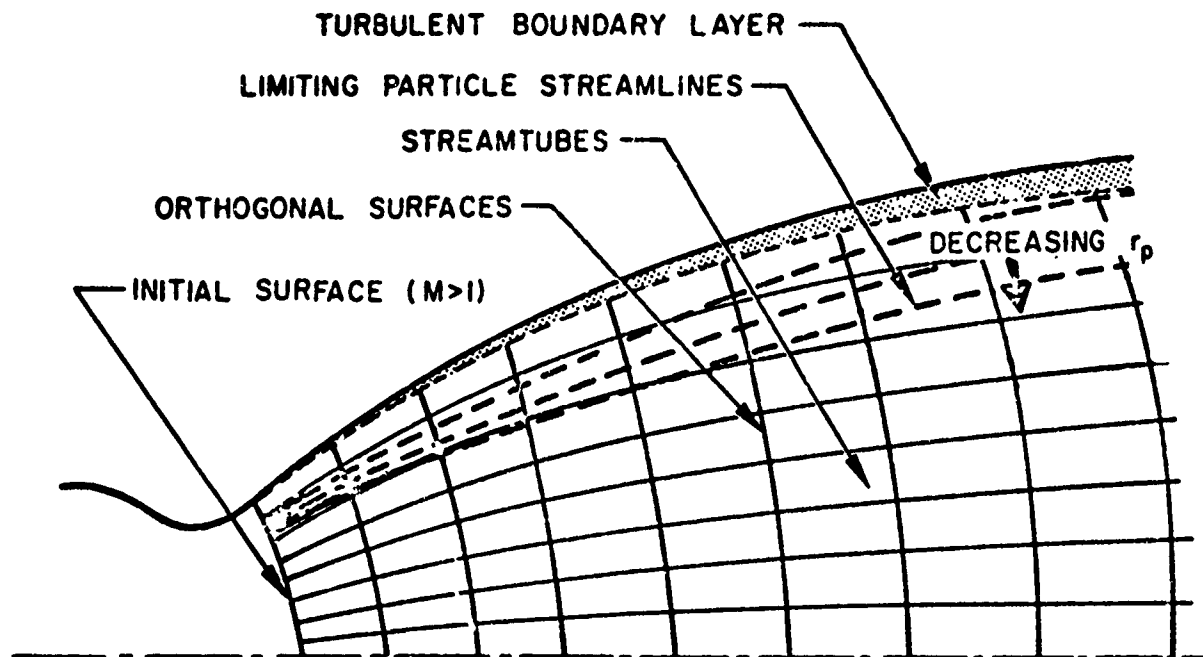


Figure 1. FULLNOZ schematic.

Operation of the code is achieved by specifying (1) initial gas and particle properties in the supersonic region just downstream of the nozzle throat,[†] (2) the nozzle wall contour, (3) a chemical reaction mechanism and rate coefficients, (4) physical properties of the particles, and (5) the nozzle wall temperature (for the boundary layer calculations). With the above input data the marching scheme steps from one orthogonal surface to the next (Fig. 1) computing gas and particle properties within a specified number of streamtubes. The (essentially) continuous particle size distribution is represented by up to a maximum of 8 discrete particle sizes. All particles cross the gas streamlines, but the lighter particles follow the gas streamlines more closely than the heavy particles. As depicted in Fig. 1, limiting particle streamlines are computed for each particle size. At each orthogonal surface (i.e. each integration

[†] It would be useful to incorporate an "initializing" scheme which utilizes combustion chamber properties as initial conditions, computes flow properties through the transonic region and establishes an initial supersonic data line. Such a scheme, developed by Nickerson and Kliegel,³ is contained in the codes of Refs. 1-3, and with additional programming could be inserted into FULLNOZ.

step) the code calculates wall shear stress and heat transfer, boundary layer displacement thickness and velocity and temperature profiles. The shear stress and heat transfer are coupled to the main nozzle flow via their effect on the wall streamtube properties.[†] The turbulent boundary layer analysis utilizes the Van Driest⁷ transformations and the experimental data of Keener and Hopkins,⁸ which relates the compressible skin friction coefficient to measured velocity/temperature profiles in flows with favorable pressure gradients (see Section II.C.4). The boundary layer momentum thickness is computed via the momentum integral equation, using the wall streamtube pressure, velocity, etc., as the boundary layer 'edge' conditions. The displacement thickness is then determined from the velocity profile and momentum thickness.

The code will not handle shocks that might originate from the nozzle wall in turning the flow. The flow in the region of the shock will be treated as a strong compression wave.[‡]

[†] This implicitly assumes that the boundary layer momentum and energy thicknesses are smaller than the wall streamtube thickness. This will generally be true for high Reynolds number nozzle flows. Provision is in the program, however, to transfer momentum, heat and mass across streamtubes, so that if an appropriate transfer coefficient can be defined the boundary layer effects can be "felt" throughout the flow.

[‡] Nozzle shocks could be incorporated into FULLNOZ in a manner similar to that employed in the AIPP code^{9,10} to detect internal plume shocks, but this would require additional programming.

7. Van Driest, E.R., "Turbulent Boundary Layer in Compressible Fluids," J. Aeron. Sci. 18, 145-160 (1951).
8. Keener, E.R. and Hopkins, E.J., "Van Driest Generalization Applied to Turbulent Skin Friction and Velocity Profiles Measured on the Wall of a Mach 7.4 Wind Tunnel," AIAA J. 11, 1784-1785 (1973).
9. Pergament, H.S. and Kelly, J.T., "A Fully-Coupled Underexpanded Rocket Plume Program. (The AIPP Code). Part I. Analytical and Numerical Techniques," AeroChem TP-302a, AFPR-TR-74-59, November 1974.
10. Pergament, H.S. and Kelly, J.T., "A Fully-Coupled Underexpanded Rocket Plume Program (The AIPP Code). Part II. Program User's Manual," AeroChem TP- (in preparation).

The code was written for operation on a CDC 6400 computer and requires approximately 122 K (octal) of core storage. For operation on machines with larger core storage the user may want to increase the dimension of some parameters, such as total number of species (25 maximum) and total number of reactions (40 maximum).

This report serves primarily as a program user's manual,[†] but also contains all the governing equations (Section II) and the results of some preliminary parametric calculations for the Minuteman, second stage nozzle (Section IV.A). The user should be cautioned that, although the code has been formally debugged for several test cases, extensive calculations have not (as of April 1975) been made; therefore some operational problems may be experienced. If these do occur, please contact the authors.

II. GOVERNING EQUATIONS

This section gives the governing differential and finite-difference equations used in the code. Also included are the auxiliary equations for calculating thermodynamic and chemical kinetic properties of the system, gas/particle drag and heat transfer coefficients and turbulent boundary layer properties.

A. Differential Equations

1. Gas Phase

For most high Reynolds number nozzle flows of interest the turbulent transport of mass, momentum and energy throughout the main nozzle flow will be negligible. Thus in the equations that follow all transport terms will be identically zero. In practice this is achieved by setting ITURB (Card 4, Cols. 56-60) equal to zero. If turbulent diffusion across streamtubes is to be included due either to initial non-uniformities or to the propagation of boundary layer effects across the flow, appropriate (constant) values of the eddy transport terms, μ , Pr and Le will have to be input on Card 10. If a suitable expression for eddy viscosity in terms of local properties can be developed for transport across streamtubes within nozzles, this expression can easily be added to the program in subroutine TRANSP.

[†] The same numerical integration techniques and similar program logic is incorporated into the AIFP code,^{9,10} to which the reader is referred for additional information on program subroutines.

Global Continuity

$$(\rho u)_s + \rho u \frac{\sin \theta}{r} + \rho u \theta_n = 0 \quad (1)$$

s-Momentum

$$\rho u u_s + p_s = \frac{1}{r} \left[r \mu u_n \right]_n - \sum_{j=1}^J \frac{g}{2} \frac{\rho_{p_j}^f \rho_{p_j}^\mu}{\rho_s r p_j} (u - u_{p_j}) \quad (2)$$

n-Momentum

$$\rho u^2 \theta_s + p_n = \sum_{j=1}^J \frac{g}{2} \frac{\mu_g \rho_{p_j}^f \rho_{p_j}^\mu}{\rho_s r p_j} v_{p_j} \quad (3)$$

Species Continuity[†]

$$\rho n(C_i)_s = \rho \dot{W}_i + \frac{1}{r} \left[r \mu \frac{Le}{Pr} (C_i)_n \right]_n \quad (4)$$

Energy

$$\begin{aligned} \rho u H_s = & \frac{1}{r} \left[r \frac{\mu}{Pr} H_n \right]_n + \frac{1}{r} \left[\left(1 - \frac{1}{Pr} \right) \mu \left(\frac{u^2}{2} \right)_n r \right]_n \\ & + \frac{1}{r} \left[\sum_{i=1}^I \left\{ \frac{\mu}{Pr} (Le - 1) r h_i (C_i)_n \right\} \right]_n \\ & + \sum_{j=1}^J \frac{g}{2} \frac{\mu_g \rho_{p_j}^f \rho_{p_j}^\mu}{\rho_s r p_j} \left\{ v_{p_j}^2 + (u - u_{p_j})^2 \right. \\ & \quad \left. + \frac{2}{3} \frac{g_{p_j} C_p}{f_{p_j} Pr_g} (T_{p_j} - T) \right\} \end{aligned} \quad (5)$$

[†]In this formulation, \dot{W}_i is expressed in units of sec^{-1} .

The drag (f_{pj}) and heat transfer (g_{pj}) factors are defined as, $f_{pj} = C_D Re_p / 24$ and $g_{pj} = (Nu/2) (T_p - T_r) / (T_p - T)$, where T_r is the recovery temperature based on the relative velocity between the gas and particle.

The following auxiliary expressions are required:

Species Production

$$\dot{W}_i = \frac{M_i}{\rho} \sum_{m=1}^N (v''_{i,m} - v'_{i,m}) \left[k_{fm} \prod_{l=1}^L Y_l^{v'_{l,m}} - \frac{k_{fm}}{K_p} \prod_{l=1}^L Y_l^{v''_{l,m}} \right] \quad (6)$$

Equation of State

$$\rho = \frac{P}{RT \sum_i \frac{C_i}{M_i}} \quad (7)$$

Stagnation Enthalpy

$$H = \frac{u^2}{2} + h(T) \quad (8)$$

2. Particles

For the condensed phases present within the flow a continuum particle cloud assumption is made and therefore field conservation equations for continuity, momentum and energy can be written for the particles. The (essentially) continuous distribution of particle sizes at each point in the flow is modeled by several groups of constant size particles representative of the distribution. For a given group, j , the conservation equations, written in a streamline oriented coordinate system are:

Continuity

$$(r \rho_{p_j} u_{p_j})_s + (r \rho_{p_j} v_{p_j})_n + r \rho_{p_j} u_{p_j} \Theta_n - r \rho_{p_j} v_{p_j} \Theta_s = 0 \quad (9)$$

s-Momentum

$$u_{p_j} (u_{p_j})_s + v_{p_j} (u_{p_j})_n - v_{p_j} (v_{p_j} \Theta_n - u_{p_j} \Theta_s) =$$

$$- \frac{9}{2} \frac{\mu_g^f p_j}{\rho_s r_{p_j}^2} (u_{p_j} - u) \quad (10)$$

n-Momentum

$$u_{p_j} (v_{p_j})_s + v_{p_j} (v_{p_j})_n + u_{p_j} (v_{p_j} \Theta_n + u_{p_j} \Theta_s) =$$

$$- \frac{9}{2} \frac{\mu_g^f p_j}{\rho_s r_{p_j}^2} v_{p_j} \quad (11)$$

Energy[†]

$$u_{p_j} (C_s T_{p_j})_s + v_{p_j} (C_s T_{p_j})_n = - \frac{3 k_g g_{p_j}}{r_{p_j}^2 \rho_s} (T_{p_j} - T) \quad (12)$$

[†]The effects of chemical reactions on the surface of particles are not included in this analysis.

When the particle undergoes a phase change (liquid to solid) it is kept at the solidification temperature until the total heat of solidification is released (via radiative and convective heat transfer) to the gas.

B. Finite-Difference Equations

A finite difference formulation of the gas phase and particle cloud governing equations is utilized on a grid which lies along and perpendicular to the streamlines.

1. Gas-Phase

The momentum and energy equations (Eqs. (2), (3) and (5)) are solved via an explicit finite-difference marching technique, whereas the species continuity equation (Eq. (4)), utilizes an implicit finite difference formulation developed in an earlier study at AeroChem.¹¹ This mixed form of the difference equations is necessary for the economical operation of the present code since, for near-equilibrium chemistry, the explicit finite-difference form of Eq. (4) leads to stability-limited (impractically small) integration step sizes. Equations (2), (3) and (5) are left in explicit form since the required integration step sizes for stability are reasonable; an implicit formulation of these equations would unnecessarily complicate the calculations.

To minimize the effects of large tube-to-tube property variations on the calculation of streamline curvature, a VonMises-type transformation is employed, i.e.

$$\dot{m} = 2\pi \int \rho u r \, dn \quad (13)$$

Using Eq. (13) the finite difference forms of the differential equations become:

Global Continuity

$$\dot{m}_k = \rho_{k,l+1} u_{k,l+1} A_{k,l+1} \quad (14)$$

11. Mikatarian, R.R., Kau, C.J., and Pergament, H.S., "A Fast Computer Program for Nonequilibrium Rocket Plume Predictions," Final Report, AeroChem TP-282, AFRPL-TR-72-94, August 1972.

s-Momentum

$$\begin{aligned} \dot{m}_k (u_{k,l+1} - u_{k,l}) + \frac{1}{2} (A_{k,l+1} + A_{k,l}) (p_{k,l+1} - p_{k,l}) = \\ \Delta_k (r_k \tau_k \delta s_k) 2\pi - \frac{1}{2} (A_{k,l+1} + A_{k,l}) \\ \times \sum_{j=1}^J \frac{9}{2} \left(\frac{\rho_{p_j} f_{p_j} \mu_g}{\rho_s r_{p_j}^2} \right)_{k,l} (u_{k,l} - u_{p_j,k,l}) \end{aligned} \quad (15)$$

n-Momentum

$$\begin{aligned} \left(\frac{\partial \theta}{\partial s} \right)_{k,l} = - \frac{8\pi r_{k,l}}{u_{k,l} + u_{k+1,l}} \left(\frac{p_{k+1,l} - p_{k,l}}{\dot{m}_k + \dot{m}_{k+1}} \right) \\ + \frac{2}{((\rho u^2)_{k+1,l} + (\rho u^2)_{k,l})} \\ \times \sum_{j=1}^J \frac{9}{2} \left(\frac{\rho_{p_j} f_{p_j} \mu_g}{\rho_s r_{p_j}^2} \right)_{k,l} v_{p_j,k,l} \end{aligned} \quad (16)$$

Species Continuity[†]

$$\dot{m}_k (C_{i,k,l+1} - C_{i,k,l}) = \Delta_k (r_k D_{i,k} \delta s_k) 2\pi + \frac{2 \dot{m}_k \delta s_k}{(u_{k,l+1} + u_{k,l})} \dot{W}_{i,k} \quad (17)$$

[†] The species mass fractions at station $k, l + 1$ ($C_{i,k,l+1}$) are determined by linearizing the chemistry terms ($(\dot{W}_i)_k$) and inverting the resulting matrix (see Ref. 11).

Energy

$$\begin{aligned}
 \dot{m}_k \left(h_{k,l+1} - h_{k,l} + \frac{1}{2} u_{k,l+1}^2 - \frac{1}{2} u_{k,l}^2 \right) = \\
 \Delta_k \left(r_k \left[q_k + \left(\frac{1}{2} (u_k^2 + u_{k+1}^2) \right)^{1/2} \tau_k + \sum_{i=1}^I h_{i,k} D_{i,k} \right] \delta s_k \right) 2\pi \\
 + \frac{1}{2} (A_{k,l+1} + A_{k,l}) \sum_{j=1}^J \left(\frac{9}{2} \left(\frac{\rho_{p_j}^f \rho_{p_j}^g}{\rho_s r_{p_j}^2} \right)_{k,l} \left[v_{p_j,k,l}^2 \right. \right. \\
 \left. \left. + (u_{k,l}^2 - u_{p_j,k,l}^2) + \frac{2}{3} \left(\frac{g_{p_j} C_{p_g}}{f_{p_j} Pr} \right)_{k,l} (T_{p_j,k,l} - T_{k,l}) \right] \right) \quad (18)
 \end{aligned}$$

State

$$\rho_{k,l} = \frac{p_{k,l}}{R T_{k,l} \sum_i \left(\frac{C_i}{M_i} \right)_{k,l}} \quad (19)$$

Enthalpy

$$H = \frac{u_{k,l}^2}{2} + \sum_i \sum_{jj} a_{i,jj} T_{k,l}^{jj-1} C_{i,k,l} \quad (20)$$

where in Eqs. (15), (17) and (18),

$$\tau_k = -2\pi r_k^2 (u_{k+1,l}^2 - u_{k,l}^2) \frac{(\rho_{k,l} \mu_{k,l} + \rho_{k+1,l} \mu_{k+1,l})}{(\dot{m}_{k+1} + \dot{m}_k)} \quad (21)$$

and

$$q_k = -2\pi r_k^2 (u_{k+1,l} + u_{k,l}) \times \frac{(\rho_{k,l} k_{k,l} + \rho_{k+1,l} k_{k+1,l})(T_{k+1,l} - T_{k,l})}{(\dot{m}_{k+1} + \dot{m}_k)} \quad (22)$$

and

$$D_{i,k} = - \frac{2\pi r_k^2 (u_{k+1,l} + u_{k,l}) (\rho_{k+1,l}^2 \bar{D}_{k+1,l} + \rho_{k,l}^2 \bar{D}_{k,l}) (C_{i,k+1,l} - C_{i,k,l})}{(\dot{m}_{k+1} + \dot{m}_k)} \quad (23)$$

2. Particles

The finite difference forms of Eqs. (9-12) are:

Continuity

$$\begin{aligned} & (r\rho_{p_j} u_{p_j})_{k,l+1} - (r\rho_{p_j} u_{p_j})_{k,l} + ((r\rho_{p_j} V_{p_j})_{k+1,l} - (r\rho_{p_j} V_{p_j})_{k,l}) \frac{\delta s_k}{\delta n_k} \\ & + \delta s_k (r\rho_{p_j} u_{p_j})_{k,l} \Theta_{n_{k,l}} - \delta s_k (r\rho_{p_j} V_{p_j})_{k,l} \Theta_{s_{k,l}} = 0 \end{aligned} \quad (24)$$

s-Momentum

$$\begin{aligned}
 & u_{p_{jk,l}} (u_{p_{jk,l+1}} - u_{p_{jk,l}}) + v_{p_{jk,l}} (u_{p_{jk+1,l}} - u_{p_{jk,l}}) \frac{\delta s_k}{\delta n_k} \\
 & - \delta s_k v_{p_{jk,l}} (v_{p_j \Theta_n} - u_{p_j \Theta_s})_{k,l} = \\
 & - \frac{9}{2} \left(\frac{f_{p_j} \mu_g}{\rho_s r_{p_j}^2} \right)_{k,l} (u_{p_{jk,l}} - u_{k,l}) \delta s_k
 \end{aligned} \tag{25}$$

n-Momentum

$$\begin{aligned}
 & u_{p_{jk,l}} (v_{p_{jk,l+1}} - v_{p_{jk,l}}) + v_{p_{jk,l}} (v_{p_{jk+1,l}} - v_{p_{jk,l}}) \frac{\delta s_k}{\delta n_k} \\
 & + u_{p_{jk,l}} (v_{p_j \Theta_n} + u_{p_j \Theta_s})_{k,l} \delta s_k = \\
 & - \frac{9}{2} \left(\frac{\mu_g f_{p_j}}{\rho_s r_{p_j}^2} \right)_{k,l} v_{p_{jk,l}} \delta s_k
 \end{aligned} \tag{26}$$

Energy

$$\begin{aligned}
 & u_{p_{jk,l}} C_s (T_{p_{jk,l+1}} - T_{p_{jk,l}}) + v_{p_{jk,l}} C_s (T_{p_{jk+1,l}} - T_{p_{jk,l}}) \frac{\delta s_k}{\delta n_k} = \\
 & - \frac{3\epsilon\sigma}{r_{p_{jk,l}} \rho_s} T_{p_{jk,l}}^4 \delta s_k - 3 \left(\frac{k_g g_{p_j}}{r_{p_j}^2 \rho_s} \right)_{k,l} \\
 & \times (T_{p_{jk,l}} - T_{k,l}) \delta s_k
 \end{aligned} \tag{27}$$

3. Integration Step Size

The integration step size must be limited in order to perform a stable calculation. The stability of the explicit finite difference scheme for solving the gas dynamic equations is discussed by Boynton and Thomson,⁶ who show that the stable step size for laminar flow is determined from,

$$\delta s \leq \frac{\delta n}{2} \left[\frac{1}{Re} + \frac{1}{(M^2 - 1)^{1/2}} \right]^{-1} \quad (28)$$

For turbulent flow the eddy viscosity replaces the laminar viscosity in the expression for Re .[†] The step size determined from Eq. (28) may not be sufficiently small if the chemistry is very "fast". Thus, an additional "chemistry" control has been incorporated into the code: at each integration step (orthogonal surface) the species mass fractions are tested for sign. If any C_i goes negative the integration step size is halved and the calculations are repeated until the C_i in question becomes positive or the step size becomes less than the minimum allowable step size. In the latter case the program terminates.

The particle cloud system of equations (Eqs. (9-12)) contains derivatives related only to the convection of mass, momentum and energy. In the momentum and energy equations, the convection terms are equal to the particle/gas interaction terms. Unlike the gas flow equations the particle equations have no wave or diffusive nature. Consequently when these equations are solved explicitly there is no stability limitation on the integration step size and the step size can be determined from Eq. (28).

4. Solution Procedure

The gas phase and particle cloud finite difference equations are solved on a grid consisting of the streamtubes and the surfaces orthogonal to them, as illustrated in Fig. 1. All gas and particle flow properties, streamline positions and angles must be known along the initial orthogonal surface downstream of the throat ($M > 1$). Starting with the first streamtube (adjacent to the axis) the streamtubes are extended a distance δs to a downstream surface, utilizing the radius of curvature, $(\theta_s)'$ obtained from the normal momentum equation (Eq. (16)) and the known initial pressure distribution along the

[†] Eq. (28) is not, of course, valid in the limit, $Re \rightarrow 0$. In that case (i.e. the situation for most practical nozzle flows), $\delta s \leq \frac{\delta n}{2} [M^2 - 1]^{1/2}$.

surface. The resulting streamtube areas are then used to determine all the necessary gas properties at the downstream surface from Eqs. (14), (15), and (17)-(23). The transfer of mass, momentum and energy into or out of each streamtube is arranged such that what is lost from a given streamtube is gained by the adjacent tube. Thereby mass, momentum and energy are automatically conserved.

In order to render this marching scheme conditionally stable a single iteration on the radius of curvature calculation is required,⁶ with the new value, $(\Theta_g)''$, determined using the downstream surface properties. $(\Theta_g)''$ is then combined with $(\Theta_g)'_k$ in the following expression,⁶

$$(\Theta_g) = (1 - \alpha)(\Theta_g)' + \alpha(\Theta_g)'' \quad (29)$$

This new value for the radius of curvature is then used in the calculation. A value for α of 0.55 is used in the present code. (The scheme is conditionally stable for $\alpha \geq 1/2$).

After the gas phase equations have been solved for the first streamtube the particle properties are determined by sequentially applying the particle momentum, continuity and energy equations. The calculation of gas and particle properties then moves outward to the next streamtube and the procedure is repeated up to the last streamtube[†] where boundary conditions must be applied.

C. Auxiliary Calculations

1. Thermodynamic Input Data

The thermodynamic data are input via curve fits of specific heats¹² of individual species in the JANNAF tables.¹³ These curve fits have the form,

$$C_{p_i} = L_{1i} + L_{2i}\Phi + L_{3i}\Phi^2 + L_{4i}\Phi^3 + L_{5i}\Phi^{-2} \text{ cal/mole-}^\circ\text{K} \quad (30)$$

[†] In general the limiting particle streamlines for each particle size will not extend to the last streamtube.

12. Cruise, D.R., "Information Manual for the Theoretical Propellant Evaluation Program," Naval Weapons Center PEP NOTE TN-U-1 (plus additions), December 1964.
13. JANNAF Thermochemical Tables (Dow Chemical Company, Midland, Mich.), continuously updated.

where $\Phi = T(^{\circ}\text{K})/1000$. The enthalpy is then expressed as,

$$h_i = \int_0^T C_{P_i} dT + L_{6i} \quad \text{kcal/mole} \quad (31)$$

where $L_{6i} = H_f^0 - \int_0^{298} C_{P_i} dT$; H_f^0 being the heat of formation at 298°K .

The entropy is expressed as,

$$S_i = \int_0^T C_{P_i} \frac{dT}{T} + L_{7i} \quad \text{cal/mole-}^{\circ}\text{K} \quad (32)$$

where L_{7i} is the entropy integration constant. The coefficients $L_1 - L_7$ are input on Card group 8.

2. Chemical Kinetic Input Data

Ten possible reaction types are included in the program:

Reaction Type

(1)	A + B	\rightleftharpoons	C + D
(2)	A + B + M	\rightleftharpoons	C + M
(3)	A + B	\rightleftharpoons	C + D + E
(4)	A + B	\rightleftharpoons	C
(5)	A + M	\rightleftharpoons	C + D + M
(6)	A + B	\rightarrow	C + D
(7)	A + B + M	\rightarrow	C + M
(8)	A + B	\rightarrow	C + D + E
(9)	A + B	\rightarrow	C
(10)	A + M	\rightarrow	C + D + M

Reaction types (6)-(10) correspond to reaction types (1)-(5), but proceed in the forward direction only. In Reactions (2), (5), (7) and (10), M is an arbitrary third body. In this program, all species are assumed to have equal third body efficiencies.

The forward rate coefficient, k_f , is input to the code as one of the following 8 types

Rate Coefficient Type†

- (1) $k_f = A$
- (2) $k_f = AT^{-1}$
- (3) $k_f = AT^{-2}$
- (4) $k_f = AT^{-\frac{1}{2}}$
- (5) $k_f = A \exp(B/RT)$
- (6) $k_f = AT^{-1} \exp(B/RT)$
- (7) $k_f = AT^{-\frac{3}{2}}$
- (8) $k_f = AT^N \exp(B/RT)$

The equilibrium constant, K_p , is determined from

$$\ln K_p = - \Delta G/RT \quad (33)$$

where the Gibbs free energy, ΔG , for individual reactions is computed from the input thermodynamic data.

† Rate coefficient data for typical rocket nozzle and plume reactions may be found, e.g., in Ref. 14.

- 14. Jensen, D.E. and Jones, G.A., "Gas-Phase Reaction Rate Coefficients for Rocketry Applications," Rocket Propulsion Establishment Technical Report No. 71/9, October 1971.

3. Particle/Gas Drag and Heat Transfer Coefficients

The momentum and energy exchange (via convective heat transfer) between the small diameter particles and the combustion products in the nozzle cannot adequately be described by simple theoretical expressions (e.g. Stokes law).¹⁵ Empirical correlations of drag and heat transfer coefficients developed by Crowe¹⁶ have therefore been incorporated in the particle/gas interaction terms (see, e.g. Eqs. (2), (3), and (5)).

a. Drag Coefficient - The drag coefficient has been correlated by Crowe¹⁶ in terms of a normalized value,

$$\bar{C}_D = (\dot{C}_D - C_{D_I}) / (C_{D_{FM}} - C_{D_I}) \quad (34)$$

where C_{D_I} is the drag coefficient at very large Reynolds number and $C_{D_{FM}}$ is the free molecular drag coefficient. Note that for $Re_p \ll 1$, $\bar{C}_D \rightarrow 1$, while for $Re_p \gg 1$, $\bar{C}_D \rightarrow 0$.

The expressions needed to evaluate C_D from Eq. (32) are:

$$C_{D_I} = 0.66 + 0.26 [\exp(4 \ln M_p) - 1] + 0.17 \exp[-2.5(\ln M_p / 1.4)^2] \quad (35)$$

$$C_{D_{FM}} = \frac{\exp(-S_1^2/2)}{\sqrt{\pi} S_1^3} (1 + 2S_1^2) + \frac{4(S_1^4 + S_1^2) - 1}{2S_1^4} \operatorname{erf}(S_1) + \frac{2}{3} \frac{\sqrt{\pi}}{\sqrt{T/T_p}} \quad (36)$$

$$\text{where, } S_1 = \sqrt{\gamma/2} M_p \quad (37)$$

$$\bar{C}_D = G(Kn) D(Kn, Re_p) \quad (38)$$

$$G(Kn) = \frac{Kn^{0.4} \exp(1.2 Kn^{0.5})}{1 + Kn^{0.4} \exp(1.2 Kn^{0.5})} \quad (39)$$

15. Soo, S. L., Fluid Dynamics of Multiphase Systems (Blaisdell Publ. Co, Waltham, Mass., 1967).

16. Crowe, C. T., "On the Momentum and Heat Transfer Equations for Two-Phase Plumes," Washington State Univ., March 1971.

and

$$D(Kn, Re_p) = 1 - \exp\left[-\frac{Re_p}{8} Kn^{0.6} \exp(Kn) (C_{D_0} - 0.4)\right] \quad (40)$$

$$Kn = 1.26 \sqrt{\gamma} \frac{M_p}{Re_p} \quad (41)$$

and, $C_{D_0} = 24/Re_p \quad (42)$

The expression used in the particle/gas interaction terms is then,

$$f_p = FFF \frac{C_D}{C_{D_0}} \quad (43)$$

where FFF is a factor (input on Card 15, Cols. 1-10) used to arbitrarily vary C_D to account for uncertainties in the above analysis.

b. Heat Transfer Coefficient - Heat transfer from the particle to the gas is expressed in terms of a Nusselt number as,

$$q = 2\pi r_p Nu k(T_p - T_r) \quad (44)$$

where T_r is the recovery temperature. From Crowe¹⁶ we get the following expression,

$$Nu = Nu_{KD} + \frac{\gamma + 1}{\gamma} Re_p Pr_g \exp(-Re_p/2M_p) \quad (45)$$

$$Nu_{KD} = Nu_o / \left(1 + \frac{5\gamma^{1.5}}{\gamma + 1} (M_p/Re_p Pr_g) Nu_o \right) \quad (46)$$

Nu_o is the Nusselt number in incompressible flow, expressed as

$$Nu_o = 2.0 + 0.459 Re_p^{0.55} Pr_g^{0.33} \quad (47)$$

The recovery temperature is defined as,

$$T_r = T + r \left((u - u_p)^2 + v_p^2 \right) / 2C_p \quad (48)$$

where the recovery factor r is¹⁶;

$$r = 0.9 + (r_{FM} - 0.9) \exp(-Re_p / 2M_p) \quad (49)$$

and

$$r_{FM} = \frac{\gamma}{\gamma + 1} \left(2 + 0.67 \exp(-M^2/\beta) \right) \quad (50)$$

The expression used in the particle/gas interaction term is,

$$g_p = FFG \frac{Nu}{2} \frac{T_p - T_r}{T_p - T} \quad (51)$$

where FFG is a factor (input on Card 15, Cols. 11-20) used to arbitrarily vary g_p to account for uncertainties in the above analysis.

D. Turbulent Boundary Layer Equations[†]

The turbulent compressible boundary layer analysis is initiated by calculating the corresponding adiabatic flat plate, zero pressure gradient, incompressible boundary layer properties, based on a $1/7$ power law¹⁷

[†] The "free stream" properties (subscript "e") for the boundary layer analysis are taken to be the properties in the last (wall) streamtube. It is implicitly assumed that the boundary layer displacement thickness is smaller than the width of the wall streamtube.

17. Schlichting, H., Boundary Layer Theory, 6th Ed. (McGraw-Hill, New York, 1968), p. 599.

velocity profile.[†] Expressions for the incompressible momentum and boundary layer thickness are,

$$\Theta_i(x) = 0.036 / \text{Re}_x^{0.2} \quad (52)$$

$$\delta_i(x) = 10.286 \Theta_i(x) \quad (53)$$

The incompressible skin friction coefficient is evaluated from the Karman-Schoenherr relation.²⁰

$$C_{F_i} = [17.08 (\log_{10} \text{Re}_{\Theta_i})^2 + 25.11 \log_{10} \text{Re}_{\Theta_i} + 6.012]^{-1} \quad (54)$$

Transformation from the compressible to the incompressible boundary layer is accomplished via the Van Driest equations,⁷ which relate the velocity and temperature profiles as follows,

$$\bar{T}_t = \frac{T_e - T_w}{T_{t_e} - T_w} = f\left(\frac{U}{U_e}\right) \quad (55)$$

[†] It was originally anticipated that the complete boundary layer equations including nonequilibrium chemistry would be solved by finite differences and coupled to the nozzle flow solution. A number of boundary layer codes are available, including those of Herring and Mellor¹⁸ and the Aerotherm BLIMP code,¹⁹ which would be very useful for this purpose. However, initial attempts to incorporate the BLIMP code into FULLNOZ showed that it would take more effort than was warranted at this time. Consequently, a more simplified analysis was incorporated into the present code.

18. Herring, H.J. and Mellor, G.L., "A Method of Calculating Compressible Turbulent Boundary Layers," NASA CR-1144, September 1968.
19. Tong, H., Buckingham, A.C., and Morse, H.L., "Nonequilibrium Chemistry Boundary Layer Integral Matrix Procedure," Aerotherm Final Report No. 73-67, July 1973.
20. Hopkins, E.J., Keener, E.R., and Louie, P.T., "Direct Measurements of Turbulent Skin Friction on a Nonadiabatic Flat Plate at Mach Number 6.5 and Comparisons with Eight Theories," NASA TN D-5675, February 1970.

The general functional relation used in the analysis is

$$\bar{T}_t = \left(\frac{U}{U_e} \right)^n \quad (56)$$

Equation (56) becomes the Crocco relation for $n = 1$ and the quadratic for $n = 2$.

A general expression between T/T_e and U/U_e can then be written

$$\frac{T}{T_e} = A \left(\frac{U}{U_e} \right)^2 + B \left(\frac{U}{U_e} \right)^n + C \quad (57)$$

where

$$A = \left(\frac{T_{te}}{T_e} - 1 \right); B = \left(\frac{T_{te}}{T_e} - \frac{T_w}{T_e} \right); C = \frac{T_w}{T_e}$$

From Back and Cuffel,²¹ wall cooling ($T_w < T_{aw}$) tends to make the exponent n closer to 1.0, while wall heating causes n to approach 2 or more. The value of n selected here was 1.2. It was determined by fitting recent data of Keener and Hopkins⁸ for which, $T_w/T_{aw} = 0.32$. (This corresponds to a wall temperature of 1000-1300 K.) Equation (57) was used to generate a table of T/T_e vs. U/U_e , which is then used to determine the compressible skin friction coefficient via the Van Driest transformation

$$\frac{C_{fc}}{C_{fi}} = \left\{ \int_0^1 \left(\frac{T_e}{T} \right)^{1/2} d \left(\frac{U}{U_e} \right) \right\}^2 \quad (58)$$

The Stanton number was determined from the relation

$$S_t = 0.35 C_{fc} \quad (59)$$

which is supported by the experimental data of Back and Cuffel²¹ for accelerating flows.

21. Back, L.H. and Cuffel, R.F., "Relationship Between Temperature and Velocity Profiles in a Turbulent Boundary Layer along a Supersonic Nozzle with Heat Transfer," AIAA J. 8, 2066-2069 (1970).

Wall shear stress and heat transfer are then calculated from

$$\tau_w = \frac{1}{2} \rho_e U_e^2 C_{f_c} \quad (60)$$

$$\dot{q}_w = -S_t \rho_e U_e C_p (T_{t_e} - T_w) \quad (61)$$

In order to obtain the velocity profiles the friction velocity profile, $(U/U_\tau)_i = f(U/U_e)$ is first determined from another Van Driest transformation,⁷

$$\left(\frac{U}{U_\tau}\right)_i = \left(\frac{2}{C_{f_c}}\right)^{1/2} \int_0^{U/U_e} \left(\frac{T_e}{T}\right)^{1/2} d\left(\frac{U}{U_e}\right) \quad (62)$$

where the friction velocity is defined as,

$$U_{\tau_i} = \left(\rho_e U_e^2 C_{f_i} / 2\rho_w\right)^{1/2} \quad (63)$$

$(y/\delta)_i$ is then obtained from the standard incompressible boundary layer profiles^{22†}

$$\frac{yU_\tau}{\nu} = \frac{U}{U_\tau}; \quad \frac{U}{U_\tau} < 5 \quad (64)$$

$$\frac{yU_\tau}{\nu} = \exp\left\{\frac{U/U_\tau + 3.05}{2.5}\right\}; \quad 5 \leq \frac{U}{U_\tau} \leq 13.96 \quad (65)$$

$$\frac{yU_\tau}{\nu} = \exp\left\{\frac{U/U_\tau - 5.05}{2.5}\right\}; \quad \frac{U}{U_\tau} > 13.96 \quad (66)$$

† The values of yU_τ/ν are normalized by $\exp\{[U/U_\tau \text{ max} - 5.05]/2.5\}$ to avoid the calculation of the ν profile in the boundary layer.

22. Kays, W.M., Convective Heat and Mass Transfer (McGraw-Hill, New York, 1966).

Another Van Driest transformation yields $(y/\delta)_c = (y/\delta)_i$. Once the compressible profiles are known, Θ/δ , δ^*/δ , and the shape factor are determined from

$$\Theta/\delta = \int_0^1 \left(\frac{U}{U_e} \right) \left(1 - \frac{U}{U_e} \right) d\left(\frac{y}{\delta} \right) \quad (67)$$

$$\delta^*/\delta = \int_0^1 \left(1 - \frac{U}{U_e} \right) d\left(\frac{y}{\delta} \right) \quad (68)$$

$$H_{12} = \frac{\delta^*/\delta}{\Theta/\delta} \quad (69)$$

The variation of momentum thickness along the nozzle is determined via integration of the momentum integral equation,

$$\frac{C_f}{2} = \frac{d\Theta}{dx} + [H_{12} + 2] \Theta \frac{1}{U_e} \frac{dU_e}{dx} + \frac{\Theta}{\rho_e} \frac{d\rho_e}{dx} + \frac{\Theta}{R} \frac{dR}{dx} \quad (70)$$

by the use of backward differences.[†] δ and δ^* are then evaluated from Eqs. (67) and (68).

III. PREPARATION OF INPUT DATA

All necessary information for preparing input data is given below; Fig. 2 defines some of the input for a sample case. Many of the input parameters that are left blank are used in the companion rocket plume code¹⁰ (AIPP), but not in FULLNOZ.

[†] This requires that a throat value of Θ be assumed.

Card No.	Cols.	Fortran Name	Description	Format
3	23-34	XLMAX	Maximum distance (cm) along axis for which calculations will be made. XLMAX should be set somewhat larger than axial distance to nozzle exit plane in order to complete calculation at last wall point	E12.5
4	1- 5	ITYPE	Inner boundary condition (nozzle axis) Set ITYPE = 1	15
	6-10	IKIND	Outer boundary condition; set IKIND = 1 for wall boundary condition	
	11-15		Not used; leave blank	
	16-20		Not used; leave blank	
	21-25		Not used; leave blank	
	26-30	IBUGSH	Debug printout index; 0 - No debug printout 1 - Extensive printout for debugging	15
	31-35		Not used; leave blank	
	36-40		Not used; leave blank	
	41-45		Not used; leave blank	
	46-50	IPART	Particle indicator; 0 - No particles 1 - Particles in flow	15
	51-55		Not used; leave blank	
	56-60	ITURB	Turbulent flow indicator; 0 - Inviscid 3 - Constant values of turbulent mixing parameters, μ , Pr and Le are input on Card 10	15
5	1- 5	KMAX	Initial number of streamtubes plus one (axis is counted as K = 1; maximum is 40)	15
	6-10	NN	Number of terms for C_p polynomial curve fit plus one; for curve fits supplied with program, ¹² NN = 6. If fewer coefficients are used for a given species NN is left at 6 and zeroes are input for the missing coefficients (see Card 8)	

Card No.	Cols.	Fortran Name	Description	Format
5	11-15	KP	Output control; number of orthogonal surfaces between print stations. To control print via axial distance (DXLSS, Card 3, Cols. 11-22) instead of orthogonal surfaces set KP larger than LPLANE (Card 5, Cols. 36-40)	
	16-20	MMAX	Number of points needed to describe shape of inner boundary; always set equal to 2 for axis boundary	I5
	21-25	NMAX	Number of points required to define nozzle wall contour (50 maximum)	I5
	26-30	NDS	Total number of gas species in flow (25 maximum)	I5
	31-35	NITER	Maximum number of iterations allowed for iterative solutions (e.g. calculation of streamtube properties for variable Y) Recommended value: NITER = 50	I5
	36-40	LPLANE	Maximum number of integration steps for entire calculation	I5
	41-45	IKINE	Number of chemical reactions (40 maximum)	I5

NOTE: IF NO BOUNDARY LAYER CALCULATIONS ARE TO BE MADE CARD 6 MAY BE LEFT BLANK, BUT MUST BE INCLUDED

6	1- 5	TWALL	Nozzle wall temperature ($^{\circ}\text{K}$) (assumed constant)	F5.0
	8	IBLFLG	Boundary layer property printout indicator; 0 - Print Re , δ_1 , θ_1 , C_{f_1} , U_T , δ_c , δ_c^* , θ_c , C_{f_c}/C_{f_1} , H_{12} , q_w , T_w 1 - Print above plus velocity and temperature profiles	I1

Card No.	Cols.	Fortran Name	Description	Format
6	11	IBL	Boundary layer calculation indicator; 0 - Do not calculate boundary layer 1 - Calculate boundary layer	11
7	1- 9	ALPHAH	Factor that multiplies maximum stable step size (see Eq. (28)); Recommended value: ALPHAH = 0.8	F9.4
	10-18	EPSLON	Amount by which streamtube Mach numbers must exceed one in order for the calculation to continue. Recommended value: EPSLON = 0.01	F9.4
	19-27	TOL	Convergence tolerance for iterative solutions; Recommended value: TOL = 1×10^{-4}	F9.4
	28-36	DELTA	Metric exponent [†] ; 0 - Two-dimensional flow 1 - Axially symmetric flow	F9.4
	37-45	ATOL	Maximum allowable fractional change in streamtube area per step. Recommended value: ATOL = 0.1	F9.4

NOTE: CARD GROUP 8 DEFINES THE THERMODYNAMIC DATA FOR EACH SPECIES. THERE ARE 2 CARDS PER SPECIES AND NDS SPECIES

8.1.1	1-13	A(1)	$\left. \begin{array}{l} L_1 \\ L_2 \\ L_3 \\ L_4 \end{array} \right\}$	Specific heat polynomial constants for first species, see Section II.C.1 (cal/mole °K)	E13.5
	14-26	A(2)			
	27-39	A(3)			
	40-52	A(4)			

[†] The particle conservation equations are written only for axially symmetric flow. Thus two-dimensional solutions can only be obtained for flows without particles.

<u>Card No.</u>	<u>Cols.</u>	<u>Fortran Name</u>	<u>Description</u>	<u>Format</u>
8.1.2	1-13	A(5)	L ₅ , Specific heat polynomial constant for first species	E13.5
	14-26	A(6)	L ₆ , Enthalpy constant of integration for first species	E13.5

$$\left(\Delta H_f^\circ - \int_0^{298} C_p dT \right) (\text{kcal/mole})$$

27-39	CS(1)	L ₇ , entropy constant of integration for first species (cal/mole °K)
-------	-------	--

⋮

8.NDS.1 Repeat thermodynamic data for NDS
8.NDS.2 species

NOTE: CARD GROUP 9 IDENTIFIES EACH SPECIES AND SPECIFIES VARIOUS TRANSPORT PROPERTIES. MUST BE IN SAME ORDER AS CARD GROUP 8

9.1	1- 4	IDENT(1)	Species name; first species (the remaining data on Card 9.1 also apply to the first species)	A4
	13-24	MUO(1)	Viscosity at reference temperature [†] (g/cm-sec); only used when flow contains particles. FOR IPART = 0, MUO MUST = 0 FOR ALL SPECIES	E12.4
	25-36	TO(1)	Reference temperature for viscosity (°K)	E12.4
	37-48	OMEGA(1)	Exponent describing viscosity/temperature relation; $\mu \propto T^\omega$	E12.4
	49-60	PR(1)	Reciprocal of species Prandtl number	E12.4
	61-72	SC(1)	Reciprocal of species Schmidt number	E12.4
	73-80	MW(1)	Species molecular weight	E8.2

⋮

[†] Values for common gases can be found in most physics and chemistry handbooks.

<u>Card No.</u>	<u>Cols.</u>	<u>Fortran Name</u>	<u>Description</u>	<u>Format</u>
9.NDS			Repeat for each species	
			NOTE: CARD 10 IS NOT NEEDED IF ITURB = 0 (CARD 4, COLS. 56-60)	
10	1-10	TLE	Turbulent Lewis number (constant)	
	11-20	TPR	Turbulent Prandtl number (constant)	
	21-30	EDDYK	Eddy viscosity, g/cm-sec (constant)	
			NOTE: CARD GROUPS 11 AND 12 DEFINE THE LOCATION OF THE INITIAL ORTHOGONAL SURFACE, AND THE FLOW ANGLE, PRESSURE, TEMPERATURE, VELOCITY AND SPECIES MASS FRACTIONS WITHIN EACH STREAMTUBE	
11	1-12	X(1)	Initial axial position on inner boundary (cm)	E12.4
	13-24	R(1)	Initial radial position on inner boundary (cm); usually equal to 0 for axis boundary	E12.4
	25-36	PHI(1)	Initial flow angle [†] on inner boundary (radians); usually equal to 0 for axis boundary	E12.4
12.1.1	1-12	X(2)	Axial position at outer boundary of first streamtube (cm)	E12.4
	13-24	R(2)	Radial position at outer boundary of first streamtube (cm)	E12.4
	25-36	PHI(2)	Flow angle at outer boundary of first streamtube (radians)	E12.4
	37-48	P(2)	Average pressure in first streamtube (atm)	E12.4
	49-60	T(2)	Average temperature in first streamtube (°K)	E12.4
	61-72	U(2)	Average velocity in first streamtube (cm/sec)	E12.4
12.1.2	1-10	C(1, 2)	Average mass fraction of first species in first streamtube	E10.3

[†] Flow angle is defined as the angle between the flow velocity vector and the axis.

Card No.	Cols.	Fortran Name	Description	Format
12.1.2	11-20	C(2,2)	Average mass fraction of second species in first streamtube	E10.3
	⋮		⋮	⋮
	71-80	C(8,2)	Average mass fraction of eighth species in first streamtube	E10.3

⋮
12.1.I($\frac{NDS}{8}$)

NOTE: I($\frac{NDS}{8}$) DENOTES THE NEXT INTEGER LARGER

THAN ($\frac{NDS}{8}$). REPEAT CARD 12.1.2 TO INCLUDE
NDS SPECIES, 8 SPECIES/CARD

12.2.1	1-12	X(3)	Axial position at outer boundary of second streamtube (cm)	E12.4
	⋮		⋮	
	61-72	U(3)	Average velocity in second streamtube (cm/sec)	E12.4

12.2.2	1-10	C(1,3)		E10.3
	⋮			⋮
	71-80	C(8,3)		⋮

12.2.I($\frac{NDS}{8}$)		C(NDS, 3)	Average mass fraction of last species in second streamtube	E10.3
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12.(KMAX-1).1	1-12	X(KMAX)	Axial position at outer boundary of last (KMAX-1) streamtube (cm)	E12.4
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12.(KMAX-1).I($\frac{NDS}{8}$)		C(NDS, KMAX)	Average mass fraction of last species in last streamtube	E10.3
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NOTE: CARD 13 DEFINES THE POSITION OF THE INNER
BOUNDARY

13.1	1-12	XW(1)	Initial axial position (cm); repeat of X(1) on Card 10	E12.4
	13-24	RW(1)	Initial radial position (cm); repeat of R(1) on Card 10	E12.4

Card No.	Cols.	Fortran Name	Description	Format
13.1	25-36	PHIW(1)	Leave rest of card blank; program will determine these inner boundary properties	E12.4
	37-48	PW(1)		E12.4
	49-60	SW(1)		E12.4
13.2	1-12	XW(2)	Axial distance greater than XLMAX (Card 3, Cols. 23-34)(cm). Since the inner boundary is an axis the program will interpolate linearly between XW(1) and XW(2) to get axial locations of orthogonal surfaces. Therefore, set $XW(2) = 1 \times 10^{10}$	E12.4
	13-24	RW(2)	Final radial position of inner boundary (cm); since inner boundary is an axis set RW(2) equal to R(1) on Card 11	E12.4
	25-36	PHIW(2)	Leave rest of card blank; program will determine these inner boundary properties	E12.4
	37-48	PW(2)		E12.4
	49-60	SW(2)		E12.4

NOTE: CARD GROUP 14 DEFINES THE POSITION OF THE OUTER BOUNDARY (NOZZLE WALL CONTOUR)

14.1	1-12	XB(1)	Initial axial position of nozzle wall (cm); equal to X(KMAX) on Card 12. (KMAX-1).1	E12.4
	13-24	RB(1)	Initial radial position of nozzle wall (cm); equal to R, KMAX) on Card 12. (KMAX-1).1	E12.4
14.2	1-12	XB(2)	Axial position (cm) of second point along nozzle wall	E12.4
	13-24	RB(2)	Radial position (cm) of second point along nozzle wall	E12.4
14.NMAX	1-12	XB(NMAX)	Final axial position (cm) of nozzle wall	E12.4
	13-24	RB(NMAX)	Final radial position (cm) of nozzle wall	E12.4

NOTE: CARD GROUPS 15-19 AND 20-24 (OR 25-32) ARE INCLUDED ONLY IF PARTICLES ARE IN FLOW, i.e. IPART = 1 (CARD 4, COLS. 46-50)

15	1-10	FFF	Factor which multiplies particle/gas drag coefficient (see Section II.C.3)	E10.3
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<u>Card No.</u>	<u>Cols.</u>	<u>Fortran Name</u>	<u>Description</u>	<u>Format</u>
15.	11-20	FFG	Factor which multiplies particle/gas heat transfer coefficient (see Section II. C. 3)	E10.3
	21-30	CL	Liquid particle specific heat (cal/g-°K)	E10.3
	31-40	CS	Solid particle specific heat (cal/g-°K)	E10.3
	41-50	HTRAN	Particle heat of solidification (cal/g)	E10.3
	51-60	WT	Particle molecular weight	E10.3
16	1-10	RHSS	Particle density (g/cm ³)	E10.3
	11-20		Not used; leave blank	
	21-30		Not used; leave blank	
	31-40	TPS	Particle solidification temperature (°K)	E10.3
17	1- 5	NPG	Number of particle groups (8 maximum)	I5
	6-10	NC	Index noting whether particle properties are constant or variable along initial orthogonal surface 0 - constant 1 - variable	I5
<p>NOTE: NBL() INDICATES RADIAL EXTENT OF EACH PARTICLE GROUP ALONG INITIAL ORTHOGONAL SURFACE (SEE FIG. 2); DEFINED AS LAST STREAMTUBE NUMBER CONTAINING PARTICLES PLUS 2. FOR EXAMPLE, IF THE FIRST PARTICLE GROUP EXTENDS TO STREAMTUBE NO. 12, NBL(1) = 14</p>				
18	1- 5	NBL(1)	Last streamtube containing first particle group along initial orthogonal surface (streamtube number + 2)	I5
	.			
	.			
		NBL(NPG)	Last streamtube containing NPG particle group (streamtube number + 2)	I5
19	1-10	RP(1)	Radius of first particle group (cm)	8E10.3
	.	.		
	.	.		
		RP(NPG)	Radius of NPG particle group (cm)	

<u>Card No.</u>	<u>Cols.</u>	<u>Fortran Name</u>	<u>Description</u>	<u>Format</u>
NOTE: IF PARTICLE PROPERTIES ALONG INITIAL ORTHOGONAL SURFACE ARE <u>CONSTANT</u> (NC = 0) CARDS 20-24 ARE INPUT; IF NC = 1 THE NEXT CARD GROUP IS NO. 25				
20	1-10	WI(1)	Velocity of first particle group in stream-line direction (cm/sec)	E10.3
	11-20	WI(2)	Velocity of second particle group in streamline direction (cm/sec)	E10.3

21		WI(NPG)	Velocity of NPG particle group in stream-line direction (cm/sec)	E10.3
	1-10	VI(1)	Velocity of first particle group normal to streamline (cm/sec)	E10.3
	11-20	VI(2)	Velocity of second particle group normal to streamline (cm/sec)	E10.3

22
		VI(NPG)	Velocity of NPG particle group normal to streamline (cm/sec)	E10.3
	1-10	TPI(1)	Temperature of first particle group ($^{\circ}$ K)	E10.3
	11-20	TPI(2)	Temperature of second particle group ($^{\circ}$ K)	E10.3
23

		TPI(NPG)	Temperature of NPG particle group ($^{\circ}$ K)	E10.3
	1-10	RHP(1)	Particle cloud density of second particle group (g/cm ³)	E10.3
	11-20	RHPI(2)	Particle cloud density of second particle group (g/cm ³)	E10.3

		RHPI(NPG)	Particle cloud density of NPG particle group (g/cm ³)	E10.3

<u>Card No.</u>	<u>Cols.</u>	<u>Fortran Name</u>	<u>Description</u>	<u>Format</u>
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NOTE: IF A PARTICLE GROUP IS AT THE SOLIDIFICATION TEMPERATURE THE AMOUNT OF HEAT WHICH HAS BEEN TRANSFERRED FROM THE LIQUID PARTICLES (i. e. SOME FRACTION OF THE TOTAL HEAT OF SOLIDIFICATION) IS INPUT ON CARD 24. IF THE PARTICLE TEMPERATURE IS ABOVE OR BELOW THE SOLIDIFICATION TEMPERATURE, DENG(J) = 0 (J = 1, NPG)

24	1-10	DENG(1)	Heat transferred from first particle group at solidification temperature (cal)	E10.3
	11-20	DENG(2)	Heat transferred from second particle group at solidification temperature (cal)	E10.3

		DENG(NPG)	Heat transferred from NPG particle group at solidification temperature (cal)	E10.3

NOTE: IF PARTICLE PROPERTIES ALONG INITIAL ORTHOGONAL SURFACE ARE VARIABLE (NC = 1) CARDS 25-32 ARE INPUT (SEE FIG. 2)

25.1	1-10	W(1,1)	Streamwise velocity of first particle group at $r = 0$ (cm/sec)	E10.3
	11-20	W(1,2)	Streamwise velocity of second particle group at $r = 0$ (cm/sec)	E10.3

		W(1,NPG)	Streamwise velocity of NPG particle group at $r = 0$ (cm/sec)	E10.3
26.1	1-10	V(1,1)	Normal velocity of first particle group at $r = 0$ (cm/sec); generally = 0	E10.3
	11-20	V(1,2)	Normal velocity of second particle group at $r = 0$ (cm/sec); generally = 0	E10.3

		V(1,NPG)	Normal velocity of NPG particle group at $r = 0$ (cm/sec); generally = 0	E10.3

<u>Card No.</u>	<u>Cols.</u>	<u>Fortran Name</u>	<u>Description</u>	<u>Format</u>
27.1	1-10	TP(1,1)	Temperature of first particle group at $r = 0$ ($^{\circ}\text{K}$)	E10.3
	11-20	TP(1,2)	Temperature of second particle group at $r = 0$ ($^{\circ}\text{K}$)	E10.3
	.	.		
	.	.		
		TP(1,NPG)	Temperature of NPG particle group at $r = 0$ ($^{\circ}\text{K}$)	E10.3
28.1	1-10	RHP(1,1)	Particle cloud density of first particle group at $r = 0$ (g/cm^3)	E10.3
	11-20	RHP(1,2)	Particle cloud density of second particle group at $r = 0$ (g/cm^3)	E10.3
	.	.		.
	.	.		.
		RHP(1,NPG)	Particle cloud density of NPG particle group at $r = 0$ (g/cm^3)	E10.3
29.1	1 -5	ICOND(1,1)	Index which indicates whether first particle group is at solidification temperature at $r = 0$ ICOND = 0; No = 1; Yes	I5
	6 -10	ICOND(1,2)	Same as above for second particle group	I5

		ICOND(1,NPG)	Same as above for NPG particle group	I5
30.1	1-10	DENG(1,1)	Heat transferred from first particle group at $r = 0$ at solidification temperature (cal) (See NOTE on page)	E10.3
	11-20	DENG(1,2)	Same as above for second particle group	E10.3

		DENG(1,NPG)	Same as above for NPG particle group	E10.3
25.2	Same as 25.1, 26.1, 27.1, 28.1, 29.1, and 30.1, except for first streamtube			
26.2				
27.2				
28.2				
29.2				
30.2				

Card No.	Cols.	Fortran Name	Description	Format
25.(NBL-1)	}		.	
26.(NBL-1)			.	
27.(NBL-1)				
28.(NBL-1)			Same as above except for last stream- tube containing particles	
29.(NBL-1)				
30.(NBL-1)				

NOTE: CARDS 31 AND 32 LOCATE THE INITIAL BOUNDARY
OF EACH PARTICLE GROUP

31	1-10	TRBDY(1)	Distance from axis, along initial orthog- onal surface, to boundary of first particle group (cm) (see Fig. 2)	E10.3
	11-20	TRBDY(2)	Same as above for second particle group	E10.3
	.	.		.
	.	.		.
		TRBDY(NPG)	Same as above for NPG particle group	E10.3
32	1-10	DNDSP(1)	Distance from outer boundary of last streamtube containing particle to boundary of first particle group at initial orthogonal surface (cm) (see Fig. 2)	E10.3
	11-20	DNDSP(2)	Same as above for boundary of second particle group	E10.3
	.	.		.
	.	DNDSP(NPG)	Same as above for boundary of NPG particle group	.

NOTE: THE FOLLOWING CARDS CONTAIN THE REACTION
MECHANISM AND RATE COEFFICIENTS. USE ONLY
IF IKINE (CARD 5, COLS. 41-45) IS GREATER THAN
0. (SEE SECTION II.C.2 FOR REACTION AND RATE
COEFFICIENT TYPES)

33.1	1- 4	IZD(1)	Species A	A4
	7		+ sign	
	8-11	IZD(2)	Species B	A4
	14		+ sign	
	15-20		Blank or M	A4

<u>Card No.</u>	<u>Cols.</u>	<u>Fortran Name</u>	<u>Description</u>	<u>Format</u>
33.1	21		= sign	
	22-25	IZD(3)	Species C	A4
	28		+ sign (if needed)	
	29-32	IZD(4)	Species D	A4
	35		+ sign (if needed)	
	36-39	IZD(5)	Species E	A4
	49-50	IRR	Reaction type (1 to 10)	I2
	51	IRT	Rate coefficient type (1 to 8)	I1
	52-59	RC(1)	Pre-exponential factor, A (cm-molecule-sec units)	E8.2
	60-63	RC(2)	Temperature exponent, N	F4.1
	64-72	RC(3)	Activation energy, B (cal/mole)	F9.1

•
•
•

33. IKINE

Same as above for IKINE reaction

IV. PRELIMINARY RESULTS

This section gives the results of several sample calculations made with **FULLNOZ** and an analysis performed to determine heterogeneous electron-ion recombination rates in solid propellant nozzle flows.

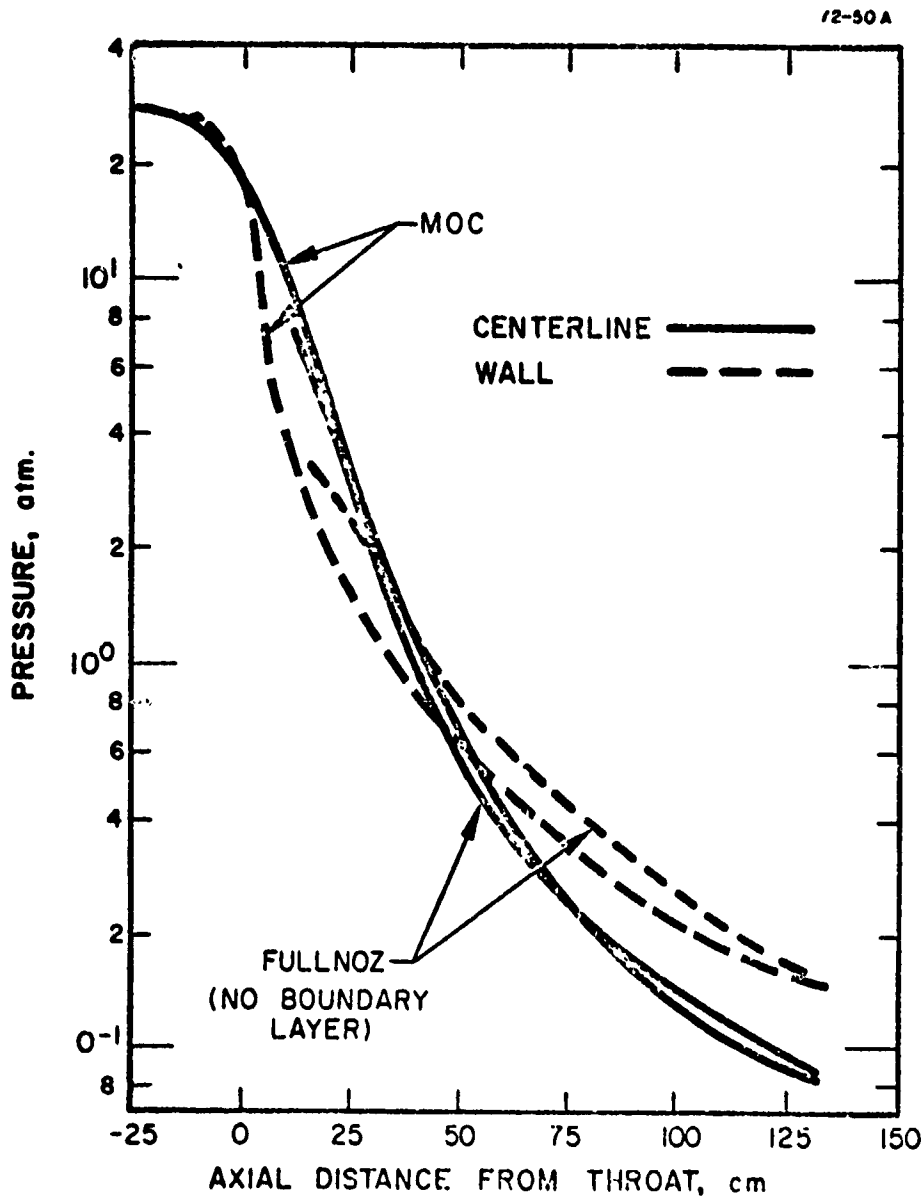


Figure 3. Comparison between Method of Characteristics (MOC) and FULLNOZ calculations of MM-Stage 2 nozzle pressure distributions.

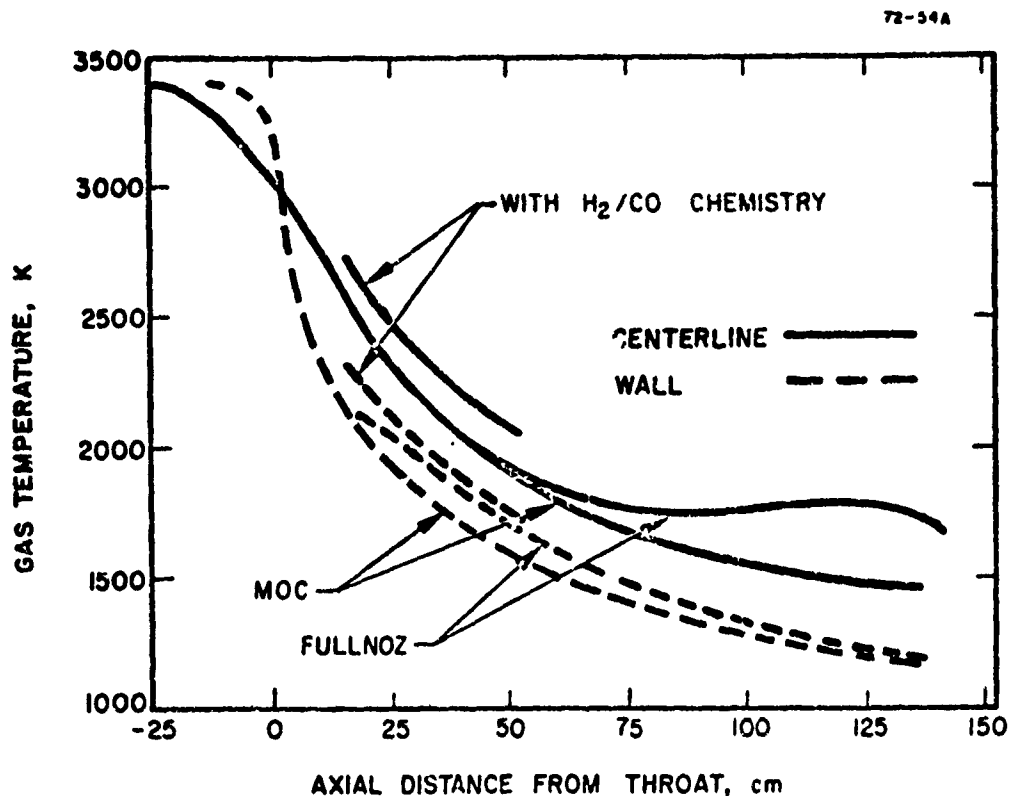


Figure 4. Comparison between Method of Characteristics (MOC) and FULLNOZ calculations of MM-Stage 2 nozzle gas temperature distributions.

A. Sample Calculations

Initial calculations with FULLNOZ were compared with calculations for the Minuteman, Stage 2 nozzle using a two-phase constant γ method of characteristics (MOC) code.³ The purpose of this comparison was to check the numerical accuracy of the code. Input data for the calculations are given in Ref. 23.

Figures 3 and 4 show the pressure and temperature distributions along the centerline and wall (with no boundary layer effects). The pressure distributions compare very well, but FULLNOZ temperatures are slightly higher than those calculated via the MOC code. The centerline gas temperature calculated with FULLNOZ shows a more pronounced effect of gas/parti-

23. Pergament, H.S. and Mikatarian, R.R., "Predictions of Minuteman Exhaust Plume Electrical Properties," AeroChem TP-281, July 1972.

cle interactions than was demonstrated by the MOC code. The results of a short run with FULLNOZ, including a set of 10 reactions involving H_2/CO chemistry, (see Ref. 23 for the reaction mechanism and rate coefficients) are also shown on Fig. 4.

Figures 5 and 6 show the influence of boundary layer heat transfer and shear stress on the temperature and velocity within the wall streamtube. The largest effects are observed for the cold ($500^\circ K$) wall; when the wall is near the adiabatic wall temperature ($\approx 3000^\circ K$) the results are similar to those for no boundary layer.

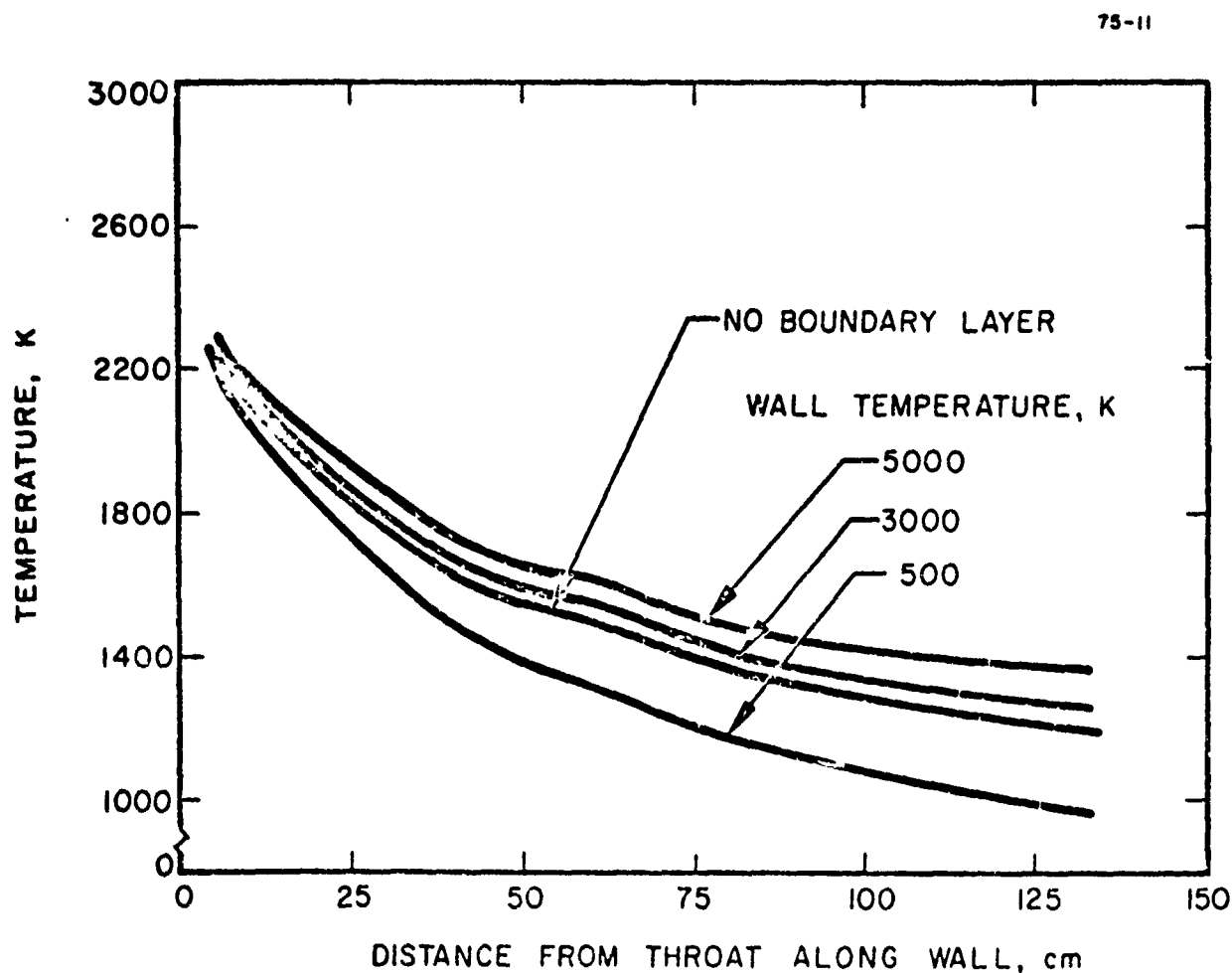


Figure 5. Influence of nozzle boundary layer on wall streamtube temperature.

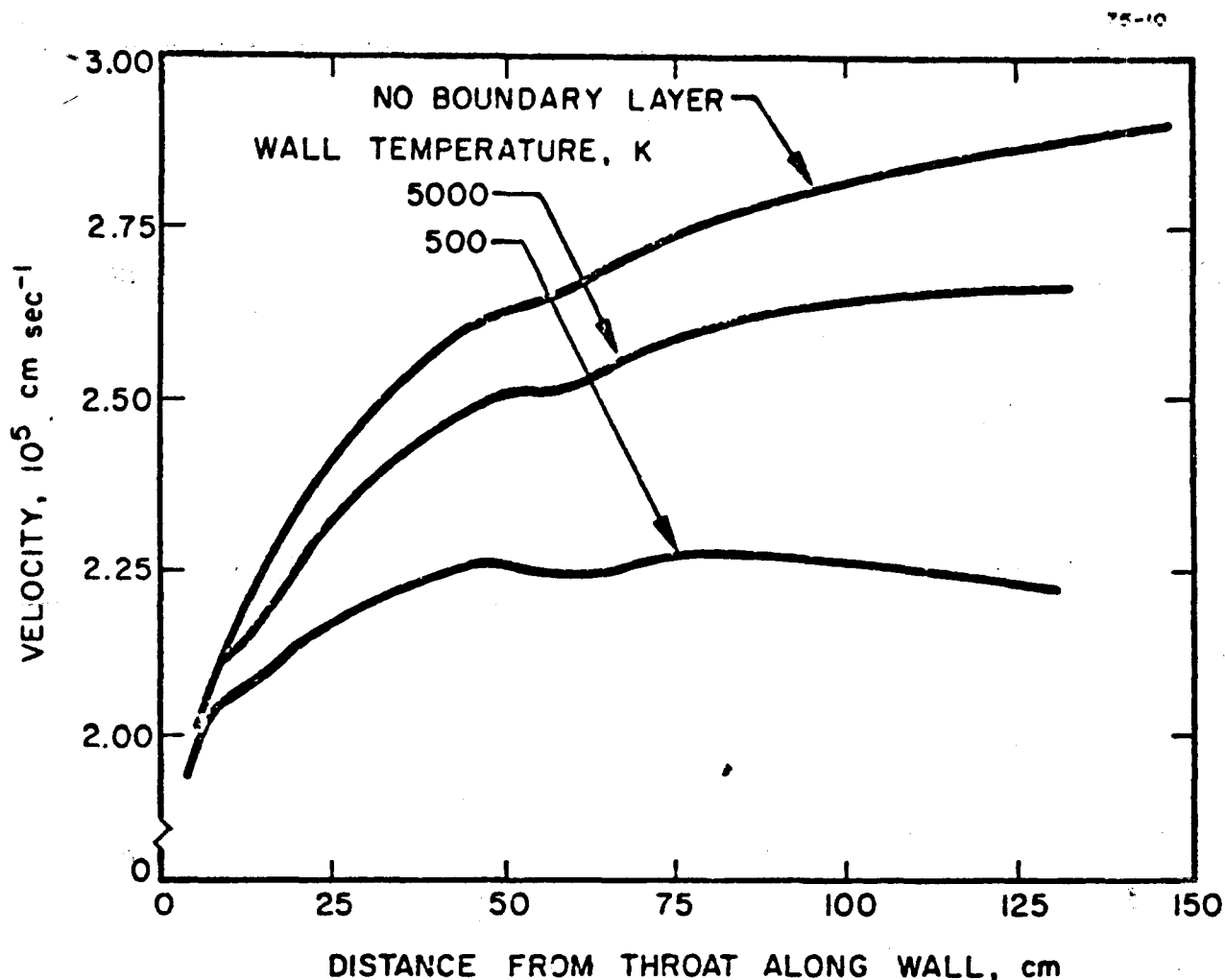


Figure 6. Influence of nozzle boundary layer on wall streamtube velocity.

Figure 7 shows the boundary layer velocity and temperature profile at the nozzle exit plane; Table 1 demonstrates that the boundary layer displacement and momentum thicknesses are much less than the wall streamtube thickness, i.e. all boundary layer effects are confined to the wall streamtube.

Figures 8 and 9 show the results of a parametric series of calculations in which the gas/particle drag and heat transfer coefficients were arbitrarily varied (via the factors FFF and FFG on Card 15) over their approximate ranges of uncertainty to test the effect on exit plane and particle properties. Figure 8 shows that varying FFF and FFG can have significant effects on gas temperatures and velocities. Figure 9 demonstrates that the 4μ diameter particles can be either at the solidification temperature or completely solidified, depending on the value chosen for FFG.

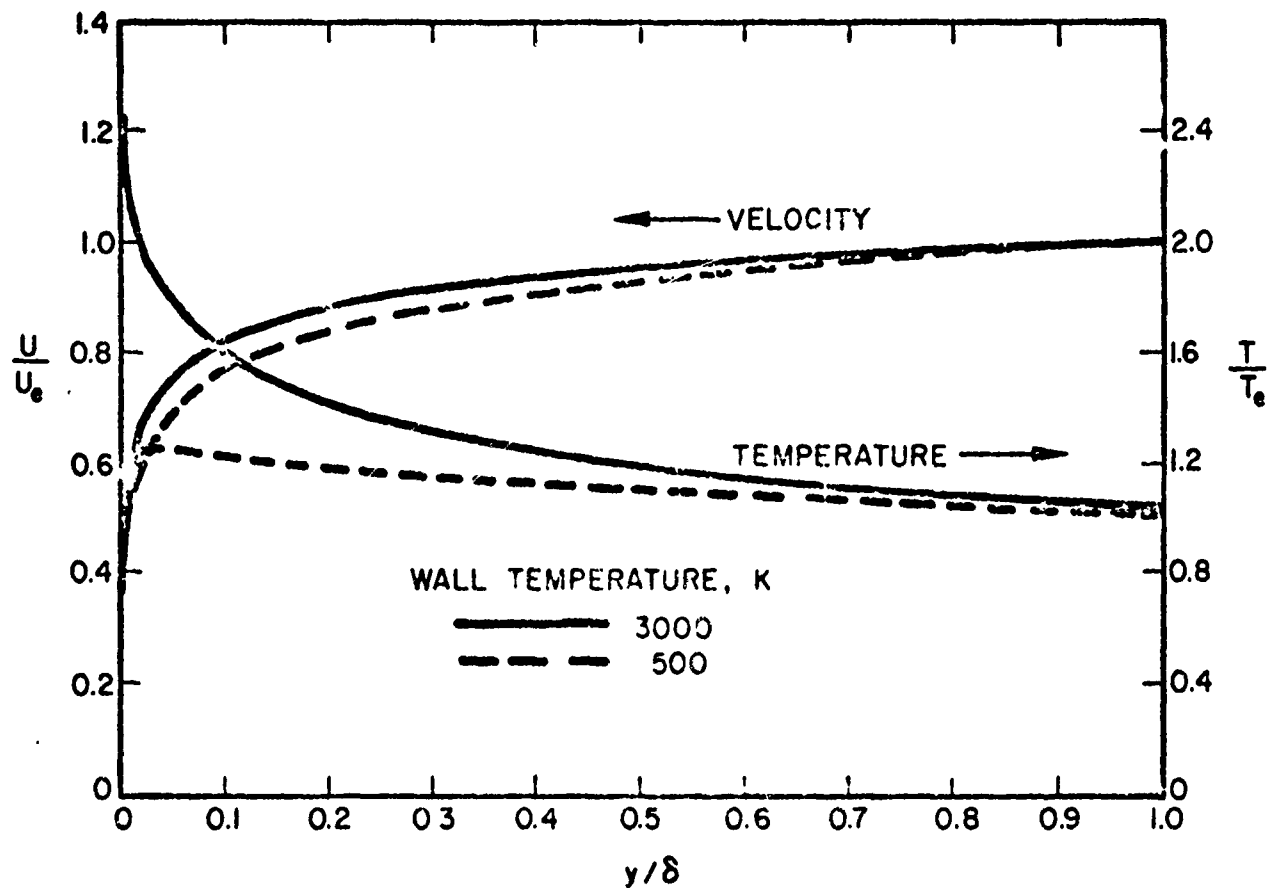


Figure 7. Boundary layer profiles at MM-Stage 2 nozzle exit plane.

TABLE 1

NOZZLE EXIT PLANE BOUNDARY LAYER PARAMETERS

MM-Stage 2 (Ref. 23)

Nozzle Exit Radius = 60.9 cm

Axial Distance From Throat = 134 cm

	Wall Temperature, °K	
	500	3000
Wall Streamtube Width, cm	1.24	1.38
Boundary Layer Thickness, cm	4.45	2.50
Displacement Thickness, cm	0.451	0.196
Momentum Thickness, cm	0.355	0.161

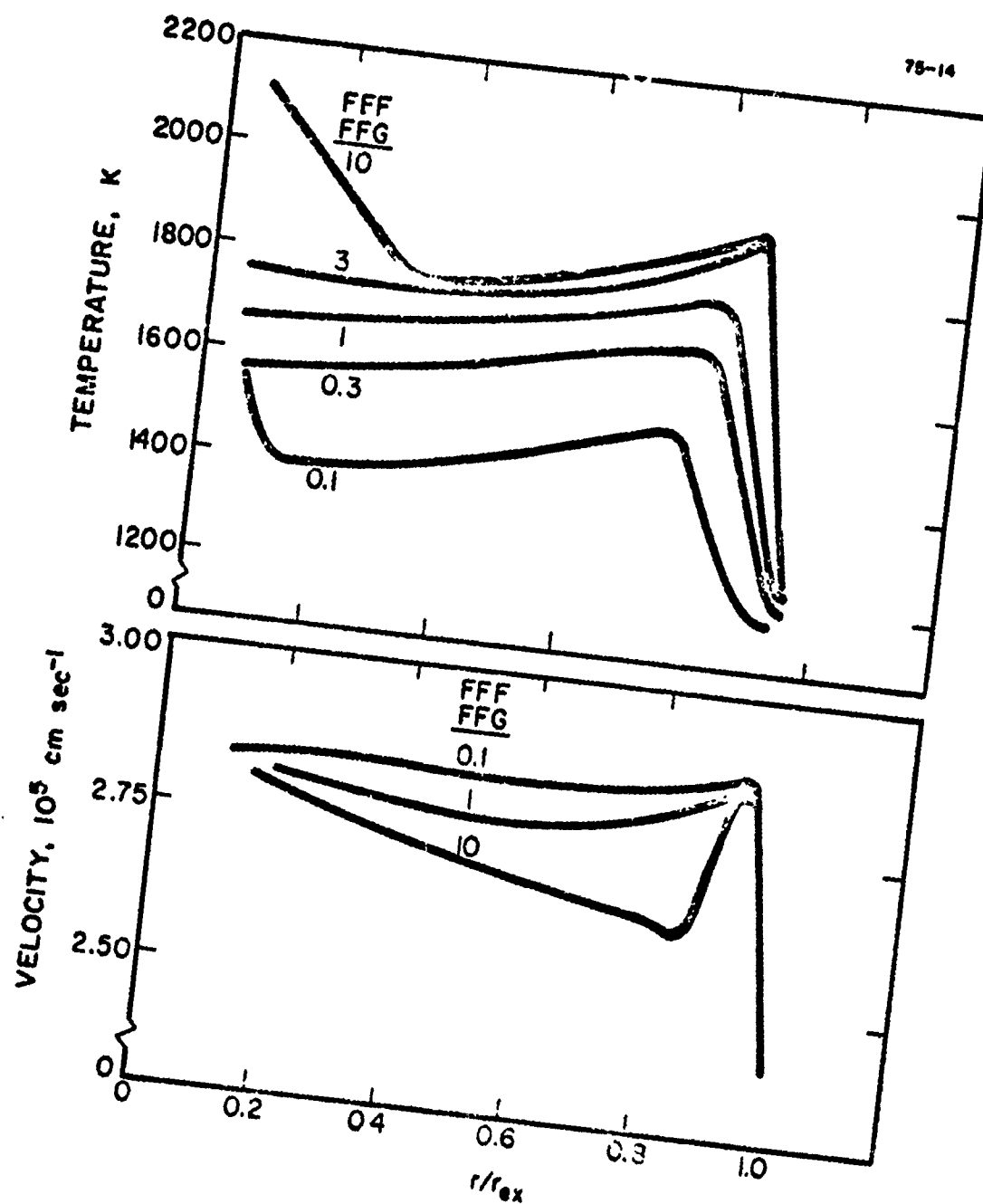


Figure 8. Influence of particle drag and heat transfer coefficients on exit plane gas properties.

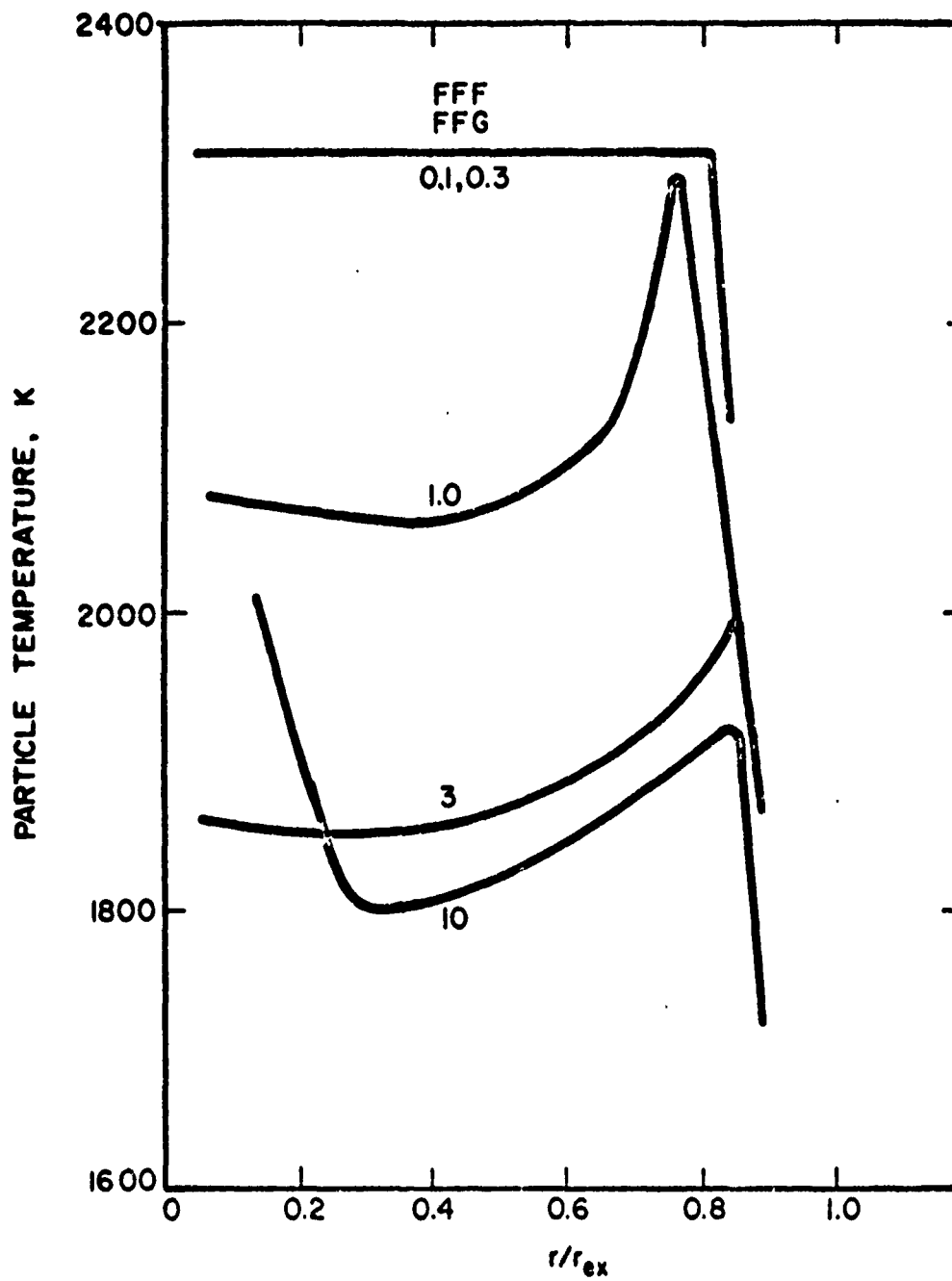


Figure 9. Influence of particle drag and heat transfer coefficients on exit plane particle temperatures.

B. Heterogeneous Electron Ion Recombination

One of the unanswered questions in determining nozzle exit plane electron mole fractions is the extent to which heterogeneous electron/ion recombination on the surface of solid particles can enhance homogeneous electron/ion recombination rates. Towards this end we have adopted a model developed at AeroChem by Calcote, Kurzius and Silla²⁴ in which a negatively charged particle (particles are negatively charged because the electrons will reach the surface more rapidly than positive ions) is neutralized by positive ions striking the surface. Thus, instead of requiring a three-body collision for recombination, only a two-body collision between the solid particle and positive ion need occur. To a first approximation the particle electron recombination rate can be equated to the rate at which positive ions strike the solid particles. However, if negative ions are present (for typical solid propellants mole fractions of Cl^- are from 2 to 3 orders of magnitude greater than electron mole fractions) the electron recombination rate will decrease, since some of the positive ions will react with negative ions rather than electrons.

The solid particle electron recombination coefficient, α_{pe} , is defined from

$$\left(\frac{dn_e}{dt}\right)_p = -\alpha_{pe} n_+ n_p \quad (71)$$

where n_e is electron density, t is time, n_+ is positive ion density and n_p is the particle number density. If particle diameters are small compared to the gas mean free path[†] (free molecular flow) and the electron densities are sufficiently high that the particles remain negatively charged, α_{pe} is essentially the random ion flux to the particle (with a correction factor for negative ions),

$$\alpha_{pe} = \pi r^2 \left(\frac{8kT}{\pi m_+}\right)^{1/2} \left[1 + \left(\frac{n_-}{n_e}\right) \left(\frac{m_e}{m_-}\right)^{1/2}\right]^{-1} \quad (72)$$

[†] If particle diameters are not small compared to the mean free path corrections will have to be made to the expression for α_{pe} .

24. Calcote, H. F., Kurzius, S. C., and Silla, H., "Solid Propellant Flame Ionization and the Effect of Chemical Additives," Third Radar Attenuation Symposium, CPLA Publ. No. 46 (Applied Physics Lab., Johns Hopkins Univ., Silver Spring, 1964), pp. 17-40.

where r is the particle radius, k is the Boltzmann constant, T is the gas temperature, m_+ , m_e , and m_- are the masses of positive ions, electrons and negative ions, respectively, and n_- is the negative ion density.

Equations (71) and (72) represent the formal method for incorporating heterogeneous electron/ion recombination into FULLNOZ. However, we must also account for the possibility that, at the high temperatures near the nozzle throat, the particle is emitting electrons via thermionic emission. Under these conditions a steady state is achieved by balancing the random current density to the particle (from the plasma) by thermionic emission. Equating these electron currents results in a "critical" electron density for which the net current flow to the particle is zero and an initially neutral particle will remain neutral. This critical electron density is defined by,

$$(n_e)_{cr} = B \left(\frac{T_p}{T_g} \right) T_g^{3/2} \exp \left[-11,605 E_w / T_p \right] \quad (73)$$

where B is the thermionic emission constant, T_p is the particle temperature, T_g is the gas temperature and E_w is the effective work function of the particle in volts. Thus, heterogeneous electron/ion recombination is only of potential importance for $n_e > (n_e)_{cr}$.

The technique adopted to incorporate the above equations into FULLNOZ is:

1. At each integration step determine whether the local electron density is greater than the critical electron density. (During the initial stages of the expansion process, where particle temperatures are very high it is likely that $n_e < (n_e)_{cr}$.)
2. If $n_e < (n_e)_{cr}$ then electron/ion recombination will not be significant and Eqs. (71) and (72) will not be employed.
3. If $n_e > (n_e)_{cr}$ Eqs. (71) and (72) will be incorporated directly into the general kinetic scheme (with possible corrections to Eq. (72) due to non-free molecular flow effects).

The above procedure has not as yet been incorporated into the code.

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APPENDIX A

SAMPLE INPUT DATA

Obtained
FULLNOZ

TURNING POINT

1 14

CARD NO

1
2 *** FULLY COUPLED NOZZLE FLOW PROGRAM - FULLNOZ - SAMPLE TEST CASE 9/75 ***
3 7200 2.000E+2 1.524E+2
4 1 1 0 1 0
5 25 6 10 2 22 10 50 900 10
6 1000 1 1
7 0.8 0.01 1.0E-4 1.0 0.1
8 11.64108562 E+1 1.8575287 E+1 44865856E+0 36280951 E-1 CO 1
2 -598.1864 E-2 28443584E+2 544.10327 E+2 CO 2
8 21.58741331E+1 1.13592404E+02 -86056609E1 21500497E+1 CO2 1
2 -30567063E+1 -96439872E+2 54321459E+2 CO2 2
8 31.49683629 E+1 -23301268E-3 59667528 E-4 -51872545E-5 H 1
2 -18467323E-4 50618655 E+2 33404654 E+2 H 2
8 41.60596451 E+1 13412152 E+1 -16508752E+0 88760457 E-2 H2 1
2 46616242 E-1 -17005029E+1 38443271 E+2 H2 2
8 51.58818265 E+1 17395375 E+1 -32536218E+0 22313547 E-1 OH 1
2 75177541 E-1 79212009 E+1 50938996 E+2 OH 2
8 61.60311014 E+1 47049758 E+1 -96291245E+0 69087996 E-1 H2O 1
2 60421152 E-1 -59593534E+2 51387265 E+2 H2O 2

Obtained
FULLNOZ

2 14

CARD NO

8 71.62643469 E+1 19328092 E+1 -46158274E+0 37083135 E-1 N2 1
2 91852768 E-2 19242408E+1 52824774 E+2 N2 2
8 81.50617760 E+1 -11736656E+0 39405358 E-1 -20531577E-2 O 1
2 18127099 E-1 58115304 E+2 44728172 E+2 O 2
8 91.72417589 E+1 12398122 E+1 -18448469E+0 10507580 E-1 O2 1
2 -59604067E-1 -24219968E+1 57031740 E+2 O2 2
8 101.48485464 E+1 32006164 E+0 -28545471E+0 83218696 E-1 K 1
2 43725018 E-2 19867370 E+2 44106314 E+2 K 2

91 CO	139. E-6	300.	0.75	1.4	1.4	28.
2 CO2	139. E-6	300.	0.75	1.4	1.4	44.
3 H	83.5 E-6	300.	0.75	1.4	1.4	1.
4 H2	83.5 E-6	300.	0.75	1.4	1.4	2.
5 OH	125.5 E-6	300.	0.75	1.4	1.4	17.
6 H2O	125.5 E-6	300.	0.75	1.4	1.4	18.
7 N2	160. E-6	300.	0.75	1.4	1.4	28.
8 O	166. E-6	300.	0.75	1.4	1.4	16.
9 O2	189. E-6	300.	0.75	1.4	1.4	32.
10 K	129.7 E-6	300.	0.75	1.4	1.4	39.1

~NOTE~ CARD 10 OMITTED FOR ITURB=0 (CARD 4, COL 56-60)



FORTRAN Coding Form

FULLNOZ												3	14
CARD NO	FORTRAN STATEMENT												
11	12.668	0.0	0.0										
12.11	12.663	0.6053	0.01804	8.500	2761.	1.534E+5							
2	3.0E-1	6.0E-2	6.3E-4	2.4E-2	2.6E-3	1.9E-1	4.2E-1	7.8E-5					
3	5.6E-5	1.1E-6											
12.21	12.646	1.2110	0.03607	8.473	2759.	1.534E+5							
2	NOTE CARDS 12.22 TO 12.24.2 ARE IDENTICAL TO CARD 12.1.2												
3	CARDS 12.23 TO 12.24.3 ARE IDENTICAL TO CARD 12.1.3												
12.31	12.619	1.8158	0.05411	8.443	2758.	1.534E+5							
2	SEE NOTE												
3													
12.41	12.581	2.4199	0.07214	8.400	2757.	1.536E+5							
2	SEE NOTE												
3													
12.51	12.532	3.0238	0.09018	8.343	2755.	1.538E+5							
2	SEE NOTE												
3													
12.61	12.472	3.6262	0.10821	8.273	2751.	1.541E+5							
2	SEE NOTE												
3													



FORTRAN Coding Form

FULLNOZ												4	14
CARD NO	FORTRAN STATEMENT												
1271 3	12.401 SEE NOTE	4.2274	0.12625	8.193	2745.	1.555E+5							
1281 2 3	12.319 SEE NOTE	4.8278	0.14428	8.103	2739.	1.550E+5							
1291 2 3	12.227 SEE NOTE	5.4260	0.16232	8.003	2732.	1.556E+5							
1201 2 3	12.123 SEE NOTE	6.0225	0.18035	7.895	2726.	1.562E+5							
1211 2 3	12.009 SEE NOTE	6.6176	0.19839	7.775	2719.	1.570E+5							
1221 2 3	11.885 SEE NOTE	7.2100	0.21642	7.643	2712.	1.578E+5							



FORTRAN Coding Form

FULL NOZ

5 14

CARD
NO

FORTRAN STATEMENT

12131	11.749	7.8000	0.23446	7.498	2708.	1.588E+5
2	SEE NOTE					
3						
12141	11.603	8.3880	0.25249	7.328	2700.	1.600E+5
2	SEE NOTE					
3						
12151	11.447	8.9728	0.27053	7.138	2688.	1.613E+5
2	SEE NOTE					
3						
12161	11.280	9.5546	0.28856	6.935	2671.	1.629E+5
2	SEE NOTE					
3						
12171	11.102	10.1340	0.30660	6.715	2648.	1.647E+5
2	SEE NOTE					
3						
12181	10.914	10.7090	0.32463	6.480	2624.	1.670E+5
2	SEE NOTE					
3						



FORTRAN Coding Form

FULL NOZ

6 14

CARD
NO

FORTRAN STATEMENT

12191	10.716	11.2810	0.34267	6.225	2594.	1.676E+5
2	SEE NOTE					
3						
12201	10.507	11.8500	0.36070	5.950	2563.	1.725E+5
2	SEE NOTE					
3						
12211	10.289	12.4150	0.37874	5.680	2550.	1.761E+5
2	SEE NOTE					
3						
12221	10.060	12.9750	0.39677	5.390	2533.	1.798E+5
2	SEE NOTE					
3						
12231	9.8200	13.5300	0.41481	5.065	2502.	1.847E+5
2	SEE NOTE					
3						
12241	9.5715	14.0840	0.43284	4.602	2401.	1.904E+5
2	SEE NOTE					
3						

FULLNOZ

7 14

CARD NO	CONTRIBUTION STATEMENT			
13.1	12.668	0.0		
2.	1.0E+10	0.0		
14.1	9.5715	14.084		
2.	9.957	14.275		
3.	10.312	14.503		
4.	11.582	15.215		
5.	12.852	15.850		
6.	14.122	16.434		
7.	16.662	17.729		
8.	19.202	18.999		
9.	21.742	20.218		
10.	26.822	22.606		
11.	31.902	24.994		
12.	36.982	27.305		
13.	44.602	30.683		
14.	52.222	33.985		
15.	59.842	37.084		
16.	73.914	42.062		
17.	87.173	46.838		
18.	99.035	50.719		
19.	111.150	54.508		
20.	123.490	58.039		
21.	134.420	60.909		
22.	152.400	65.634		



FULLNOZ

014

CARD NO							
15.	1.0	1.0	0.339	0.323	255.	102.	
16.	4.0			2320.			
17.	4.	1.					
18.	19.	19.	19.	19.			
19.	1.0E-4	2.0E-4	3.0E-4	4.0E-4			

$NC(CARD\ 17, COL. 10) = 1$, THEREFORE CARD NO 25 IS NEXT

25.1	0.147	E+6	0.139	E+6	0.131	E+6	0.124	E+6
26.1	-0.276	E+3	-0.559	E+3	-0.768	E+3	-0.101	E+4
27.1	2813.		2893		2913		2973	
28.1	0.279	E-4	0.143	E-3	0.182	E-3	0.554	E-4
29.1	0		0		0			
30.1	0.0		0.0		0.0		0.0	0.0

25.2 0 147 E+6 0 139 L+6 0 132 E+6 0 125 E+6
26.2 -0 292 E+3 0 592 E+3 0 814 E+3 0 107 L+4
27.2 2813 2893 2913 2973
28.2 0 275 E-4 0 143 E-3 0 182 E-3 0 554 E-4
29.2 CARDS 29.2 TO 29.18 ARE IDENTICAL
30.2 NOTE CARDS 30.2 TO 30.18 ARE IDENTICAL

25.3	0.148	E+6	0.139	E+6	0.132	E+6	0.125	E+6
26.3	0.310	E+3	0.625	E+3	0.864	E+3	0.114	E+4

[illegible]

CARD NO	FORTRAN STATEMENT							
273	2813.	2893.	2913.	2973.				
283	0.279 E-4	0.143 E-3	0.182 E-3	0.554 E-4				
293								
303								
254	0.148 E+6	0.139 E+6	0.132 E+6	0.125 E+6				
264	-0.831 E+3	-0.671 E+3	-0.922 E+3	-0.121 E+4				
274	2813.	2893.	2913.	2973.				
284	0.279 E-4	0.143 E-3	0.182 E-3	0.554 E-4				
294								
304								
255	0.148 E+6	0.140 E+6	0.132 E+6	0.125 E+6				
265	-0.386 E+3	-0.719 E+3	-0.989 E+3	-0.130 E+4				
275	2813.	2893.	2913.	2973.				
285	0.279 E-4	0.143 E-3	0.182 E-3	0.554 E-4				
295								
305								
256	0.148 E+6	0.140 E+6	0.132 E+6	0.125 E+6				
266	-0.382 E+3	-0.775 E+3	-0.106 E+4	-0.140 E+4				
276	2813.	2893.	2913.	2973.				
286	0.279 E-4	0.143 E-3	0.182 E-3	0.554 E-4				
296								



10 14

CARD		FORTRAN STATEMENT	
NO			
30.6			
25.7	0.149 E+6	0.140 E+6	0.132 E+6
26.7	0.414 E+3	0.839 E+3	0.115 E+4
27.7	2803.	2883.	2913.
28.7	0.279 E-4	0.143 E-3	0.182 E-3
29.7			0.554 E-4
30.7			
25.8	0.149 E+6	0.141 E+6	0.133 E+6
26.8	0.451 E+3	0.915 E+3	0.126 E+4
27.8	2803.	2883.	2913.
28.8	0.279 E-4	0.143 E-3	0.182 E-3
29.8			0.554 E-4
30.8			
25			
25.9	0.150 E+6	0.141 E+6	0.133 E+6
26.9	0.497 E+3	0.101 E+4	0.138 E+4
27.9	2803.	2883.	2913.
28.9	0.275 E-4	0.143 E-3	0.182 E-3
29.9			0.554 E-4
30.9			



FORTRAN Coding Form

FULLNOZ

11 14

CARD
NO

FORTRAN STATEMENT

25.10	0.150 E+6	0.142 E+6	0.134 E+6	0.126 E+6
26.10	0.552 E+3	0.112 E+4	0.154 E+4	0.202 E+4
27.10	2793.	2873.	2903.	2933.
28.10	0.279 E-4	0.143 E-3	0.182 E-3	0.554 E-4
29.10				
30.10				
25.11	0.151 E+6	0.143 E+6	0.134 E+6	0.127 E+6
26.11	0.621 E+3	0.126 E+4	0.173 E+4	0.27 E+4
27.11	2793.	2873.	2903.	
28.11	0.279 E-4	0.143 E-3	0.182 E-3	0.554 E-4
29.11				
30.11				
25.12	0.152 E+6	0.143 E+6	0.135 E+6	0.127 E+6
26.12	0.709 E+3	0.144 E+4	0.198 E+4	0.260 E+4
27.12	2783.	2873.	2893.	2943.
28.12	0.279 E-4	0.143 E-3	0.182 E-3	0.554 E-4
29.12				
30.12				
25.13	0.153 E+6	0.144 E+6	0.136 E+6	0.128 E+6
26.13	0.828 E+3	0.168 E+4	0.231 E+4	0.303 E+4
27.13	2783.	2873.	2893.	2943.



FORTRAN Coding Form

FULLNOZ

12 14

CARD
NO

FORTRAN STATEMENT

28.13	0.279 E-4	0.143 E-3	0.182 E-3	0.554 E-4
29.13				
30.13				
25.14	0.154 E+6	0.145 E+6	0.136 E+6	0.128 E+6
26.14	0.993 E+3	0.201 E+4	0.277 E+4	0.363 E+4
27.14	2763.	2853.	2883.	2933.
28.14	0.279 E-4	0.143 E-3	0.182 E-3	0.554 E-4
29.14				
30.14				
25.15	0.155 E+6	0.146 E+6	0.137 E+6	0.129 E+6
26.15	0.124 E+4	0.252 E+4	0.346 E+4	0.454 E+4
27.15	2763.	2853.	2883.	2933.
28.15	0.279 E-4	0.143 E-3	0.182 E-3	0.554 E-4
29.15				
30.15				
25.16	0.156 E+6	0.147 E+6	0.138 E+6	0.130 E+6
26.16	0.166 E+4	0.336 E+4	0.461 E+4	0.606 E+4
27.16	2743.	2823.	2873.	2923.
28.16	0.279 E-4	0.143 E-3	0.182 E-3	0.554 E-4
29.16				
30.16				



FORTRAN Coding Form

FULL NO. 2				13 14													
				FORTRAN STATEMENT													
0.159	E+6	0.142	E+6	0.139	E+6	0.131	E+6										
-0.280	E+4	-0.504	E+4	-0.692	E+4	-0.909	E+4										
2743.		2823.		2873.		2923.											
0.279	E-4	0.143	E-3	0.182	E-3	0.554	E-4										
0.159	E+6	0.140	E+6	0.140	E+6	0.132	E+6										
-0.497	E+4	-0.101	E+5	-0.138	E+5	-0.182	E+5										
2713.		2793.		2853.		2903.											
0.279	E-4	0.143	E-3	0.182	E-3	0.554	E-4										



FORTRAN Coding Form

FULL NO. 2								14 15			
CARD NO.	FORTRAN STATEMENT										
33.10	+0	+M	=D2	+M				21 1 0E-29	0		
.20	+H	+M	=OH	+M				22 1 0E-29	1.0		
.3M	+H	+M	=H2	+M				22 5 0E-29	1.0		
.4M	+OH	+M	=H20	+M				22 2 0E-28	1.0		
.5C.0	+0	+M	=CO2	+M				26 1 0E-29	1.0	-2500	
.6DH	+OH		=OH2	+0				15 1 0E-11		-780	
.7DH	+H2		=H20	+H				15 3 6E-11		-5200	
.80	+H2		=OH	+H				15 2 9E-11		-9460	
.9M	+02		=OH	+0				15 3 7E-10		-16800	
.10C.0	+OH		=CO2	+H				15 9 0E-13		-1080	

APPENDIX B

SAMPLE OUTPUT

B-i

FULLY COUPLED NOZZLE FLUX PROGRAM

AERUCHEM RESEARCH LABORATORIES, INC.
PRINCETON, NEW JERSEY 08540

*** FULLY COUPLED NOZZLE FLUX PROGRAM - FULLNOZ - SAMPLE TEST CASE 9/75 ***

```

11. IN DATA SETS 1
NSET= 7200  VALSS= 7.000E+02  SLABS= 1.520E+02  SLUGS=0.  IPUS= 0  NPTS= 0

NMAX  N  NP  IP  IYPE  IKN0  NMAX  NMAX  NUS  IPO  NLEN  IUFF  LPLAN
25  0  10  0  1  1  2  22  10  -0  50  0  900
IPIN  IPC  IMCH  IMU0  ITO  ICSN  ITAPD  IAME  IPAN  IN  ITUN  IPTU
0  0  0  0  -0  -0  -0  10  1  -0  0  -0

METAP  N  NLEN
1.0130E+00  0.1020E+01  1.9071E+00

ALPHAN  EMBLON  ICL  DELTA  ATOL  FSTEP  GRAD  FRAC  FRACTN
0.00  0.00  0.00  1.0  1.00  -0.00  -0.00  -0.00  -0.00

IHE=1  -0.0  11  UMLA  PM  SC  4A
CU  1.340E+00  3.000E+02  7.500E-01  1.400E+00  1.400E+00  2.000E+01
CU2  1.340E+00  3.000E+02  7.500E-01  1.400E+00  1.400E+00  4.000E+01
M  0.350E+05  1.000E+02  7.500E-01  1.400E+00  1.400E+00  1.000E+00
M2  0.350E+05  1.000E+02  7.500E-01  1.400E+00  1.400E+00  2.000E+00
UM  1.250E+00  1.000E+02  7.500E-01  1.400E+00  1.400E+00  1.700E+01
M20  1.250E+00  3.000E+02  7.500E-01  1.400E+00  1.400E+00  1.000E+01
M2  1.000E+00  1.000E+02  7.500E-01  1.400E+00  1.400E+00  2.000E+01
U  1.000E+00  3.000E+02  7.500E-01  1.400E+00  1.400E+00  1.000E+01
U2  1.000E+00  3.000E+02  7.500E-01  1.400E+00  1.400E+00  3.000E+01
K  1.297E+00  3.000E+02  7.500E-01  1.400E+00  1.400E+00  4.000E+01

K  A  N  U  PM1  P  I  H
1  1.207E+01  0.  0.  0.  0.  0.
2  1.200E+01  0.053E+01  1.000E+02  0.500E+00  2.701E+03  1.530E+05
3  1.205E+01  1.211E+00  3.007E+02  0.473E+00  2.750E+03  1.530E+05
4  1.202E+01  1.010E+00  5.011E+02  0.400E+00  2.750E+03  1.535E+05
5  1.250E+01  0.400E+00  7.010E+02  0.400E+00  2.757E+03  1.537E+05
6  1.255E+01  3.000E+00  9.010E+02  0.300E+00  2.755E+03  1.530E+05
7  1.247E+01  3.000E+00  1.002E+03  0.273E+00  2.751E+03  1.541E+05
8  1.240E+01  4.227E+00  1.203E+03  0.193E+00  2.705E+03  1.540E+05
9  1.232E+01  0.420E+00  1.403E+03  0.103E+00  2.730E+03  1.551E+05
10  1.225E+01  3.000E+00  1.603E+03  0.003E+00  2.732E+03  1.550E+05
11  1.212E+01  0.003E+00  1.800E+03  7.000E-01  2.720E+03  1.560E+05
12  1.201E+01  0.010E+00  1.900E+03  7.775E+00  2.719E+03  1.571E+05
13  1.180E+01  7.010E+00  2.100E+03  7.000E+00  2.712E+03  1.570E+05
14  1.175E+01  7.000E+00  2.305E+03  7.000E+00  2.700E+03  1.580E+05
15  1.100E+01  0.300E+00  2.525E+03  7.320E+00  2.700E+03  1.600E+05
16  1.105E+01  0.973E+00  2.705E+03  7.130E+00  2.600E+03  1.610E+05
17  1.120E+01  0.555E+00  2.800E+03  0.935E+00  2.671E+03  1.630E+05
18  1.110E+01  1.013E+01  3.000E+03  0.715E+00  2.600E+03  1.640E+05
19  1.091E+01  1.071E+01  3.200E+03  0.400E+00  2.604E+03  1.671E+05
20  1.072E+01  1.120E+03  3.427E+03  0.250E+00  2.590E+03  1.670E+05
21  1.051E+01  1.105E+03  3.607E+03  0.050E+00  2.563E+03  1.720E+05
22  1.029E+01  1.042E+03  3.707E+03  0.000E+00  2.550E+03  1.702E+05
23  1.000E+01  1.000E+03  3.900E+03  0.300E+00  2.533E+03  1.700E+05
24  0.921E+00  1.153E+03  4.100E+03  0.000E+00  2.502E+03  1.807E+05
25  0.572E+00  1.000E+03  4.300E+03  0.002E+00  2.001E+03  1.900E+05

I  K  N  N  PM1A  PM  SN
1  1.207E+01  0.  0.  0.  0.
2  1.000E+01  -0.  0.  0.  0.

A  A0  N  N  PM1h  PM  SN
1  0.572E+00  1.400E+03  0.  0.  0.
2  0.457E+00  1.400E+03  0.  0.  0.
3  1.031E+01  1.050E+03  0.  0.  0.
4  1.150E+01  1.522E+03  0.  0.  0.
5  1.205E+01  1.505E+03  0.  0.  0.
6  1.412E+01  1.603E+03  0.  0.  0.
7  1.600E+01  1.773E+03  0.  0.  0.
8  1.900E+01  1.900E+03  0.  0.  0.
9  2.170E+01  2.020E+03  0.  0.  0.
10  2.600E+01  2.201E+03  0.  0.  0.
11  3.100E+01  2.400E+03  0.  0.  0.
12  3.600E+01  2.731E+03  0.  0.  0.
13  4.000E+01  3.000E+03  0.  0.  0.
14  5.022E+01  3.300E+03  0.  0.  0.
15  5.900E+01  3.700E+03  0.  0.  0.
16  7.300E+01  4.000E+03  0.  0.  0.
17  0.717E+01  4.000E+03  0.  0.  0.
18  0.400E+01  5.075E+03  0.  0.  0.
19  1.112E+02  5.451E+03  0.  0.  0.
20  1.255E+02  5.000E+03  0.  0.  0.
21  1.300E+02  0.000E+01  0.  0.  0.
22  1.500E+02  0.500E+01  0.  0.  0.

```

B - 2

8 AS 1.2401E+01 MS 2.2270E+00 PHIS 1.2425E-01 IS 2.7030E+03 PS 0.1930E+00 US 1.5450E+05
 MAS .1271E+01 DELTA .0050E+00 MS .0000E+02 MTS .3040E+03 TAMS 0. UM 0.
 PIS .1070E+02 MMUS .7210E-03 SUM 0. SUMDUTS .0541E+04

SPCLIC	CU	CS	.3000E+00	XS	.2125E+00	ADUTS	0.
SPCLIC	CU2	CS	.0000E+01	XS	.2700E-01	ADUTS	0.
SPCLIC	M	CS	.0300E-03	XS	.1249E-01	ADUTS	0.
SPCLIC	M2	CS	.2400E-01	XS	.2300E+00	ADUTS	0.
SPCLIC	UM	CS	.2000E-02	XS	.5033E-02	ADUTS	0.
SPCLIC	M2U	CS	.1900E+00	XS	.2093E+00	ADUTS	0.
SPCLIC	M2	CS	.0200E+00	XS	.2975E+00	ADUTS	0.
SPCLIC	U	CS	.7000E-04	XS	.9000E-04	ADUTS	0.
SPCLIC	U2	CS	.5000E-04	XS	.3471E-04	ADUTS	0.
SPCLIC	A	CS	.1100E-05	XS	.5454E-06	ADUTS	0.

9 AS 1.2519E+01 MS 2.2270E+00 PHIS 1.2425E-01 IS 2.7390E+03 PS 0.1930E+00 US 1.5505E+05
 MAS .1270E+01 DELTA .0000E+00 MS .0550E+02 MTS .3030E+03 TAMS 0. UM 0.
 PIS .1050E+02 MMUS .7152E-03 SUM 0. SUMDUTS .0253E+04

SPCLIC	CU	CS	.3000E+00	XS	.2125E+00	ADUTS	0.
SPCLIC	CU2	CS	.0000E+01	XS	.2700E-01	ADUTS	0.
SPCLIC	M	CS	.0300E-03	XS	.1249E-01	ADUTS	0.
SPCLIC	M2	CS	.2400E-01	XS	.2300E+00	ADUTS	0.
SPCLIC	UM	CS	.2000E-02	XS	.5033E-02	ADUTS	0.
SPCLIC	M2U	CS	.1900E+00	XS	.2093E+00	ADUTS	0.
SPCLIC	M2	CS	.0200E+00	XS	.2975E+00	ADUTS	0.
SPCLIC	U	CS	.7000E-04	XS	.9000E-04	ADUTS	0.
SPCLIC	U2	CS	.5000E-04	XS	.3471E-04	ADUTS	0.
SPCLIC	A	CS	.1100E-05	XS	.5454E-06	ADUTS	0.

10 AS 1.2727E+01 MS 2.2270E+00 PHIS 1.2425E-01 IS 2.7320E+03 PS 0.0030E+00 US 1.5461E+05
 MAS .1302E+01 DELTA .0050E+00 MS .0402E+02 MTS .3015E+03 TAMS 0. UM 0.
 PIS .1047E+02 MMUS .7062E-03 SUM 0. SUMDUTS .1000E+05

SPCLIC	CU	CS	.3000E+00	XS	.2125E+00	ADUTS	0.
SPCLIC	CU2	CS	.0000E+01	XS	.2700E-01	ADUTS	0.
SPCLIC	M	CS	.0300E-03	XS	.1249E-01	ADUTS	0.
SPCLIC	M2	CS	.2400E-01	XS	.2300E+00	ADUTS	0.
SPCLIC	UM	CS	.2000E-02	XS	.5033E-02	ADUTS	0.
SPCLIC	M2U	CS	.1900E+00	XS	.2093E+00	ADUTS	0.
SPCLIC	M2	CS	.0200E+00	XS	.2975E+00	ADUTS	0.
SPCLIC	U	CS	.7000E-04	XS	.9000E-04	ADUTS	0.
SPCLIC	U2	CS	.5000E-04	XS	.3471E-04	ADUTS	0.
SPCLIC	A	CS	.1100E-05	XS	.5454E-06	ADUTS	0.

11 AS 1.2123E+01 MS 2.2270E+00 PHIS 1.2403E-01 IS 2.7260E+03 PS 7.8950E+00 US 1.5620E+05
 MAS .1309E+01 DELTA .0050E+00 MS .0099E+02 MTS .3010E+03 TAMS 0. UM 0.
 PIS .1054E+02 MMUS .7002E-03 SUM 0. SUMDUTS .1270E+05

SPCLIC	CU	CS	.3000E+00	XS	.2125E+00	ADUTS	0.
SPCLIC	CU2	CS	.0000E+01	XS	.2700E-01	ADUTS	0.
SPCLIC	M	CS	.0300E-03	XS	.1249E-01	ADUTS	0.
SPCLIC	M2	CS	.2400E-01	XS	.2300E+00	ADUTS	0.
SPCLIC	UM	CS	.2000E-02	XS	.5033E-02	ADUTS	0.
SPCLIC	M2U	CS	.1900E+00	XS	.2093E+00	ADUTS	0.
SPCLIC	M2	CS	.0200E+00	XS	.2975E+00	ADUTS	0.
SPCLIC	U	CS	.7000E-04	XS	.9000E-04	ADUTS	0.
SPCLIC	U2	CS	.5000E-04	XS	.3471E-04	ADUTS	0.
SPCLIC	A	CS	.1100E-05	XS	.5454E-06	ADUTS	0.

12 AS 1.2009E+01 MS 2.2100E+00 PHIS 1.2403E-01 IS 2.7190E+03 PS 7.7750E+00 US 1.5707E+05
 MAS .1310E+01 DELTA .0050E+00 MS .0545E+02 MTS .3004E+03 TAMS 0. UM 0.
 PIS .1019E+02 MMUS .6913E-03 SUM 0. SUMDUTS .1500E+05

SPCLIC	CU	CS	.3000E+00	XS	.2125E+00	ADUTS	0.
SPCLIC	CU2	CS	.0000E+01	XS	.2700E-01	ADUTS	0.
SPCLIC	M	CS	.0300E-03	XS	.1249E-01	ADUTS	0.
SPCLIC	M2	CS	.2400E-01	XS	.2300E+00	ADUTS	0.
SPCLIC	UM	CS	.2000E-02	XS	.5033E-02	ADUTS	0.
SPCLIC	M2U	CS	.1900E+00	XS	.2093E+00	ADUTS	0.
SPCLIC	M2	CS	.0200E+00	XS	.2975E+00	ADUTS	0.
SPCLIC	U	CS	.7000E-04	XS	.9000E-04	ADUTS	0.
SPCLIC	U2	CS	.5000E-04	XS	.3471E-04	ADUTS	0.
SPCLIC	A	CS	.1100E-05	XS	.5454E-06	ADUTS	0.

13 AS 1.1803E+01 MS 2.2100E+00 PHIS 2.1042E-01 IS 2.7120E+03 PS 7.6630E+00 US 1.5709E+05
 MAS .1320E+01 DELTA .0050E+00 MS .0192E+02 MTS .3000E+03 TAMS 0. UM 0.
 PIS .1004E+02 MMUS .6813E-03 SUM 0. SUMDUTS .1823E+05

SPCLIC	CU	CS	.3000E+00	XS	.2125E+00	ADUTS	0.
SPCLIC	CU2	CS	.0000E+01	XS	.2700E-01	ADUTS	0.
SPCLIC	M	CS	.0300E-03	XS	.1249E-01	ADUTS	0.
SPCLIC	M2	CS	.2400E-01	XS	.2300E+00	ADUTS	0.
SPCLIC	UM	CS	.2000E-02	XS	.5033E-02	ADUTS	0.
SPCLIC	M2U	CS	.1900E+00	XS	.2093E+00	ADUTS	0.
SPCLIC	M2	CS	.0200E+00	XS	.2975E+00	ADUTS	0.
SPCLIC	U	CS	.7000E-04	XS	.9000E-04	ADUTS	0.
SPCLIC	U2	CS	.5000E-04	XS	.3471E-04	ADUTS	0.
SPCLIC	A	CS	.1100E-05	XS	.5454E-06	ADUTS	0.

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14 X 1.1749E+01 M 7.0000E+00 PHI 2.5000E-01 T 2.7000E+03 P 7.0700E+00 U 1.5000E+05
 M 1.1335E+01 DELTA .0050E+00 M .7000E+02 N 1.3010E+03 T 0. T 0.
 P 1.1500E+02 MM 0.0000E+03 S 0. SUMU 2.1700E+05

SPCLC LU C .3000E+00 X .2125E+00 N 0.
 SPCLC LU2 C .0000E+01 X .2700E+01 N 0.
 SPCLC M C .0300E+03 X .1200E+01 N 0.
 SPCLC M2 C .2000E+01 X .2300E+00 N 0.
 SPCLC UM C .2000E+02 X .3033E+02 N 0.
 SPCLC MU C .1900E+00 X .2093E+00 N 0.
 SPCLC M2 C .0200E+00 X .2075E+00 N 0.
 SPCLC U C .7000E+00 X .9000E+00 N 0.
 SPCLC U2 C .5000E+00 X .3071E+00 N 0.
 SPCLC N C .1100E+05 X .5050E+00 N 0.

15 X 1.1000E+01 M 0.3000E+00 PHI 2.5000E-01 T 2.7000E+03 P 7.1200E+00 U 1.0000E+05
 M 1.1547E+01 DELTA .0050E+00 M .7500E+02 N 1.3010E+03 T 0. T 0.
 P 1.1500E+02 MM 0.0000E+03 S 0. SUMU 2.0000E+05

SPCLC LU C .3000E+00 X .2125E+00 N 0.
 SPCLC LU2 C .0000E+01 X .2700E+01 N 0.
 SPCLC M C .0300E+03 X .1200E+01 N 0.
 SPCLC M2 C .2000E+01 X .2300E+00 N 0.
 SPCLC UM C .2000E+02 X .3033E+02 N 0.
 SPCLC MU C .1900E+00 X .2093E+00 N 0.
 SPCLC M2 C .0200E+00 X .2075E+00 N 0.
 SPCLC U C .7000E+00 X .9000E+00 N 0.
 SPCLC U2 C .5000E+00 X .3071E+00 N 0.
 SPCLC N C .1100E+05 X .5050E+00 N 0.

16 X 1.1000E+01 M 0.3000E+00 PHI 2.5000E-01 T 2.0000E+03 P 7.1500E+00 U 1.0137E+05
 M 1.1547E+01 DELTA .0050E+00 M .0000E+02 N 1.3010E+03 T 0. T 0.
 P 1.1500E+02 MM 0.0000E+03 S 0. SUMU 2.7000E+05

SPCLC LU C .3000E+00 X .2125E+00 N 0.
 SPCLC LU2 C .0000E+01 X .2700E+01 N 0.
 SPCLC M C .0300E+03 X .1200E+01 N 0.
 SPCLC M2 C .2000E+01 X .2300E+00 N 0.
 SPCLC UM C .2000E+02 X .3033E+02 N 0.
 SPCLC MU C .1900E+00 X .2093E+00 N 0.
 SPCLC M2 C .0200E+00 X .2075E+00 N 0.
 SPCLC U C .7000E+00 X .9000E+00 N 0.
 SPCLC U2 C .5000E+00 X .3071E+00 N 0.
 SPCLC N C .1100E+05 X .5050E+00 N 0.

17 X 1.1000E+01 M 0.3000E+00 PHI 2.0000E-01 T 2.0710E+03 P 0.9300E+00 U 1.0290E+05
 M 1.1547E+01 DELTA .0050E+00 M .0120E+02 N 1.7000E+03 T 0. T 0.
 P 1.1500E+02 MM 0.0000E+03 S 0. SUMU 3.1500E+05

SPCLC LU C .3000E+00 X .2125E+00 N 0.
 SPCLC LU2 C .0000E+01 X .2700E+01 N 0.
 SPCLC M C .0300E+03 X .1200E+01 N 0.
 SPCLC M2 C .2000E+01 X .2300E+00 N 0.
 SPCLC UM C .2000E+02 X .3033E+02 N 0.
 SPCLC MU C .1900E+00 X .2093E+00 N 0.
 SPCLC M2 C .0200E+00 X .2075E+00 N 0.
 SPCLC U C .7000E+00 X .9000E+00 N 0.
 SPCLC U2 C .5000E+00 X .3071E+00 N 0.
 SPCLC N C .1100E+05 X .5050E+00 N 0.

18 X 1.1000E+01 M 1.0130E+01 PHI 3.0000E-01 T 2.0000E+03 P 0.7150E+00 U 1.0070E+05
 M 1.1547E+01 DELTA .0000E+00 M .0071E+02 N 1.3700E+03 T 0. T 0.
 P 1.1500E+02 MM 0.0000E+03 S 0. SUMU 3.5310E+05

SPCLC LU C .3000E+00 X .2125E+00 N 0.
 SPCLC LU2 C .0000E+01 X .2700E+01 N 0.
 SPCLC M C .0300E+03 X .1200E+01 N 0.
 SPCLC M2 C .2000E+01 X .2300E+00 N 0.
 SPCLC UM C .2000E+02 X .3033E+02 N 0.
 SPCLC MU C .1900E+00 X .2093E+00 N 0.
 SPCLC M2 C .0200E+00 X .2075E+00 N 0.
 SPCLC U C .7000E+00 X .9000E+00 N 0.
 SPCLC U2 C .5000E+00 X .3071E+00 N 0.
 SPCLC N C .1100E+05 X .5050E+00 N 0.

19 X 1.0000E+01 M 1.0700E+01 PHI 3.0000E-01 T 2.0000E+03 P 0.0000E+00 U 1.0700E+05
 M 1.1547E+01 DELTA .0050E+00 M .3707E+02 N 1.3700E+03 T 0. T 0.
 P 1.1500E+02 MM 0.0000E+03 S 0. SUMU 3.0000E+05

SPCLC LU C .3000E+00 X .2125E+00 N 0.
 SPCLC LU2 C .0000E+01 X .2700E+01 N 0.
 SPCLC M C .0300E+03 X .1200E+01 N 0.
 SPCLC M2 C .2000E+01 X .2300E+00 N 0.
 SPCLC UM C .2000E+02 X .3033E+02 N 0.
 SPCLC MU C .1900E+00 X .2093E+00 N 0.
 SPCLC M2 C .0200E+00 X .2075E+00 N 0.
 SPCLC U C .7000E+00 X .9000E+00 N 0.
 SPCLC U2 C .5000E+00 X .3071E+00 N 0.
 SPCLC N C .1100E+05 X .5050E+00 N 0.

20 AS 1.0716E+01 MS 1.1201E+01 PHIS 3.4207E-01 IS 2.5900E+03 PS 6.2050E+00 US 1.0702E+01
 WFLY .0037E+00 DELTA .0002E+00 MS .7207E+02 HTS .9500E+03 TANS 0. US 0.
 PIS .1027E+02 MMUS .4002E+03 SUM 0. SUMMUIS .0300E+05

SPECLT LU CS .3000E+00 XS .2125E+00 ADUTS 0.
 SPECLT LU2 CS .0000E+01 XS .2700E-01 ADUTS 0.
 SPECLT M CS .0300E-03 XS .1200E-01 ADUTS 0.
 SPECLT M2 CS .2000E-01 XS .2300E+00 ADUTS 0.
 SPECLT UM CS .2000E-02 XS .3033E-02 ADUTS 0.
 SPECLT M2U CS .1000E+00 XS .2093E+00 ADUTS 0.
 SPECLT M2 CS .0200E+00 XS .2075E+00 ADUTS 0.
 SPECLT U CS .7000E-04 XS .9000E-04 ADUTS 0.
 SPECLT U2 CS .5000E-04 XS .3071E-04 ADUTS 0.
 SPECLT A CS .1100E-05 XS .5050E-06 ADUTS 0.

21 AS 1.0507E+01 MS 1.1050E+01 PHIS 3.0070E-01 IS 2.5050E+03 PS 5.9500E+00 US 1.7700E+01
 DELTA .0002E+00 DELTA .0002E+00 MS .7201E+01 HTS .3033E+03 TANS 0. US 0.
 PIS .1020E+02 MMUS .5012E-03 SUM 0. SUMMUIS .0700E+05

SPECLT LU CS .3000E+00 XS .2125E+00 ADUTS 0.
 SPECLT LU2 CS .0000E+01 XS .2700E-01 ADUTS 0.
 SPECLT M CS .0300E-03 XS .1200E-01 ADUTS 0.
 SPECLT M2 CS .2000E-01 XS .2300E+00 ADUTS 0.
 SPECLT UM CS .2000E-02 XS .3033E-02 ADUTS 0.
 SPECLT M2U CS .1000E+00 XS .2093E+00 ADUTS 0.
 SPECLT M2 CS .0200E+00 XS .2075E+00 ADUTS 0.
 SPECLT U CS .7000E-04 XS .9000E-04 ADUTS 0.
 SPECLT U2 CS .5000E-04 XS .3071E-04 ADUTS 0.
 SPECLT A CS .1100E-05 XS .5050E-06 ADUTS 0.

22 AS 1.0200E+01 MS 1.0215E+01 PHIS 3.7070E-01 IS 2.5500E+03 PS 5.0000E+00 US 1.7010E+01
 DELTA .0002E+00 DELTA .0002E+00 MS .7203E+00 HTS .3700E+03 TANS 0. US 0.
 PIS .1500E+02 MMUS .5305E-03 SUM 0. SUMMUIS .5197E+05

SPECLT LU CS .3000E+00 XS .2125E+00 ADUTS 0.
 SPECLT LU2 CS .0000E+01 XS .2700E-01 ADUTS 0.
 SPECLT M CS .0300E-03 XS .1200E-01 ADUTS 0.
 SPECLT M2 CS .2000E-01 XS .2300E+00 ADUTS 0.
 SPECLT UM CS .2000E-02 XS .3033E-02 ADUTS 0.
 SPECLT M2U CS .1000E+00 XS .2093E+00 ADUTS 0.
 SPECLT M2 CS .0200E+00 XS .2075E+00 ADUTS 0.
 SPECLT U CS .7000E-04 XS .9000E-04 ADUTS 0.
 SPECLT U2 CS .5000E-04 XS .3071E-04 ADUTS 0.
 SPECLT A CS .1100E-05 XS .5050E-06 ADUTS 0.

23 AS 1.0000E+01 MS 1.0275E+01 PHIS 3.9077E-01 IS 2.5530E+03 PS 5.3000E+00 US 1.7001E+01
 DELTA .0002E+00 DELTA .0002E+00 MS .7725E+01 HTS .3700E+03 TANS 0. US 0.
 PIS .1500E+02 MMUS .5100E-03 SUM 0. SUMMUIS .5000E+05

SPECLT LU CS .3000E+00 XS .2125E+00 ADUTS 0.
 SPECLT LU2 CS .0000E+01 XS .2700E-01 ADUTS 0.
 SPECLT M CS .0300E-03 XS .1200E-01 ADUTS 0.
 SPECLT M2 CS .2000E-01 XS .2300E+00 ADUTS 0.
 SPECLT UM CS .2000E-02 XS .3033E-02 ADUTS 0.
 SPECLT M2U CS .1000E+00 XS .2093E+00 ADUTS 0.
 SPECLT M2 CS .0200E+00 XS .2075E+00 ADUTS 0.
 SPECLT U CS .7000E-04 XS .9000E-04 ADUTS 0.
 SPECLT U2 CS .5000E-04 XS .3071E-04 ADUTS 0.
 SPECLT A CS .1100E-05 XS .5050E-06 ADUTS 0.

24 AS 1.0000E+00 MS 1.1530E+01 PHIS 4.1001E-01 IS 2.5000E+03 PS 5.0000E+00 US 1.0077E+01
 DELTA .0002E+00 DELTA .0002E+00 MS .7231E+02 HTS .3000E+03 TANS 0. US 0.
 PIS .1531E+02 MMUS .0000E-03 SUM 0. SUMMUIS .0100E+05

SPECLT LU CS .3000E+00 XS .2125E+00 ADUTS 0.
 SPECLT LU2 CS .0000E+01 XS .2700E-01 ADUTS 0.
 SPECLT M CS .0300E-03 XS .1200E-01 ADUTS 0.
 SPECLT M2 CS .2000E-01 XS .2300E+00 ADUTS 0.
 SPECLT UM CS .2000E-02 XS .3033E-02 ADUTS 0.
 SPECLT M2U CS .1000E+00 XS .2093E+00 ADUTS 0.
 SPECLT M2 CS .0200E+00 XS .2075E+00 ADUTS 0.
 SPECLT U CS .7000E-04 XS .9000E-04 ADUTS 0.
 SPECLT U2 CS .5000E-04 XS .3071E-04 ADUTS 0.
 SPECLT A CS .1100E-05 XS .5050E-06 ADUTS 0.

25 AS 1.5715E+01 MS 1.0100E+01 PHIS 4.5200E-01 IS 2.4010E+03 PS 0.0000E+00 US 1.0000E+01
 DELTA .0002E+00 DELTA .0002E+00 MS .7203E+02 HTS .3000E+03 TANS 0. US 0.
 PIS .1000E+02 MMUS .0000E-03 SUM 0. SUMMUIS .0000E+05

SPECLT LU CS .3000E+00 XS .2125E+00 ADUTS 0.
 SPECLT LU2 CS .0000E+01 XS .2700E-01 ADUTS 0.
 SPECLT M CS .0300E-03 XS .1200E-01 ADUTS 0.
 SPECLT M2 CS .2000E-01 XS .2300E+00 ADUTS 0.
 SPECLT UM CS .2000E-02 XS .3033E-02 ADUTS 0.
 SPECLT M2U CS .1000E+00 XS .2093E+00 ADUTS 0.
 SPECLT M2 CS .0200E+00 XS .2075E+00 ADUTS 0.
 SPECLT U CS .7000E-04 XS .9000E-04 ADUTS 0.
 SPECLT U2 CS .5000E-04 XS .3071E-04 ADUTS 0.
 SPECLT A CS .1100E-05 XS .5050E-06 ADUTS 0.

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GRAMM
FFF = .3E+01,
FFG = .3E+01,
UHG = -0.0,
TL = .330E+00,
CS = .323E+00,
TPO = .230E+00,
RMS = .4E+01,
RT = .10E+03,
WTRM = .250E+03,
SIG = -0.0,
EP = -0.0,
WPS = 0,
MC = 1,
RMS = 10,
NP = .1E+03, .2E+03, .3E+03, .4E+03, 0.0, 0.0, 0.0, 0.0,
NI = 0.0, 0.0, 0.0, 0.0, 0.0, 0.0, 0.0, 0.0,
VI = 0.0, 0.0, 0.0, 0.0, 0.0, 0.0, 0.0, 0.0,
TPI = 0.0, 0.0, 0.0, 0.0, 0.0, 0.0, 0.0, 0.0,
RPI = 0.0, 0.0, 0.0, 0.0, 0.0, 0.0, 0.0, 0.0,

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SEND
STREAMLINE NO. 1
DOWNSTREAM VELOCITY 1.470E+05 1.390E+05 1.310E+05 1.200E+05
CROSS-STREAM VELOCITY -2.700E+02 -5.500E+02 -7.000E+02 -1.010E+03
PARTICLE TEMPERATURE 2.013E+03 2.003E+03 2.013E+03 2.073E+03
PARTICLE DENSITY 2.700E+05 1.430E+04 1.020E+04 5.500E+05
STREAMLINE NO. 2
DOWNSTREAM VELOCITY 1.470E+05 1.390E+05 1.320E+05 1.200E+05
CROSS-STREAM VELOCITY -2.920E+02 -5.920E+02 -6.140E+02 -1.070E+03
PARTICLE TEMPERATURE 2.013E+03 2.003E+03 2.013E+03 2.073E+03
PARTICLE DENSITY 2.700E+05 1.430E+04 1.020E+04 5.500E+05
STREAMLINE NO. 3
DOWNSTREAM VELOCITY 1.400E+05 1.390E+05 1.320E+05 1.250E+05
CROSS-STREAM VELOCITY -3.100E+02 -6.200E+02 -6.040E+02 -1.100E+03
PARTICLE TEMPERATURE 2.013E+03 2.003E+03 2.013E+03 2.073E+03
PARTICLE DENSITY 2.700E+05 1.430E+04 1.020E+04 5.500E+05
STREAMLINE NO. 4
DOWNSTREAM VELOCITY 1.400E+05 1.390E+05 1.320E+05 1.250E+05
CROSS-STREAM VELOCITY -3.310E+02 -6.710E+02 -6.220E+02 -1.210E+03
PARTICLE TEMPERATURE 2.013E+03 2.003E+03 2.013E+03 2.073E+03
PARTICLE DENSITY 2.700E+05 1.430E+04 1.020E+04 5.500E+05
STREAMLINE NO. 5
DOWNSTREAM VELOCITY 1.400E+05 1.400E+05 1.320E+05 1.250E+05
CROSS-STREAM VELOCITY -3.550E+02 -7.190E+02 -6.000E+02 -1.300E+03
PARTICLE TEMPERATURE 2.013E+03 2.003E+03 2.013E+03 2.073E+03
PARTICLE DENSITY 2.700E+05 1.430E+04 1.020E+04 5.500E+05
STREAMLINE NO. 6
DOWNSTREAM VELOCITY 1.400E+05 1.400E+05 1.320E+05 1.250E+05
CROSS-STREAM VELOCITY -3.820E+02 -7.750E+02 -1.060E+03 -1.400E+03
PARTICLE TEMPERATURE 2.013E+03 2.003E+03 2.013E+03 2.073E+03
PARTICLE DENSITY 2.700E+05 1.430E+04 1.020E+04 5.500E+05
STREAMLINE NO. 7
DOWNSTREAM VELOCITY 1.400E+05 1.400E+05 1.320E+05 1.250E+05
CROSS-STREAM VELOCITY -4.140E+02 -8.390E+02 -1.150E+03 -1.510E+03
PARTICLE TEMPERATURE 2.003E+03 2.003E+03 2.013E+03 2.063E+03
PARTICLE DENSITY 2.700E+05 1.430E+04 1.020E+04 5.500E+05
STREAMLINE NO. 8
DOWNSTREAM VELOCITY 1.400E+05 1.410E+05 1.330E+05 1.200E+05
CROSS-STREAM VELOCITY -4.510E+02 -9.150E+02 -1.260E+03 -1.650E+03
PARTICLE TEMPERATURE 2.003E+03 2.003E+03 2.013E+03 2.063E+03
PARTICLE DENSITY 2.700E+05 1.430E+04 1.020E+04 5.500E+05
STREAMLINE NO. 9
DOWNSTREAM VELOCITY 1.500E+05 1.410E+05 1.330E+05 1.200E+05
CROSS-STREAM VELOCITY -4.970E+02 -1.010E+03 -1.300E+03 -1.820E+03
PARTICLE TEMPERATURE 2.003E+03 2.003E+03 2.013E+03 2.063E+03
PARTICLE DENSITY 2.700E+05 1.430E+04 1.020E+04 5.500E+05
STREAMLINE NO. 10
DOWNSTREAM VELOCITY 1.500E+05 1.420E+05 1.340E+05 1.200E+05
CROSS-STREAM VELOCITY -5.520E+02 -1.120E+03 -1.540E+03 -2.070E+03
PARTICLE TEMPERATURE 2.700E+05 2.073E+03 2.003E+03 2.063E+03
PARTICLE DENSITY 2.700E+05 1.430E+04 1.020E+04 5.500E+05

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STREAMLINE NO.	11			
DOWNSTREAM VELOCITY	1.510E+05	1.430E+05	1.340E+05	1.270E+05
CROSS-STREAM VELOCITY	-0.210E+02	-1.200E+03	-1.730E+03	-2.270E+03
PARTICLE TEMPERATURE	2.703E+03	2.073E+03	2.003E+03	2.033E+03
PARTICLE DENSITY	2.700E+05	1.030E+04	1.020E+04	5.500E+05
STREAMLINE NO.	12			
DOWNSTREAM VELOCITY	1.520E+05	1.430E+05	1.350E+05	1.270E+05
CROSS-STREAM VELOCITY	-7.000E+02	-1.000E+03	-1.900E+03	-2.000E+03
PARTICLE TEMPERATURE	2.703E+03	2.073E+03	2.003E+03	2.033E+03
PARTICLE DENSITY	2.700E+05	1.030E+04	1.020E+04	5.500E+05
STREAMLINE NO.	13			
DOWNSTREAM VELOCITY	1.530E+05	1.400E+05	1.300E+05	1.200E+05
CROSS-STREAM VELOCITY	-0.000E+02	-1.000E+03	-2.310E+03	-3.030E+03
PARTICLE TEMPERATURE	2.703E+03	2.073E+03	2.003E+03	2.033E+03
PARTICLE DENSITY	2.700E+05	1.030E+04	1.020E+04	5.500E+05
STREAMLINE NO.	14			
DOWNSTREAM VELOCITY	1.500E+05	1.450E+05	1.300E+05	1.200E+05
CROSS-STREAM VELOCITY	-0.930E+02	-2.010E+03	-2.770E+03	-3.030E+03
PARTICLE TEMPERATURE	2.703E+03	2.073E+03	2.003E+03	2.033E+03
PARTICLE DENSITY	2.700E+05	1.030E+04	1.020E+04	5.500E+05
STREAMLINE NO.	15			
DOWNSTREAM VELOCITY	1.550E+05	1.400E+05	1.370E+05	1.290E+05
CROSS-STREAM VELOCITY	-1.700E+03	-2.570E+03	-3.000E+03	-4.500E+03
PARTICLE TEMPERATURE	2.703E+03	2.073E+03	2.003E+03	2.033E+03
PARTICLE DENSITY	2.700E+05	1.030E+04	1.020E+04	5.500E+05
STREAMLINE NO.	16			
DOWNSTREAM VELOCITY	1.500E+05	1.470E+05	1.300E+05	1.300E+05
CROSS-STREAM VELOCITY	-1.000E+03	-3.300E+03	-4.010E+03	-6.000E+03
PARTICLE TEMPERATURE	2.703E+03	2.073E+03	2.073E+03	2.033E+03
PARTICLE DENSITY	2.700E+05	1.030E+04	1.020E+04	5.500E+05
STREAMLINE NO.	17			
DOWNSTREAM VELOCITY	1.500E+05	1.400E+05	1.390E+05	1.310E+05
CROSS-STREAM VELOCITY	-4.000E+03	-5.000E+03	-6.070E+03	-9.000E+03
PARTICLE TEMPERATURE	2.703E+03	2.073E+03	2.073E+03	2.033E+03
PARTICLE DENSITY	2.700E+05	1.030E+04	1.020E+04	5.500E+05
STREAMLINE NO.	18			
DOWNSTREAM VELOCITY	1.500E+05	1.500E+05	1.400E+05	1.320E+05
CROSS-STREAM VELOCITY	-0.470E+03	-1.010E+04	-1.300E+04	-1.020E+04
PARTICLE TEMPERATURE	2.713E+03	2.073E+03	2.053E+03	2.033E+03
PARTICLE DENSITY	2.700E+05	1.030E+04	1.020E+04	5.500E+05
STREAMLINE NO.	19			
DOWNSTREAM VELOCITY	0.	0.	0.	0.
CROSS-STREAM VELOCITY	0.	0.	0.	0.
PARTICLE TEMPERATURE	0.	0.	0.	0.
PARTICLE DENSITY	0.	0.	0.	0.
STREAMLINE NO.	20			
DOWNSTREAM VELOCITY	0.	0.	0.	0.
CROSS-STREAM VELOCITY	0.	0.	0.	0.
PARTICLE TEMPERATURE	0.	0.	0.	0.
PARTICLE DENSITY	0.	0.	0.	0.
STREAMLINE NO.	21			
DOWNSTREAM VELOCITY	0.	0.	0.	0.
CROSS-STREAM VELOCITY	0.	0.	0.	0.
PARTICLE TEMPERATURE	0.	0.	0.	0.
PARTICLE DENSITY	0.	0.	0.	0.
STREAMLINE NO.	22			
DOWNSTREAM VELOCITY	0.	0.	0.	0.
CROSS-STREAM VELOCITY	0.	0.	0.	0.
PARTICLE TEMPERATURE	0.	0.	0.	0.
PARTICLE DENSITY	0.	0.	0.	0.
STREAMLINE NO.	23			
DOWNSTREAM VELOCITY	0.	0.	0.	0.
CROSS-STREAM VELOCITY	0.	0.	0.	0.
PARTICLE TEMPERATURE	0.	0.	0.	0.
PARTICLE DENSITY	0.	0.	0.	0.
STREAMLINE NO.	24			
DOWNSTREAM VELOCITY	0.	0.	0.	0.
CROSS-STREAM VELOCITY	0.	0.	0.	0.
PARTICLE TEMPERATURE	0.	0.	0.	0.
PARTICLE DENSITY	0.	0.	0.	0.
STREAMLINE NO.	25			
DOWNSTREAM VELOCITY	0.	0.	0.	0.
CROSS-STREAM VELOCITY	0.	0.	0.	0.
PARTICLE TEMPERATURE	0.	0.	0.	0.
PARTICLE DENSITY	0.	0.	0.	0.
1	0	0	02	22
2	0	0	01	21
3	0	0	02	22
4	0	0	01	21
5	0	0	02	22
6	0	0	01	21
7	0	0	02	22
8	0	0	01	21
9	0	0	02	22
10	0	0	01	21
11	0	0	02	22
12	0	0	01	21
13	0	0	02	22
14	0	0	01	21
15	0	0	02	22
16	0	0	01	21
17	0	0	02	22
18	0	0	01	21
19	0	0	02	22
20	0	0	01	21
21	0	0	02	22
22	0	0	01	21
23	0	0	02	22
24	0	0	01	21
25	0	0	02	22
26	0	0	01	21
27	0	0	02	22
28	0	0	01	21
29	0	0	02	22
30	0	0	01	21
31	0	0	02	22
32	0	0	01	21
33	0	0	02	22
34	0	0	01	21
35	0	0	02	22
36	0	0	01	21
37	0	0	02	22
38	0	0	01	21
39	0	0	02	22
40	0	0	01	21
41	0	0	02	22
42	0	0	01	21
43	0	0	02	22
44	0	0	01	21
45	0	0	02	22
46	0	0	01	21
47	0	0	02	22
48	0	0	01	21
49	0	0	02	22
50	0	0	01	21
51	0	0	02	22
52	0	0	01	21
53	0	0	02	22
54	0	0	01	21
55	0	0	02	22
56	0	0	01	21
57	0	0	02	22
58	0	0	01	21
59	0	0	02	22
60	0	0	01	21
61	0	0	02	22
62	0	0	01	21
63	0	0	02	22
64	0	0	01	21
65	0	0	02	22
66	0	0	01	21
67	0	0	02	22
68	0	0	01	21
69	0	0	02	22
70	0	0	01	21
71	0	0	02	22
72	0	0	01	21
73	0	0	02	22
74	0	0	01	21
75	0	0	02	22
76	0	0	01	21
77	0	0	02	22
78	0	0	01	21
79	0	0	02	22
80	0	0	01	21
81	0	0	02	22
82	0	0	01	21
83	0	0	02	22
84	0	0	01	21
85	0	0	02	22
86	0	0	01	21
87	0	0	02	22
88	0	0	01	21
89	0	0	02	22
90	0	0	01	21
91	0	0	02	22
92	0	0	01	21
93	0	0	02	22
94	0	0	01	21
95	0	0	02	22
96	0	0	01	21
97	0	0	02	22
98	0	0	01	21
99	0	0	02	22
100	0	0	01	21

0 X= 2.4027E+01 M= 5.7091E+00 PH= 1.5023E-01 T= 2.4517E+01 P= 3.0722E+00 U= 2.0092E+05
 M= .1779E+01 DELTA= .0175E+00 M= -.7303E+02 M= .4000E+03 T= 0.0
 P= .9135E+01 MM= .3043E+03 S= .1013E+02 SUM= .0341E+04

SPECIE CU C= .2976E+00 X= .2117E+00 ADU= -.1299E-05
 SPECIE CO2 C= .0373E+01 X= .2409E-01 ADU= .1299E-05
 SPECIE H C= .5005E-03 X= .5909E-02 ADU= .2159E-03
 SPECIE H2 C= .2030E-01 X= .2429E+00 ADU= -.2103E-03
 SPECIE OH C= .0000E-03 X= .1032E-02 ADU= -.2177E-03
 SPECIE H2O C= .1900E+00 X= .2107E+00 ADU= .2171E-03
 SPECIE H2 C= .0200E+00 X= .2900E+00 ADU= 0.
 SPECIE U C= .1399E-04 X= .1702E-04 ADU= -.0430E-06
 SPECIE U2 C= .1030E-04 X= .0405E-05 ADU= -.2197E-07
 SPECIE A C= .1100E-05 X= .5470E-06 ADU= 0.

9 X= 2.2200E+01 M= 6.5153E+00 PH= 1.4107E-01 T= 2.4400E+01 P= 3.0500E+00 U= 2.0111E+05
 M= .1777E+01 DELTA= .0181E+00 M= -.7557E+02 M= .4000E+03 T= 0.0
 P= .9091E+01 MM= .3025E+03 S= .1011E+02 SUM= .0253E+04

SPECIE CU C= .2976E+00 X= .2117E+00 ADU= -.1510E-05
 SPECIE CO2 C= .0370E+01 X= .2407E-01 ADU= .1510E-05
 SPECIE H C= .2976E-03 X= .5920E-02 ADU= .2109E-03
 SPECIE H2 C= .2030E-01 X= .2429E+00 ADU= -.2109E-03
 SPECIE OH C= .0000E-03 X= .1032E-02 ADU= -.2205E-03
 SPECIE H2O C= .1900E+00 X= .2107E+00 ADU= .2197E-03
 SPECIE H2 C= .0200E+00 X= .2900E+00 ADU= 0.
 SPECIE U C= .1307E-04 X= .1702E-04 ADU= -.0477E-06
 SPECIE U2 C= .1010E-04 X= .0323E-05 ADU= -.1917E-07
 SPECIE A C= .1100E-05 X= .5470E-06 ADU= 0.

10 X= 2.2130E+01 M= 7.3192E+00 PH= 2.0020E-01 T= 2.4400E+01 P= 3.0177E+00 U= 2.0127E+05
 M= .1780E+01 DELTA= .0193E+00 M= -.7700E+02 M= .4000E+03 T= 0.0
 P= .9012E+01 MM= .2990E+03 S= .1000E+02 SUM= .1040E+05

SPECIE CU C= .2975E+00 X= .2117E+00 ADU= -.2200E-05
 SPECIE CO2 C= .0369E+01 X= .2403E-01 ADU= .2200E-05
 SPECIE H C= .2903E-03 X= .5902E-02 ADU= .2104E-03
 SPECIE H2 C= .2000E-01 X= .2430E+00 ADU= -.2027E-03
 SPECIE OH C= .0500E-03 X= .1002E-02 ADU= -.2046E-03
 SPECIE H2O C= .1900E+00 X= .2107E+00 ADU= .2046E-03
 SPECIE H2 C= .0200E+00 X= .2900E+00 ADU= 0.
 SPECIE U C= .1330E-04 X= .1657E-04 ADU= -.0430E-06
 SPECIE U2 C= .0932E-05 X= .0103E-05 ADU= -.1390E-07
 SPECIE A C= .1100E-05 X= .5470E-06 ADU= 0.

11 X= 2.1952E+01 M= 8.1207E+00 PH= 2.5153E-01 T= 2.4503E+01 P= 2.9001E+00 U= 2.0107E+05
 M= .1700E+01 DELTA= .0211E+00 M= -.6071E+02 M= .4050E+03 T= 0.0
 P= .8930E+01 MM= .2907E+03 S= .1005E+02 SUM= .1270E+05

SPECIE CU C= .2975E+00 X= .2117E+00 ADU= -.2030E-05
 SPECIE CO2 C= .0360E+01 X= .2406E-01 ADU= .2030E-05
 SPECIE H C= .2900E-03 X= .5700E-02 ADU= .2067E-03
 SPECIE H2 C= .2000E-01 X= .2431E+00 ADU= -.2057E-03
 SPECIE OH C= .0370E-03 X= .0909E-03 ADU= -.2077E-03
 SPECIE H2O C= .1900E+00 X= .2107E+00 ADU= .2063E-03
 SPECIE H2 C= .0200E+00 X= .2900E+00 ADU= 0.
 SPECIE U C= .1290E-04 X= .1590E-04 ADU= -.0460E-06
 SPECIE U2 C= .0900E-05 X= .0090E-05 ADU= -.1120E-07
 SPECIE A C= .1100E-05 X= .5470E-06 ADU= 0.

12 X= 2.1753E+01 M= 8.9213E+00 PH= 2.5095E-01 T= 2.4503E+01 P= 2.9300E+00 U= 2.0207E+05
 M= .1740E+01 DELTA= .0240E+00 M= -.6237E+02 M= .4050E+03 T= 0.0
 P= .8830E+01 MM= .2927E+03 S= .1001E+02 SUM= .1540E+05

SPECIE CU C= .2975E+00 X= .2117E+00 ADU= -.2001E-05
 SPECIE CO2 C= .0359E+01 X= .2407E-01 ADU= .2001E-05
 SPECIE H C= .2800E-03 X= .5730E-02 ADU= .2102E-03
 SPECIE H2 C= .2000E-01 X= .2431E+00 ADU= -.2153E-03
 SPECIE OH C= .0270E-03 X= .0907E-03 ADU= -.2172E-03
 SPECIE H2O C= .1900E+00 X= .2107E+00 ADU= .2150E-03
 SPECIE H2 C= .0200E+00 X= .2900E+00 ADU= 0.
 SPECIE U C= .1201E-04 X= .1570E-04 ADU= -.0417E-06
 SPECIE U2 C= .0900E-05 X= .0090E-05 ADU= -.0923E-06
 SPECIE A C= .1100E-05 X= .5470E-06 ADU= 0.

13 X= 2.1550E+01 M= 9.7100E+00 PH= 2.5027E-01 T= 2.4510E+01 P= 2.9800E+00 U= 2.0290E+05
 M= .1740E+01 DELTA= .0277E+00 M= -.6400E+02 M= .4050E+03 T= 0.0
 P= .8710E+01 MM= .2841E+03 S= .0990E+01 SUM= .1803E+05

SPECIE CU C= .2970E+00 X= .2110E+00 ADU= -.2000E-05
 SPECIE CO2 C= .0350E+01 X= .2400E-01 ADU= .2000E-05
 SPECIE H C= .2800E-03 X= .5671E-02 ADU= .2201E-03
 SPECIE H2 C= .2000E-01 X= .2432E+00 ADU= -.2230E-03
 SPECIE OH C= .0125E-03 X= .0520E-03 ADU= -.2251E-03
 SPECIE H2O C= .1900E+00 X= .2107E+00 ADU= .2237E-03
 SPECIE H2 C= .0200E+00 X= .2900E+00 ADU= 0.
 SPECIE U C= .1200E-04 X= .1520E-04 ADU= -.0420E-06
 SPECIE U2 C= .0915E-05 X= .0090E-05 ADU= -.0923E-06
 SPECIE A C= .1100E-05 X= .5470E-06 ADU= 0.

Fig

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AN	1.0200E+01	HA	1.5217E+01	PHI	0.2513E+01	TA	2.1991E+03	PA	2.3500E+00	UA	2.1005E+05
MA	.2030E+01	DELTA	.7021E+00	HA	-.2100E+03	MTA	.3500E+03	TAA	0.	GA	0.
PI	.0511E+01	MMO	.2010E+03	SA	.0020E+01	SUMDITA	.0332E+05				

SPECIE	CU	CA	.2050E+00	XA	.2103E+00	ADDTA	-.2103E+00
SPECIE	CU2	CA	.0700E+01	XA	.3080E+01	ADDTA	.2103E+00
SPECIE	H	CA	.1125E+03	XA	.2200E+02	ADDTA	.5000E+00
SPECIE	H2	CA	.2072E+01	XA	.2000E+00	ADDTA	-.5000E+00
SPECIE	HM	CA	.2200E+03	XA	.2000E+03	ADDTA	-.5000E+00
SPECIE	H2O	CA	.1000E+00	XA	.2101E+00	ADDTA	.5000E+00
SPECIE	H2	CA	.4200E+00	XA	.2000E+00	ADDTA	0.
SPECIE	U	CA	.1010E+05	XA	.1700E+05	ADDTA	-.1000E+00
SPECIE	U2	CA	.1000E+05	XA	.0700E+00	ADDTA	-.0000E+00
SPECIE	N	CA	.1100E+05	XA	.5000E+00	ADDTA	0.

21

AN	1.0000E+01	HA	1.5020E+01	PHI	3.4350E+01	TA	2.2900E+03	PA	2.9015E+00	UA	2.1000E+05
MA	.1910E+01	DELTA	.0003E+00	HA	-.1050E+03	MTA	.3025E+03	TAA	0.	GA	0.
PI	.0707E+01	MMO	.3125E+03	SA	.0341E+01	SUMDITA	.4750E+05				

SPECIE	CU	CA	.2000E+00	XA	.2110E+00	ADDTA	.2750E+05
SPECIE	CU2	CA	.0507E+01	XA	.2052E+01	ADDTA	-.2750E+05
SPECIE	H	CA	.1207E+03	XA	.2410E+02	ADDTA	-.0012E+00
SPECIE	H2	CA	.2050E+01	XA	.2050E+00	ADDTA	.0000E+00
SPECIE	HM	CA	.2050E+03	XA	.3111E+03	ADDTA	-.0000E+00
SPECIE	H2O	CA	.1005E+00	XA	.2113E+00	ADDTA	-.0050E+00
SPECIE	H2	CA	.4200E+00	XA	.2000E+00	ADDTA	0.
SPECIE	U	CA	.1055E+05	XA	.2000E+05	ADDTA	.1700E+07
SPECIE	U2	CA	.1155E+05	XA	.7105E+00	ADDTA	-.7000E+00
SPECIE	N	CA	.1100E+05	XA	.5000E+00	ADDTA	0.

22

AN	1.0010E+01	HA	1.0050E+01	PHI	0.5705E+01	TA	2.2551E+03	PA	2.7000E+00	UA	2.1050E+05
MA	.1970E+01	DELTA	.0000E+00	HA	-.1000E+03	MTA	.3700E+03	TAA	0.	GA	0.
PI	.0520E+01	MMO	.2013E+03	SA	.0350E+01	SUMDITA	.5107E+05				

SPECIE	CU	CA	.2005E+00	XA	.2112E+00	ADDTA	-.0000E+05
SPECIE	CU2	CA	.0570E+01	XA	.2003E+01	ADDTA	.0000E+05
SPECIE	H	CA	.1503E+03	XA	.2000E+02	ADDTA	.0003E+03
SPECIE	H2	CA	.2050E+01	XA	.2050E+00	ADDTA	-.1000E+03
SPECIE	HM	CA	.3550E+03	XA	.4171E+03	ADDTA	-.2070E+03
SPECIE	H2O	CA	.1002E+00	XA	.2100E+00	ADDTA	.1050E+03
SPECIE	H2	CA	.4200E+00	XA	.2000E+00	ADDTA	0.
SPECIE	U	CA	.2050E+05	XA	.3000E+05	ADDTA	-.5032E+07
SPECIE	U2	CA	.2202E+05	XA	.1023E+05	ADDTA	.5000E+00
SPECIE	N	CA	.1100E+05	XA	.5000E+00	ADDTA	0.

23

AN	1.0000E+01	HA	1.7000E+01	PHI	0.5020E+01	TA	2.2512E+03	PA	2.5000E+00	UA	2.1020E+05
MA	.2010E+01	DELTA	.7003E+00	HA	-.1020E+03	MTA	.3700E+03	TAA	0.	GA	0.
PI	.0000E+01	MMO	.2730E+03	SA	.0312E+01	SUMDITA	.5003E+05				

SPECIE	CU	CA	.2050E+00	XA	.2102E+00	ADDTA	-.2301E+05
SPECIE	CU2	CA	.0700E+01	XA	.3001E+01	ADDTA	.2301E+05
SPECIE	H	CA	.1027E+03	XA	.2037E+02	ADDTA	.0000E+03
SPECIE	H2	CA	.2070E+01	XA	.2000E+00	ADDTA	-.2053E+03
SPECIE	HM	CA	.3103E+03	XA	.3000E+03	ADDTA	-.2070E+03
SPECIE	H2O	CA	.1003E+00	XA	.2000E+00	ADDTA	.2050E+03
SPECIE	H2	CA	.4200E+00	XA	.2000E+00	ADDTA	0.
SPECIE	U	CA	.2000E+05	XA	.3117E+05	ADDTA	-.4011E+00
SPECIE	U2	CA	.2010E+05	XA	.1250E+05	ADDTA	-.3100E+00
SPECIE	N	CA	.1100E+05	XA	.5000E+00	ADDTA	0.

24

AN	1.0000E+01	HA	1.7707E+01	PHI	0.0525E+01	TA	2.2401E+03	PA	2.5033E+00	UA	2.1010E+05
MA	.2010E+01	DELTA	.0007E+00	HA	-.1000E+03	MTA	.3030E+03	TAA	0.	GA	0.
PI	.0000E+01	MMO	.2700E+03	SA	.0277E+01	SUMDITA	.0100E+05				

SPECIE	CU	CA	.2050E+00	XA	.2107E+00	ADDTA	-.1190E+05
SPECIE	CU2	CA	.0000E+01	XA	.3033E+01	ADDTA	.1190E+05
SPECIE	H	CA	.1370E+03	XA	.2700E+02	ADDTA	.1205E+03
SPECIE	H2	CA	.2000E+01	XA	.2000E+00	ADDTA	-.1100E+03
SPECIE	HM	CA	.3070E+03	XA	.3010E+03	ADDTA	-.1212E+03
SPECIE	H2O	CA	.1000E+00	XA	.2100E+00	ADDTA	.1203E+03
SPECIE	H2	CA	.4200E+00	XA	.2000E+00	ADDTA	0.
SPECIE	U	CA	.2121E+05	XA	.2000E+05	ADDTA	-.2717E+00
SPECIE	U2	CA	.1700E+05	XA	.1110E+05	ADDTA	-.1007E+00
SPECIE	N	CA	.1100E+05	XA	.5000E+00	ADDTA	0.

25

AN	1.0000E+01	HA	1.0303E+01	PHI	0.0020E+01	TA	2.1505E+03	PA	2.0010E+00	UA	2.0057E+05
MA	.1000E+01	DELTA	.0000E+00	HA	-.2313E+03	MTA	.2900E+03	TAA	-.2077E+05	GA	-.1010E+03
PI	.0000E+01	MMO	.2030E+03	SA	.0257E+01	SUMDITA	.0503E+05				

WALL TEMPERATURE 1000. KELVIN

WALL TEMPERATURE 1000. KELVIN

WBS	.7000E+00	WBS	.2100E+00	WBS	.2120E+01	WBS	.3703E+02	WBS	.0100E+00
WBS	.7000E+00	WBS	.2100E+00	WBS	.2120E+01	WBS	.3703E+02	WBS	.0100E+00
WBS	.7000E+00	WBS	.2100E+00	WBS	.2120E+01	WBS	.3703E+02	WBS	.0100E+00

MANDATORY LAYEN PHOTOS

U/1E	1/E	U/1E 1/E	Y/1E 1
1000E+01	4094E+00	3040E+00	2100E+01
2000E+01	4771E+00	4087E+00	4197E+01
3000E+01	4055E+00	9040E+00	6270E+01
4000E+01	4093E+00	1203E+01	8320E+01
5000E+01	5034E+00	1400E+01	1030E+02
6000E+01	5127E+00	1744E+01	1234E+02
7000E+01	5221E+00	2070E+01	1430E+02
8000E+01	5317E+00	2305E+01	1630E+02
9000E+01	5413E+00	2640E+01	1832E+02
1000E+00	5510E+00	2930E+01	2037E+02
1400E+00	5940E+00	4240E+01	2474E+02
2000E+00	6474E+00	5074E+01	3000E+02
2500E+00	6450E+00	6070E+01	3035E+02
3000E+00	7375E+00	8101E+01	4044E+02
3500E+00	7780E+00	9292E+01	6165E+02
4000E+00	8172E+00	1045E+02	1030E+01
4500E+00	8575E+00	1154E+02	1203E+01
5000E+00	8845E+00	1270E+02	1415E+01
5500E+00	9130E+00	1500E+02	2011E+01
6000E+00	9300E+00	1600E+02	2527E+01
6500E+00	9594E+00	1594E+02	3040E+01
7000E+00	9764E+00	1700E+02	4044E+01
7500E+00	9907E+00	1805E+02	1254E+01
8000E+00	1001E+01	1942E+02	1402E+01
8500E+00	1000E+01	2014E+02	2051E+01
9000E+00	1000E+01	2117E+02	4000E+00
9500E+00	1000E+01	2220E+02	6000E+00
1000E+01	1000E+01	2324E+02	1000E+01

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SPECIFIC CU CM 2.94E+00 XE 2.101E+00 ADU1 = .1007E+05
SPECIFIC CL2 CM 6.644E+01 XE 4.106E+01 ADU1 = .1007E+05
SPECIFIC H CM 9.23E+04 XE 1.044E+02 ADU1 = .7007E+04
SPECIFIC H2 CM 2.076E+01 XE 2.072E+00 ADU1 = .7007E+04
SPECIFIC UM CM 1.741E+03 XE 2.104E+03 ADU1 = .7025E+04
SPECIFIC M2U CM 1.093E+00 XE 2.100E+00 ADU1 = .7036E+04
SPECIFIC H2 CM 2.076E+00 XE 2.095E+00 ADU1 = U.
SPECIFIC U CM 6.444E+06 XE 1.154E+04 ADU1 = .1173E+06
SPECIFIC H2 CM 1.73E+06 XE 4.512E+06 ADU1 = .4511E+00
SPECIFIC H CM 1.110E+05 XE 5.442E+06 ADU1 = U.
=====
--STREAMLINE 1 SS 10.20405
-- LOCAL GAS PROPERTIES
-- U 2.010E+05 T 2.454E+03 DENSITY 1.077E+04 REYNOLDS NU= 6.911E+21 Y2= 8.225E+01 H2= 8.220E+01
----- PARTICLE PHASE PROPERTIES -----
PARTICLE GROUP 1 2 3 4
DOWNSTREAM VELOCITY 1.090E+05 1.001E+05 1.530E+05 1.421E+05
CROSS-STREAM VELOCITY -3.750E+01 -2.459E+02 -5.805E+02 -0.015E+02
PARTICLE REYNOLDS NO. 1.228E+00 6.719E+00 1.453E+01 2.373E+01
PARTICLE TEMPERATURE 2.527E+03 2.634E+03 2.714E+03 2.702E+03
PARTICLE DENSITY 1.220E+05 2.015E+05 4.011E+05 3.200E+05
PARTICLE MOLE FRACTION 2.524E+00
--STREAMLINE 2 SS 10.20405
-- LOCAL GAS PROPERTIES
-- U 2.010E+05 T 2.453E+03 DENSITY 1.073E+04 REYNOLDS NU= 6.902E+21 Y2= 1.042E+00 H2= 1.042E+00
----- PARTICLE PHASE PROPERTIES -----
PARTICLE GROUP 1 2 3 4
DOWNSTREAM VELOCITY 1.092E+05 1.001E+05 1.535E+05 1.421E+05
CROSS-STREAM VELOCITY -1.103E+02 -5.540E+02 -0.011E+02 -0.006E+03
PARTICLE REYNOLDS NO. 1.240E+00 6.740E+00 1.444E+01 2.340E+01
PARTICLE TEMPERATURE 2.523E+03 2.627E+03 2.710E+03 2.700E+03
PARTICLE DENSITY 1.207E+05 4.744E+05 4.200E+05 2.412E+05
PARTICLE MOLE FRACTION 2.524E+00
--STREAMLINE 3 SS 10.20405
-- LOCAL GAS PROPERTIES
-- U 2.010E+05 T 2.454E+03 DENSITY 1.072E+04 REYNOLDS NU= 6.877E+21 Y2= 2.401E+00 H2= 2.450E+00
----- PARTICLE PHASE PROPERTIES -----
PARTICLE GROUP 1 2 3 4
DOWNSTREAM VELOCITY 1.091E+05 1.001E+05 1.534E+05 1.420E+05
CROSS-STREAM VELOCITY -2.540E+02 -0.070E+02 -0.252E+03 -0.525E+03
PARTICLE REYNOLDS NO. 1.233E+00 6.710E+00 1.447E+01 2.340E+01
PARTICLE TEMPERATURE 2.523E+03 2.620E+03 2.710E+03 2.700E+03
PARTICLE DENSITY 1.222E+05 6.015E+05 4.195E+05 2.402E+05
PARTICLE MOLE FRACTION 2.524E+00
--STREAMLINE 4 SS 10.20405
-- LOCAL GAS PROPERTIES
-- U 2.010E+05 T 2.454E+03 DENSITY 1.075E+04 REYNOLDS NU= 6.851E+21 Y2= 3.200E+00 H2= 3.275E+00
----- PARTICLE PHASE PROPERTIES -----
PARTICLE GROUP 1 2 3 4
DOWNSTREAM VELOCITY 1.091E+05 1.001E+05 1.534E+05 1.420E+05
CROSS-STREAM VELOCITY -1.000E+02 -0.200E+03 -0.700E+03 -0.010E+03
PARTICLE REYNOLDS NO. 1.230E+00 6.690E+00 1.443E+01 2.335E+01
PARTICLE TEMPERATURE 2.523E+03 2.629E+03 2.711E+03 2.700E+03
PARTICLE DENSITY 1.224E+05 6.020E+05 4.184E+05 2.400E+05
PARTICLE MOLE FRACTION 2.524E+00

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--SHEET 1 OF 5
-- LOCAL GAS PROPERTIES
-- 2,000000 IS 2,000000 DENSITY 3,000000 METRICUS NO. 0,000000 120 0,000000 020 0,000000
-- PANTICLE PHASE PROPERTIES -----
PANTICLE GROUP 1 2 3 4
DOWNSIDE VELOCITY 1,000000 1,000000 1,500000 1,000000
CROSS-STREAM VELOCITY -0,000000 -0,000000 -0,000000 -0,000000
PANTICLE METRICUS NO. 1,000000 0,000000 1,000000 0,000000
PANTICLE TEMPERATURE 2,000000 2,000000 2,000000 2,000000
PANTICLE DENSITY 1,000000 0,000000 0,000000 0,000000
PANTICLE MOMENTUM FLUX 2,000000
--SHEET 2 OF 5
-- LOCAL GAS PROPERTIES
-- 2,000000 IS 2,000000 DENSITY 3,000000 METRICUS NO. 0,000000 120 0,000000 020 0,000000
-- PANTICLE PHASE PROPERTIES -----
PANTICLE GROUP 1 2 3 4
DOWNSIDE VELOCITY 1,000000 1,000000 1,500000 1,000000
CROSS-STREAM VELOCITY -0,000000 -0,000000 -0,000000 -0,000000
PANTICLE METRICUS NO. 1,000000 0,000000 1,000000 0,000000
PANTICLE TEMPERATURE 2,000000 2,000000 2,000000 2,000000
PANTICLE DENSITY 1,000000 0,000000 0,000000 0,000000
PANTICLE MOMENTUM FLUX 2,000000
--SHEET 3 OF 5
-- LOCAL GAS PROPERTIES
-- 2,000000 IS 2,000000 DENSITY 3,000000 METRICUS NO. 0,000000 120 0,000000 020 0,000000
-- PANTICLE PHASE PROPERTIES -----
PANTICLE GROUP 1 2 3 4
DOWNSIDE VELOCITY 1,000000 1,000000 1,500000 1,000000
CROSS-STREAM VELOCITY -0,000000 -0,000000 -0,000000 -0,000000
PANTICLE METRICUS NO. 1,000000 0,000000 1,000000 0,000000
PANTICLE TEMPERATURE 2,000000 2,000000 2,000000 2,000000
PANTICLE DENSITY 1,000000 0,000000 0,000000 0,000000
PANTICLE MOMENTUM FLUX 2,000000
--SHEET 4 OF 5
-- LOCAL GAS PROPERTIES
-- 2,000000 IS 2,000000 DENSITY 3,000000 METRICUS NO. 0,000000 120 0,000000 020 0,000000
-- PANTICLE PHASE PROPERTIES -----
PANTICLE GROUP 1 2 3 4
DOWNSIDE VELOCITY 1,000000 1,000000 1,500000 1,000000
CROSS-STREAM VELOCITY -0,000000 -0,000000 -0,000000 -0,000000
PANTICLE METRICUS NO. 1,000000 0,000000 1,000000 0,000000
PANTICLE TEMPERATURE 2,000000 2,000000 2,000000 2,000000
PANTICLE DENSITY 1,000000 0,000000 0,000000 0,000000
PANTICLE MOMENTUM FLUX 2,000000
--SHEET 5 OF 5
-- LOCAL GAS PROPERTIES
-- 2,000000 IS 2,000000 DENSITY 3,000000 METRICUS NO. 0,000000 120 0,000000 020 0,000000
-- PANTICLE PHASE PROPERTIES -----
PANTICLE GROUP 1 2 3 4
DOWNSIDE VELOCITY 1,000000 1,000000 1,500000 1,000000
CROSS-STREAM VELOCITY -0,000000 -0,000000 -0,000000 -0,000000
PANTICLE METRICUS NO. 1,000000 0,000000 1,000000 0,000000
PANTICLE TEMPERATURE 2,000000 2,000000 2,000000 2,000000
PANTICLE DENSITY 1,000000 0,000000 0,000000 0,000000
PANTICLE MOMENTUM FLUX 2,000000

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--STREAMLINE 15 SE 10.20405
-- LOCAL GAS PROPERTIES
US 2.032E+05 IS 2.424E+03 DENSITY 2.423E-04 REYNOLDS NO.= 0.430E+21 Y2= 1.000E+01 H2= 1.041E+01
----- PARTICLE PHASE PROPERTIES -----
PARTICLE GROUP 1 2 3 4
DOWNSTREAM VELOCITY 1.000E+05 1.703E+05 1.550E+05 1.440E+05
CROSS-STREAM VELOCITY -5.500E+03 -0.750E+03 -0.142E+03 -0.099E+03
PARTICLE REYNOLDS NO. 1.237E+00 0.293E+00 1.347E+01 2.701E+01
PARTICLE TEMPERATURE 2.502E+03 2.012E+03 2.090E+03 2.700E+03
PARTICLE DENSITY 1.204E-05 7.143E-05 0.024E-05 3.001E-05
PARTICLE MOMENTUM FLUX 2.710E+00
--STREAMLINE 16 SE 10.20405
-- LOCAL GAS PROPERTIES
US 2.040E+05 IS 2.443E+03 DENSITY 2.751E-04 REYNOLDS NO.= 0.501E+21 Y2= 1.152E+01 H2= 1.131E+01
----- PARTICLE PHASE PROPERTIES -----
PARTICLE GROUP 1 2 3 4
DOWNSTREAM VELOCITY 1.911E+05 1.700E+05 1.555E+05 1.441E+05
CROSS-STREAM VELOCITY -4.370E+03 -0.054E+03 -0.503E+03 -1.040E+04
PARTICLE REYNOLDS NO. 1.251E+00 0.243E+00 1.351E+01 2.710E+01
PARTICLE TEMPERATURE 2.494E+03 2.000E+03 2.077E+03 2.750E+03
PARTICLE DENSITY 1.270E-05 7.210E-05 0.722E-05 3.101E-05
PARTICLE MOMENTUM FLUX 7.747E+00
--STREAMLINE 15 SE 10.20405
-- LOCAL GAS PROPERTIES
US 2.052E+05 IS 2.411E+03 DENSITY 2.000E-04 REYNOLDS NO.= 0.270E+21 Y2= 1.237E+01 H2= 1.211E+01
----- PARTICLE PHASE PROPERTIES -----
PARTICLE GROUP 1 2 3 4
DOWNSTREAM VELOCITY 1.417E+05 1.710E+05 1.547E+05 1.459E+05
CROSS-STREAM VELOCITY -5.500E+03 -0.400E+03 -1.105E+04 -1.307E+04
PARTICLE REYNOLDS NO. 1.245E+00 0.242E+00 1.324E+01 2.151E+01
PARTICLE TEMPERATURE 2.492E+03 2.003E+03 2.090E+03 2.760E+03
PARTICLE DENSITY 1.277E-05 7.330E-05 1.003E-04 3.245E-05
PARTICLE MOMENTUM FLUX 2.453E+00
--STREAMLINE 16 SE 10.20405
-- LOCAL GAS PROPERTIES
US 2.067E+05 IS 2.390E+03 DENSITY 2.500E-04 REYNOLDS NO.= 0.192E+21 Y2= 1.320E+01 H2= 1.201E+01
----- PARTICLE PHASE PROPERTIES -----
PARTICLE GROUP 1 2 3 4
DOWNSTREAM VELOCITY 1.420E+05 1.721E+05 1.572E+05 1.441E+05
CROSS-STREAM VELOCITY -7.044E+03 -1.201E+04 -1.494E+04 -1.002E+04
PARTICLE REYNOLDS NO. 1.353E+00 0.242E+00 1.323E+01 2.145E+01
PARTICLE TEMPERATURE 2.470E+03 2.542E+03 2.072E+03 2.740E+03
PARTICLE DENSITY 1.270E-05 7.535E-05 1.059E-04 3.503E-05
PARTICLE MOMENTUM FLUX 3.002E+00
--STREAMLINE 17 SE 10.20405
-- LOCAL GAS PROPERTIES
US 2.079E+05 IS 2.302E+03 DENSITY 2.414E-04 REYNOLDS NO.= 0.123E+21 Y2= 1.414E+01 H2= 1.173E+01
----- PARTICLE PHASE PROPERTIES -----
PARTICLE GROUP 1 2 3 4
DOWNSTREAM VELOCITY 1.930E+05 1.730E+05 1.500E+05 1.473E+05
CROSS-STREAM VELOCITY -0.450E+03 -1.493E+04 -2.007E+04 -2.442E+04
PARTICLE REYNOLDS NO. 1.375E+00 0.171E+00 1.313E+01 2.124E+01
PARTICLE TEMPERATURE 2.407E+03 2.540E+03 2.070E+03 2.703E+03
PARTICLE DENSITY 1.290E-05 8.100E-05 1.245E-04 4.700E-05
PARTICLE MOMENTUM FLUX 3.503E+00
--STREAMLINE 18 SE 10.20405
-- LOCAL GAS PROPERTIES
US 2.133E+05 IS 2.511E+03 DENSITY 2.521E-04 REYNOLDS NO.= 0.035E+21 Y2= 1.540E+01 H2= 1.053E+01
----- PARTICLE PHASE PROPERTIES -----
PARTICLE GROUP 1 2 3 4
DOWNSTREAM VELOCITY 1.950E+05 0. 0. 0.
CROSS-STREAM VELOCITY -0.430E+03 0. 0. 0.
PARTICLE REYNOLDS NO. 1.573E+00 0. 0. 0.
PARTICLE TEMPERATURE 2.434E+03 0. 0. 0.
PARTICLE DENSITY 1.304E-05 0. 0. 0.
PARTICLE MOMENTUM FLUX 2.500E+01
--STREAMLINE 19 SE 10.20405
-- LOCAL GAS PROPERTIES
US 2.140E+05 IS 2.190E+03 DENSITY 2.410E-04 REYNOLDS NO.= 5.007E+21 Y2= 1.500E+01 H2= 1.522E+01
----- PARTICLE PHASE PROPERTIES -----
PARTICLE GROUP 1 2 3 4
DOWNSTREAM VELOCITY 0. 0. 0. 0.
CROSS-STREAM VELOCITY 0. 0. 0. 0.
PARTICLE REYNOLDS NO. 0. 0. 0. 0.
PARTICLE TEMPERATURE 0. 0. 0. 0.
PARTICLE DENSITY 0. 0. 0. 0.
PARTICLE MOMENTUM FLUX 0.
--STREAMLINE 20 SE 10.20405
-- LOCAL GAS PROPERTIES
US 2.101E+05 IS 2.271E+03 DENSITY 3.125E-04 REYNOLDS NO.= 5.071E+21 Y2= 1.007E+01 H2= 1.503E+01
----- PARTICLE PHASE PROPERTIES -----
PARTICLE GROUP 1 2 3 4
DOWNSTREAM VELOCITY 0. 0. 0. 0.
CROSS-STREAM VELOCITY 0. 0. 0. 0.
PARTICLE REYNOLDS NO. 0. 0. 0. 0.
PARTICLE TEMPERATURE 0. 0. 0. 0.
PARTICLE DENSITY 0. 0. 0. 0.
PARTICLE MOMENTUM FLUX 0.
BOUNDARY OF PARTICLE PHASE AT SE 10.01204 1.470E+01 1.417E+01 1.370E+01 1.342E+01
BOUNDARY OF PARTICLE PHASE AT SE 11.01090 1.400E+01 1.431E+01 1.309E+01 1.353E+01
BOUNDARY OF PARTICLE PHASE AT SE 11.01900 1.513E+01 1.447E+01 1.403E+01 1.300E+01
BOUNDARY OF PARTICLE PHASE AT SE 11.02986 1.530E+01 1.442E+01 1.417E+01 1.370E+01
BOUNDARY OF PARTICLE PHASE AT SE 12.02510 1.547E+01 1.470E+01 1.451E+01 1.391E+01
BOUNDARY OF PARTICLE PHASE AT SE 12.07040 1.504E+01 1.443E+01 1.404E+01 1.405E+01

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APPENDIX C

SAMPLE OUTPUT - EXPLANATIONS

C-i

APPENDIX C

SAMPLE OUTPUT - EXPLANATIONS

The first page of output contains all the input data on Cards 1-5 and 7, followed by the species identification data (Card 9) and the distributions of properties along the initial orthogonal surface. The units of each variable are identical with those used for the input data. These are followed by the axis location and the nozzle wall contour.

The standard printed output at each value of KP (Card 5, Cols. 11-15) or DXLSS (Card 3, Cols. 11-22) gives properties for each streamtube, from K (streamtube index) = 2 to K = KMAX. In addition to the usual output* i.e. X, R, PHI, T, P, U, the following properties are printed:

MA	Mach number
DELY	streamtube width (cm)
H	enthalpy (cal/g)
HT	stagnation enthalpy (cal/g)
TAW	shear stress at tube interface (dynes/cm ²)
Q	heat flux at tube interface (cal/cm ² -sec)
PT	dynamic pressure (atm)
RHO	density (g/cm ³)
SX	distance along streamtube (cm)
SUMDOT	total mass flow bounded by streamtube (g/sec)
C	species mass fraction
X	species mole fraction
WDOT	species production rate (g/cm ³ sec)
only for viscous flows F1	total mass flow of species up to present streamtube (g/sec)
ZJ	species flux at tube interface

* X and R refer to the coordinates of the outer boundary of streamtube while the flow properties are average values across the streamtube.

Boundary layer parameters, if computed, are printed between the tube properties and the composition data in the last streamtube (at downstream print stations). The following parameters are printed:

RES	Reynolds number based on distance along wall
DELTAI	boundary layer thickness (incompressible)
THETAI	momentum thickness (incompressible)
CFI	skin friction coefficient (incompressible)
UFI	friction velocity (incompressible)
DISP	displacement thickness
DELTAC	boundary layer thickness
THETAC	momentum thickness
CFC	skin friction coefficient
H12	shape factor ($= DISP/THETAC$)
QWALL	heat flux at wall ($cal/cm^2\text{-sec}$)
TAUWALL	shear stress at wall ($dynes/cm^2$)
U/UE	B. L. velocity profile ($UE =$ velocity in last tube)
T/TE	B. L. temperature profile ($TE =$ temperature in last tube)
Y/DEL	location in B. L. ($DEL \equiv DELTAC$)

If IPART = 1, particle properties are printed; first, a NAMELIST of the input data on Cards 15 through 19 (and 20 through 23 for NC = 0), followed by the initial particle properties in each streamtube. The standard particle print gives the downstream and cross-stream velocity, temperature, Reynolds number and particle cloud density for each particle group and total particle momentum flux, for each streamtube. Since, typically, the limiting particle streamlines are within streamtube KMAX the outer tubes will contain no particles. Limiting particle streamlines are noted as, 'BOUNDARY OF PARTICLE PHASE AT $S = X.XXXXX$ ', followed by the radial position of the limiting particle streamline for each particle group. If the particles are initially in the liquid phase the program will print where they start to solidify.

APPENDIX D

FORTRAN LISTING

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2 1 TAPES,TAPES,TAPES)
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			ALIBU	MAIN	62
			AL=1	MAIN	63
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			AL(1,11,5) * SEC,ALUSS,ALMAA,ALNUU,IMU,ANI	MAIN	114
			AL(1,11,5) * SEC,ALUSS,ALMAA,ALNUU,IMU,ANI	MAIN	115

	PSIMBTA	MAIN	151
	PSIMPP	MAIN	152
	PSIMH(AMH)	MAIN	153
200	PSIMH(AMH)	MAIN	154
	400(AMH+1)/(AMH+1)	MAIN	155
	TSIMSTIME(PH/PSIMH+GM)	MAIN	156
	I (PSIMPP/PP)	MAIN	157
	(AMPP/PSIMH+GM)/(PSIMPP/PSIMH+1)+USTIME+PSIMBTA	MAIN	158
205	LSIMBTA(USTIME+LSIMBTA PSIMH+GM)	MAIN	159
	100 CUNTIME	MAIN	160
	101 AMH+GM	MAIN	161
	ICONTM	MAIN	162
	INTGREALPHAP	MAIN	163
210	ALCAGE=0	MAIN	164
	ITHEM=0	MAIN	165
	LEL	MAIN	166
	DOO=0	MAIN	167
	PLAM=0.05	MAIN	168
215	SPOT=0.0	MAIN	169
	ISAMP	MAIN	170
	LINTE	MAIN	171
	SLT=1.0E-03	MAIN	172
	ESTABLISHING STABLE STEP SIZE	MAIN	173
220	TIME=0.0	MAIN	174
	1LETPDEC=TIME	MAIN	175
	1STEP=TIME-1LETP	MAIN	176
	IF (LEL,LEL,0) GO TO 205	MAIN	177
225	IF (1STEP,61.0,5+1LETP) GO TO 210	MAIN	178
	1ULSTIME	MAIN	179
	CALL STABLE	MAIN	180
	NCOUNT=0	MAIN	181
	ICOUNT=0	MAIN	182
	PIV=0	MAIN	183
230	IF (LEL,LT,PLAM),UM,(2),LT,PLAM) GO TO 210	MAIN	184
	IF (2),GE,ALPD,AND,ISMUCH,NE,1 GO TO 200	MAIN	185
	IF (1PM,LE,0),UM,(2),LT,PLAM) GO TO 230	MAIN	186
	1ANT OUT	MAIN	187
235	GO TO 303	MAIN	188
	1AKE OUT	MAIN	189
300	6AMH+GM	MAIN	190
	6AMH+GM	MAIN	191
	PH+PP(2)	MAIN	192
	1SM+2(2)	MAIN	193
240	1SM+2(2)	MAIN	194
	1AM+2(2)	MAIN	195
	1AM+2(2)	MAIN	196
	1M+2(2)	MAIN	197
	1LEH	MAIN	198
245	0LEPH+PP(2)	MAIN	199
	IF (1LEH,1) 3500,3500,3500	MAIN	200
3000	1LEH=(1AMH+2+2,0)/(1AMH+2+2)-6AMH+GM(1LEPH)+2	MAIN	201
	1LEH=(2,0+1AMH+2+2,0)/(1AMH+2+2)	MAIN	202
	1LEH=(1LEH+1,0)+2+2,0/(1AMH+2+2)/(1AMH+2+2)+1(1LEPH)+2	MAIN	203
250	1LEH=(1LEH)+2+2+2+2,0/(1AMH+2+2)+1(1LEPH)+2	MAIN	204
	1LEH=(1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	205
255	1LEH=(1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	206
	1LEH=(1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	207
	1LEH=(1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	208
	1LEH=(1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	209
260	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	210
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	211
265	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	212
	GO TO 270	MAIN	213
270	1LEH=(1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	214
	1LEH=(1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	215
	1LEH=(1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	216
275	1LEH=(1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	217
	1LEH=(1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	218
280	1LEH=(1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	219
	1LEH=(1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	220
	1LEH=(1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	221
	1LEH=(1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	222
	1LEH=(1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	223
285	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	224
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	225
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	226
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	227
290	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	228
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	229
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	230
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	231
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	232
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	233
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	234
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	235
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	236
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	237
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	238
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	239
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	240
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	241
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	242
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	243
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	244
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	245
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	246
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	247
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	248
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	249
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	250
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	251
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	252
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	253
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	254
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	255
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	256
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	257
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	258
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	259
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	260
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	261
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	262
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	263
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	264
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	265
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	266
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	267
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	268
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	269
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	270
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	271
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	272
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	273
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	274
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	275
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	276
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	277
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	278
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	279
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	280
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	281
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	282
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	283
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	284
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	285
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	286
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	287
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	288
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	289
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	290
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	291
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	292
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	293
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	294
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	295
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	296
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	297
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	298
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	299
	IF (1LEH,1LEH)+2+2+2+2,0/(1LEPH)+2	MAIN	300

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300  S10S14(100)
301  S10S14(100-00100)
302  S10S14(100)
303  S10S14(100-00100)
304  S10S14(100)
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396  S10S14(100)
397  S10S14(100-00100)
398  S10S14(100)
399  S10S14(100-00100)
400  S10S14(100)

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505	075	AA(N)M10(M2(N)+M2(N+1))=DELTA+DELTA(N)	MAIN	530
		V2(N)M2(N+1)+DELTA(N)	MAIN	530
		CALL STAP	MAIN	540
		IF ((IFLAG,NE,0).AND.(N=1),LE,MH2),AND.(IPART,NE,0)) CALL PUNT	MAIN	541
		IF (M2ALAM) 005,005,077	MAIN	542
500	077	M2ALAM=0	MAIN	543
		IF (ICOUNT=25)250,0010,0010	MAIN	544
	005	IF (M2CASE) 0060,000,0060	MAIN	545
	000	IF (MOVE) 1000,000,1000	MAIN	546
	000	IF (M2GSM) 000,000,000	MAIN	547
505	000	CALL PUTOUT(3)	MAIN	548
		JATZ=10	MAIN	549
		IF (IPART,NE,0) CALL PUTOUT(3)	MAIN	550
	070	NAME=1	MAIN	551
		IF (M2MAX) 500,1000,1000	MAIN	552
000	C	ENTER SUBROUTINE (M211005	MAIN	553
	1000	M2ALAM=1	MAIN	554
		M2MAX=0	MAIN	555
		CALL M2MAY	MAIN	556
		IF (M2GSM) 001,1020,001	MAIN	557
005	1020	M2MAY=0	MAIN	558
		IF (M2ALAM) 1030,1030,077	MAIN	559
	1030	IF (M2CASE) 0060,1050,0060	MAIN	560
		V2(N)M2(N+1)+DELTA(N)	MAIN	561
	1050	IF (M2GSM) 1000,1070,1000	MAIN	562
010	1000	CALL PUTOUT(3)	MAIN	563
		JATZ=11	MAIN	564
		IF (IPART,NE,0) CALL PUTOUT(3)	MAIN	565
	1070	IF (IFLAG) 1000,1150,1000	MAIN	566
	1150	M2MAX=101	MAIN	567
015		GO TO 1100 L=2,MAX	MAIN	568
		JUL=1	MAIN	569
		M(J)M2(J)	MAIN	570
		A(J)M2(J)	MAIN	571
		M(J)M2(J)	MAIN	572
020		IF (J=1) 1100,1100,1062	MAIN	573
	1062	M(J)M2(J)	MAIN	574
		M(J)M2(J)	MAIN	575
		M(J)M2(J)	MAIN	576
		M(J)M2(J)	MAIN	577
025		M(J)M2(J)	MAIN	578
		M2MAX=0	MAIN	579
		GO TO 1000 L=1,MAX	MAIN	580
	1000	M2MAX=101(2,100)/M2(1)	MAIN	581
		M2MAX=0/200	MAIN	582
		M2MAX=101(2,100)	MAIN	583
030		GO TO 1000 L=1,MAX	MAIN	584
		M2MAX=101(2,100)	MAIN	585
	1000	M2MAX=101(2,100)/M2(1)	MAIN	586
		M2MAX=0	MAIN	587
035		GO TO 1000 L=1,MAX	MAIN	588
		GO TO 1000 L=1,MAX	MAIN	589
	1000	M2MAX=101(2,100)/M2(1)	MAIN	590
		M2MAX=101(2,100)/M2(1)	MAIN	591
		M2MAX=101(2,100)/M2(1)	MAIN	592
040	1100	CONTINUE	MAIN	593
		IF (M2MAX,GT,1) CALL SHUCKE	MAIN	594
		ICOUNT=ICOUNT+1	MAIN	595
		IF ((IFLAG,NE,0).AND.(IPART,NE,0))CALL PUNT	MAIN	596
		M2MAX=0	MAIN	597
045		CONTINUE	MAIN	598
		ICOUNT=0	MAIN	599
		IF (M2MAX,GT,1) CALL SHUCKE	MAIN	600
	201	IF (M2MAX,GT,1) CALL SHUCKE	MAIN	601
		M2MAX=0	MAIN	602
050	1110	M2MAX=0	MAIN	603
		CALL PUTOUT(3)	MAIN	604
		IF ((IFLAG,NE,0).AND.(IPART,NE,0))CALL PUTOUT(3)	MAIN	605
		GO TO 200	MAIN	606
	1150	IF (M2MAX,GT,1) CALL SHUCKE	MAIN	607
		IFLAG=1	MAIN	608
055		GO TO 500	MAIN	609
		ENTER SUBROUTINE	MAIN	610
	0010	M2MAX=0	MAIN	611
	0055	M2MAX=0	MAIN	612
	0000	IF (M2MAX,GT,1) GO TO 0000	MAIN	613
060		WRITE (M2MAX) (M2MAX),M2MAX	MAIN	614
		WRITE (M2MAX) (M2MAX),M2MAX	MAIN	615
		CALL PUTOUT(3)	MAIN	616
		GO TO 0000	MAIN	617
065		IFLAG=1	MAIN	618
		GO TO 0000	MAIN	619
		M2MAX=0	MAIN	620
		M2MAX=0	MAIN	621
		M2MAX=0	MAIN	622
070	0000	WRITE (M2MAX) (M2MAX),M2MAX	MAIN	623
		WRITE (M2MAX) (M2MAX),M2MAX	MAIN	624
		WRITE (M2MAX) (M2MAX),M2MAX	MAIN	625
		WRITE (M2MAX) (M2MAX),M2MAX	MAIN	626
		WRITE (M2MAX) (M2MAX),M2MAX	MAIN	627
		WRITE (M2MAX) (M2MAX),M2MAX	MAIN	628
075	0000	WRITE (M2MAX) (M2MAX),M2MAX	MAIN	629
	0005	WRITE (M2MAX) (M2MAX),M2MAX	MAIN	630
		WRITE (M2MAX) (M2MAX),M2MAX	MAIN	631
		WRITE (M2MAX) (M2MAX),M2MAX	MAIN	632
080		CALL PUTOUT(3)	MAIN	633
		END	MAIN	634

[illegible]

195	P IUTANATAN((H0(N01)-H0(N1))/(H0(N01)-H0(N1)))	BNDRY	148
	IF (GAD0) 3700,3700,3720	BNDRY	149
197	X0000(N)+.5*(X0(N01)-X0(N1))+ELL2*81N(F IUTA)	BNDRY	150
	X0000(N)+.5*(X0(N01)-X0(N1))-ELL2*81N(F IUTA)	BNDRY	151
	GO TO 3800	BNDRY	152
200	5700 X0000(N)+.5*(X0(N01)-X0(N1))-ELL2*81N(F IUTA)	BNDRY	153
	X0000(N)+.5*(X0(N01)-X0(N1))+ELL2*81N(F IUTA)	BNDRY	154
	5800 L01	BNDRY	155
	ELL200(N)+ELL200(N1)+COS(PH1(N1))	BNDRY	156
	IF (HADD) 3020,3020,3040	BNDRY	157
205	3020 AMH000=SUM1(HADD002*(ELL200(N1)+ELL200(N2)))	BNDRY	158
	GO TO 3040	BNDRY	159
	3040 AMH000=SUM1(HADD002*(ELL200(N1)+ELL200(N2)))	BNDRY	160
	3060 PH1000=.5*(PH12(N1)+ATAN((ELL200(N1)/(H0-AMH)))	BNDRY	161
	TPH000=ATAN(PH1000)	BNDRY	162
210	GE1(N2(N1)-H0(N1)+ELL200(N1))/TPH000)002=HADD002*(ELL200(N1)+ELL200(N2))	BNDRY	163
	OM000(ELL200(N1)/SIN(ELL200(N1)+ELL200(N2)))	BNDRY	164
	IF (HADD) 3010,3010,3030	BNDRY	165
	3030 OM000=OM000	BNDRY	166
	3010 DPH000=.5*((AMH000)-(ELL200(N1)+ELL200(N2))/HADD002	BNDRY	167
215	TPH000=DPH000/(COS(PH1000))002	BNDRY	168
	GE1(N2(N1)+((ELL200(N1)-H0(N1))-H0(N1)+ELL200(N1))/TPH000)001,	BNDRY	169
	TPH000 -(ELL200(N1)+ELL200(N2))/TPH000/(ATAN(PH1000))002))	BNDRY	170
	OM000=GE1(OM000)	BNDRY	171
	ELL200	BNDRY	172
220	ELL200=ELL200	BNDRY	173
	IF (HADD) 3020,3020,3040	BNDRY	174
	3020 AMH000=SUM1(HADD002*(ELL200(N1)+ELL200(N2)))	BNDRY	175
	GO TO 3040	BNDRY	176
	3040 AMH000=SUM1(HADD002*(ELL200(N1)+ELL200(N2)))	BNDRY	177
225	4000 IF (L=0) 3000,4000,4100	BNDRY	178
	4100 X2(N)=ELL200	BNDRY	179
	X2(N)=AMH	BNDRY	180
	PH12(N)=ATAN((ELL200(N1)/(AMH000))	BNDRY	181
230	4200 U=SGN((X2(N)-X(N))002*(X2(N)-X(N))002)	BNDRY	182
	IF (ABS(PH12(N)-PH1(N1))-1.000) 4300,4300,4400	BNDRY	183
	4300 OLS(N)=0	BNDRY	184
	L=0 4500	BNDRY	185
	4500 OLS(N)=O+(PH12(N)-PH1(N1))/(2.0*(PH12(N)-PH1(N1)))	BNDRY	186
	4600 PH000=((X2(N)-X(N))002*(X2(N)-X(N))002)	BNDRY	187
235	IF (ABS(PH12(N)-PH1(N1))-1.000) 4600,4600,4700	BNDRY	188
	4600 DELT(N)=0	BNDRY	189
	GO TO 4800	BNDRY	190
	4700 DELT(N)=PH12(N)-PH1(N1)/(2.0*(PH12(N)-PH1(N1)))	BNDRY	191
	4800 AAT(N)=X(N)+X2(N)+DELT(N)+DELT(N)+DELT(N)	BNDRY	192
240	X2(N)=X(N)+X2(N)+DELT(N)+DELT(N)+DELT(N)	BNDRY	193
	4900 AAT(N)=X(N)+X2(N)+DELT(N)+DELT(N)+DELT(N)	BNDRY	194
	4950 XAT(N)	BNDRY	195
	4950 XAT(N)	BNDRY	196
	GO TO 10000	BNDRY	197
245	5000 XAT(N)=XAT(N)	BNDRY	198
	5000 XAT(N)	BNDRY	199
	PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	200
	GO TO 5700	BNDRY	201
250	5700 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	202
	5800 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	203
	5900 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	204
	6000 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	205
	6100 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	206
255	6200 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	207
	6300 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	208
	6400 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	209
	6500 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	210
	6600 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	211
260	6700 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	212
	6800 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	213
	6900 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	214
	7000 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	215
	7100 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	216
	7200 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	217
265	7300 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	218
	7400 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	219
	7500 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	220
	7600 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	221
	7700 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	222
270	7800 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	223
	7900 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	224
	8000 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	225
	8100 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	226
	8200 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	227
275	8300 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	228
	8400 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	229
	8500 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	230
	8600 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	231
280	8700 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	232
	8800 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	233
	8900 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	234
	9000 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	235
	9100 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	236
285	9200 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	237
	9300 PH000=PH(N1)+PH(N01)+PH(N1)+(S2X-SB(N1))/(SB(N01)-SB(N1))	BNDRY	238

END

BNDRY 239

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1      SUBROUTINE CHEM(IKINE)
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100    ,P1M(40) ,P1M(40) ,P1M(40) ,P1M(40) ,P1M(40) A 100

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100	005 HAFORC(IN,1)=EXP(RC(IR,5)/HRT)	CHEM	52
	GO TO 009	CHEM	53
	006 HAFORC(IN,1)=EXP(RC(IR,5)/HRT)/T(R)	CHEM	54
	GO TO 009	CHEM	55
	007 HAFORC(IN,1)/T(R)/DORT(T(R))	CHEM	56
	009 CONTINUE	CHEM	57
105	A1MM=MM(IN)	CHEM	58
	GO TO (051,052,053,054,055,056,057,058,059,000),A1MM	CHEM	59
	051 J1=IMM(IN,1)	CHEM	60
	J2=IMM(IN,2)	CHEM	61
	J3=IMM(IN,3)	CHEM	62
	J4=IMM(IN,4)	CHEM	63
110	LE=(J1)+G(J2)-G(J3)-G(J4)/HRT	CHEM	64
	IF (LE,LT,-40.0) LE=40.0	CHEM	65
	IF (LE,GT,+40.0) LE=40.0	CHEM	66
	LE=AP(LE)	CHEM	67
115	CHHAF=AMMU(R)	CHEM	68
	HP(IN)=CHHAF(J1)+F(J2)	CHEM	69
	HM(IN)=CHHAF(J3)+F(J4)/E	CHEM	70
	DO 773 J=1,4	CHEM	71
	SIGN=1.0	CHEM	72
120	IF (J,GT,2) SIGN=-1.0	CHEM	73
	IMM=IMM(IN,J)	CHEM	74
	LM(IN,J)=CM(IN,J)+SIGN*CHHAF(J2)	CHEM	75
	CM(IN,J)=CM(IN,J)+SIGN*CHHAF(J1)	CHEM	76
	LM(IN,J)=CM(IN,J)-SIGN*CHHAF(J4)/E	CHEM	77
125	CM(IN,J)=CM(IN,J)-SIGN*CHHAF(J3)/E	CHEM	78
	UX(IN)=UX(IN)+SIGN*(HP(IN)-HM(IN))	CHEM	79
	773 CONTINUE	CHEM	80
	GO TO 007	CHEM	81
	052 J1=IMM(IN,1)	CHEM	82
130	J2=IMM(IN,2)	CHEM	83
	J3=IMM(IN,3)	CHEM	84
	LE=(J1)+G(J2)-G(J3)/HRT	CHEM	85
	IF (LE,LT,-40.0) LE=40.0	CHEM	86
	IF (LE,GT,+40.0) LE=40.0	CHEM	87
135	LE=AP(LE)	CHEM	88
	CHHAF=AMMU(R)/240.0	CHEM	89
	HP(IN)=CHHAF(J1)+F(J2)	CHEM	90
	HM(IN)=CHHAF(J3)/E	CHEM	91
	DO 774 J=1,3	CHEM	92
	SIGN=1.0	CHEM	93
140	IF (J,GT,2) SIGN=-1.0	CHEM	94
	IMM=IMM(IN,J)	CHEM	95
	CM(IN,J)=CM(IN,J)+SIGN*CHHAF(J2)	CHEM	96
	CM(IN,J)=CM(IN,J)+SIGN*CHHAF(J1)	CHEM	97
145	CM(IN,J)=CM(IN,J)-SIGN*CHHAF(J4)/E	CHEM	98
	UX(IN)=UX(IN)+SIGN*HP(IN)	CHEM	99
	774 CONTINUE	CHEM	100
	GO TO 006	CHEM	101
	053 J1=IMM(IN,1)	CHEM	102
150	J2=IMM(IN,2)	CHEM	103
	J3=IMM(IN,3)	CHEM	104
	J4=IMM(IN,4)	CHEM	105
	J5=IMM(IN,5)	CHEM	106
	LE=(J1)+G(J2)-G(J3)-G(J4)-G(J5)/HRT	CHEM	107
155	IF (LE,LT,-40.0) LE=40.0	CHEM	108
	IF (LE,GT,+40.0) LE=40.0	CHEM	109
	LE=AP(LE)	CHEM	110
	CHHAF=AMMU(R)	CHEM	111
	HP(IN)=CHHAF(J1)+F(J2)	CHEM	112
160	HM(IN)=CHHAF(J3)+F(J4)+F(J5)+AMMU(R)*HRT1/E	CHEM	113
	DO 775 J=1,5	CHEM	114
	SIGN=1.0	CHEM	115
	IF (J,GT,2) SIGN=-1.0	CHEM	116
	IMM=IMM(IN,J)	CHEM	117
165	CM(IN,J)=CM(IN,J)+SIGN*CHHAF(J2)	CHEM	118
	CM(IN,J)=CM(IN,J)+SIGN*CHHAF(J1)	CHEM	119
	CM(IN,J)=CM(IN,J)-SIGN*CHHAF(J4)+F(J5)+AMMU(R)*HRT1/E	CHEM	120
	CM(IN,J)=CM(IN,J)-SIGN*CHHAF(J3)+F(J5)+AMMU(R)*HRT1/E	CHEM	121
	CM(IN,J)=CM(IN,J)-SIGN*CHHAF(J5)+F(J4)+F(J5)+AMMU(R)*HRT1/E	CHEM	122
170	UX(IN)=UX(IN)+SIGN*(HP(IN)-2.0*HM(IN))	CHEM	123
	775 CONTINUE	CHEM	124
	GO TO 001	CHEM	125
	054 J1=IMM(IN,1)	CHEM	126
175	J2=IMM(IN,2)	CHEM	127
	J3=IMM(IN,3)	CHEM	128
	LE=(J1)+G(J2)-G(J3)/HRT	CHEM	129
	IF (LE,LT,-40.0) LE=40.0	CHEM	130
	IF (LE,GT,+40.0) LE=40.0	CHEM	131
	LE=AP(LE)	CHEM	132
180	CHHAF=AMMU(R)	CHEM	133
	HP(IN)=CHHAF(J1)+F(J2)	CHEM	134
	HM(IN)=CHHAF(J3)/E	CHEM	135
	DO 776 J=1,3	CHEM	136
	SIGN=1.0	CHEM	137
	IF (J,GT,2) SIGN=-1.0	CHEM	138
	IMM=IMM(IN,J)	CHEM	139
	CM(IN,J)=CM(IN,J)+SIGN*CHHAF(J2)	CHEM	140
	CM(IN,J)=CM(IN,J)+SIGN*CHHAF(J1)	CHEM	141
	CM(IN,J)=CM(IN,J)-SIGN*CHHAF(J4)/E	CHEM	142
185	UX(IN)=UX(IN)+SIGN*HP(IN)	CHEM	143
	776 CONTINUE	CHEM	144
	GO TO 000	CHEM	145
	055 J1=IMM(IN,1)	CHEM	146

195	J2=ALB+1	CHEM	147
	J3=INNN(IH,3)	CHEM	148
	J4=INNN(IH,4)	CHEM	149
	EM(I,J)=G(J3)-G(J4)/MMT	CHEM	150
	IF (E .LT. +.00,0) E=+.00,0	CHEM	151
	IF (E .GT. +.00,0) E=+.00,0	CHEM	152
200	ENDP IZ	CHEM	153
	CH=ANAL*MMU(K)/ZPA	CHEM	154
	MM(IH)=CH*OF1(J1)	CHEM	155
	MM(IH)=CH*MMMT1+MMU(K)*OF1(J3)+OF1(J4)/E	CHEM	156
	DU 772 J41,4	CHEM	157
205	IF (J .EQ. 2) GO TO 772	CHEM	158
	SIGN=1.0	CHEM	159
	IF (J .GT. 2) SIGN=-1.0	CHEM	160
	MM=INNN(IH,J)	CHEM	161
	CM(IHUN,J)=CM(IHUN,J1)+SIGN*CH	CHEM	162
210	CM(IHUN,J3)=CM(IHUN,J3)-SIGN*CH*MMMT1+MMU(K)*OF1(J4)/E	CHEM	163
	CM(IHUN,J4)=CM(IHUN,J4)-SIGN*CH*MMMT1+MMU(K)*OF1(J3)/E	CHEM	164
	MM(IHUN)=MM(IHUN)+SIGN*MM(IH)	CHEM	165
	CONTINUE	CHEM	166
	GO TO 807	CHEM	167
215	J1=INNN(IH,1)	CHEM	168
	J2=INNN(IH,2)	CHEM	169
	J3=INNN(IH,3)	CHEM	170
	J4=INNN(IH,4)	CHEM	171
	CH=ANAL*MMU(K)	CHEM	172
220	MM(IH)=CH*OF1(J1)+OF1(J2)	CHEM	173
	MM(IH)=0.0	CHEM	174
	DU 770 J41,4	CHEM	175
	SIGN=1.0	CHEM	176
	IF (J .GT. 2) SIGN=-1.0	CHEM	177
225	MM=INNN(IH,J)	CHEM	178
	CM(IHUN,J1)=CM(IHUN,J1)+SIGN*CH*MMU(K)+OF1(J2)	CHEM	179
	CM(IHUN,J2)=CM(IHUN,J2)+SIGN*CH*MMU(K)+OF1(J1)	CHEM	180
	MM(IHUN)=MM(IHUN)+SIGN*MM(IH)	CHEM	181
230	CONTINUE	CHEM	182
	GO TO 807	CHEM	183
235	J1=INNN(IH,1)	CHEM	184
	J2=INNN(IH,2)	CHEM	185
	J3=INNN(IH,3)	CHEM	186
	CH=ANAL*MMU(K)*PAV/ZPA	CHEM	187
	MM(IH)=CH*MMMT1+OF1(J1)+OF1(J2)	CHEM	188
	MM(IH)=0.0	CHEM	189
	DU 779 J41,3	CHEM	190
	SIGN=1.0	CHEM	191
	IF (J .GT. 2) SIGN=-1.0	CHEM	192
240	MM=INNN(IH,J)	CHEM	193
	CM(IHUN,J1)=CM(IHUN,J1)+SIGN*CH*MMU(K)+OF1(J2)	CHEM	194
	CM(IHUN,J2)=CM(IHUN,J2)+SIGN*CH*MMU(K)+OF1(J1)	CHEM	195
	MM(IHUN)=MM(IHUN)+SIGN*MM(IH)	CHEM	196
	CONTINUE	CHEM	197
245	GO TO 808	CHEM	198
250	J1=INNN(IH,1)	CHEM	199
	J2=INNN(IH,2)	CHEM	200
	J3=INNN(IH,3)	CHEM	201
	J4=INNN(IH,4)	CHEM	202
255	J5=INNN(IH,5)	CHEM	203
	CH=ANAL*MMU(K)	CHEM	204
	MM(IH)=CH*OF1(J1)+OF1(J2)	CHEM	205
	MM(IH)=0.0	CHEM	206
	DU 780 J41,5	CHEM	207
255	SIGN=1.0	CHEM	208
	IF (J .GT. 2) SIGN=-1.0	CHEM	209
	MM=INNN(IH,J)	CHEM	210
	CM(IHUN,J1)=CM(IHUN,J1)+SIGN*CH*MMU(K)+OF1(J2)	CHEM	211
	CM(IHUN,J2)=CM(IHUN,J2)+SIGN*CH*MMU(K)+OF1(J1)	CHEM	212
260	MM(IHUN)=MM(IHUN)+SIGN*MM(IH)	CHEM	213
	CONTINUE	CHEM	214
	GO TO 801	CHEM	215
265	J1=INNN(IH,1)	CHEM	216
	J2=INNN(IH,2)	CHEM	217
	J3=INNN(IH,3)	CHEM	218
	CH=ANAL*MMU(K)	CHEM	219
	MM(IH)=CH*MMMT1+OF1(J1)+OF1(J2)	CHEM	220
	MM(IH)=0.0	CHEM	221
	DU 776 J41,3	CHEM	222
270	SIGN=1.0	CHEM	223
	IF (J .GT. 2) SIGN=-1.0	CHEM	224
	MM=INNN(IH,J)	CHEM	225
	CM(IHUN,J1)=CM(IHUN,J1)+SIGN*CH*MMU(K)+OF1(J2)	CHEM	226
	CM(IHUN,J2)=CM(IHUN,J2)+SIGN*CH*MMU(K)+OF1(J1)	CHEM	227
275	MM(IHUN)=MM(IHUN)+SIGN*MM(IH)	CHEM	228
	CONTINUE	CHEM	229
	GO TO 806	CHEM	230
280	J1=INNN(IH,1)	CHEM	231
	J2=INNN(IH,2)	CHEM	232
	J3=INNN(IH,3)	CHEM	233
	J4=INNN(IH,4)	CHEM	234
	CH=ANAL*MMU(K)/ZPA	CHEM	235
	MM(IH)=CH*OF1(J1)	CHEM	236
	MM(IH)=0.0	CHEM	237
285	DU 777 J41,4	CHEM	238
	IF (J .EQ. 2) GO TO 777	CHEM	239
	SIGN=1.0	CHEM	240
	IF (J .GT. 2) SIGN=-1.0	CHEM	241

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      INUM = IRRR(IR,J)
      CP(IIRN,J)=CM(IIRN,J)+SIGN*CHN
299  CONTINUE
      GO TO 867
      861 NP(J5)NP(J5)+NP(IR)
      NM(J5)NM(J5)+NM(IR)
295  867 NP(J6)NP(J6)+NP(IR)
      NM(J6)NM(J6)+NM(IR)
      868 NP(J3)NP(J3)+NP(IR)
      NM(J3)NM(J3)+NP(IR)
      NP(J2)NP(J2)+NP(IR)
      NM(J2)NM(J2)+NP(IR)
300  NP(J1)NP(J1)+NP(IR)
      NM(J1)NM(J1)+NP(IR)
      1 CONTINUE
      DO 897 J=1,NUS
      PRODUCTION RATE IN MOLE/GM=CM
      ADDT(J,N)=NP(J)-NM(J)/U(R)
      897 CONTINUE
      USUR=DLB(R)/U(R)
      DO 10 I=1,NUS
      USUR=USUR+PA(I,K)
      QX(I,N)= ((I,K,A)+KMS(I,N)/MDUT(R)+USUR*RAUT(I,N)
      L0 11 J=1,NUS
      CM(I,N,JR)= CM(I,K,JR)+USUR*
      PA (I,N,EU,JR) CM(I,N,JR)= 1.0+CM(I,N,JR)
315  11 CONTINUE
      IF (IBUGBN.NE.0)
      WRITE(6,100) R,IR,(CM(I,N,JR),J=1,NUS),UX(IR)
100  FORMAT(1X,215,1PDE12,3)
      10 CONTINUE
      CALL SLOP(QX,L,NUS)
      DO 12 I=1,NUS
      IF (IBUGBN.NE.0)
      WRITE(6,100) R,IR,(CM(I,N,JR),J=1,NUS),QX(I,N)
      C2(I,N,R)= QX(I,N)
325  C CALCULATE NEW MASS FRACTION OF SPECIES
      12 CONTINUE
      RETURN
      END

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CHEM 242
CHEM 243
CHEM 244
CHEM 245
CHEM 246
CHEM 247
CHEM 248
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CHEM 251
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CHEM 276
CHEM 277
CHEM 278
CHEM 279
CHEM 280
CHEM 281

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1  C SUM=0; TIME CPU=INT(IUENT,IKINE,NUS)
      .....
      DIMENSION IUENT(25),IZU(5)
      COMMON/CHEM1/ Z1(15), IIRN(40),INT(40),HC(40,3),IMRN(40,5),
5  1 AV,CM(26,26),F1(26),NP(26),
      2 NM(26),ADDT(26,40),LSI(26),QX(26)
      LUMH=AVYZ/INTZ,JATZ
      JATZ=0
      JATZ=0
10  AV=0,USEZ
      DO 2 I=1,IKINE
      READ (5,100) (IZU(J),J=1,5),IMRN(1),INT(1),(NC(1,AM),AM=1,3)
      WRITE(6,101) 1, (IZU(J),J=1,5),IMRN(1),INT(1),(NC(1,AM),AM=1,3)
      DO 5 J=1,5
15  IMRN (1,J)=0
      DO 3 L=1,NUS
      3 (IZU(J)=IDENT(L)) 3,4,3
      4 IMRN(1,J)=L
      3 CONTINUE
      5 CONTINUE
      5 CONTINUE
      2 CONTINUE
100  FORMAT (A4,3I,4A,10X,A4,3I,4A,4 12,1I,4A,2F4,1,49,1)
101  FORMAT (2X,12,2I,4A,3I,4A,10X,A4,3I,4A,4I,4A,9I,12,1I,4A,2F
25  1 F4,1,F4,1)
      RETURN
      END

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CPUTIN 2
CPUTIN 3
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CPUTIN 25
CPUTIN 26
CPUTIN 27
CPUTIN 28

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1  SUBROUTINE UNAG
      COMMON A(25,7)
      1 A(10) ,ALFA(25,40),ALPHAH ,ALPHAY ,AHAG
      2 A(10) ,BETAP ,BETA ,C11(25) ,C(25,40)
5  3 A(10) ,C2(25,40) ,CABAH(25) ,CBAB(25) ,CP
      4 A(10) ,CPM ,CPM ,CSM(25) ,CSM1(25) ,CSTREM(25)
      5 A(10) ,P2IN(25,25),DEFF(25) ,DEFF(25) ,DEFLA
      6 A(10) ,LELSS(40) ,DELS ,DELSO ,ULS(40)
      7 A(10) ,P12 ,DEFLY(40) ,DIS ,DUPY(40)
10  8 A(10) ,DPMIUS(10) ,EPCUN
      9 A(10) ,EXTNA(50) ,FSTEP ,FMAX ,GNAD
      10 A(10) ,H1 ,H(40) ,HJ ,HNM(25)
      11 A(10) ,H2PM(25) ,H3PM(25) ,ICUNST ,ICOUNT ,IOENT(25)
      12 A(10) ,LENNCH ,EXTNA(50) ,IFLAG ,IIND
      13 A(10) ,IPILL ,ISMUCK ,ITPI ,IPU
15  14 A(10) ,IP1FF ,K ,KAY ,KAYS ,KATZ
      15 A(10) ,KAX ,KAY
      16 A(10) ,KAP ,KAX ,LL ,LPLANE ,MA
      17 A(10) ,KASH ,KUT ,KMAX ,NUO ,MU
      18 A(10) ,K2 ,KUS ,KX ,KMX ,PMH
20  19 A(10) ,KXUUL ,NDS ,NITEM ,KMAX ,KN
      20 A(10) ,KXUUL ,NDS ,NITEM ,KMAX ,KN
      21 A(10) ,KXUUL ,NDS ,NITEM ,KMAX ,KN
      22 A(10) ,KXUUL ,NDS ,NITEM ,KMAX ,KN
      23 A(10) ,KXUUL ,NDS ,NITEM ,KMAX ,KN
25  24 A(10) ,KXUUL ,NDS ,NITEM ,KMAX ,KN
      25 A(10) ,KXUUL ,NDS ,NITEM ,KMAX ,KN
      26 A(10) ,KXUUL ,NDS ,NITEM ,KMAX ,KN
      27 A(10) ,KXUUL ,NDS ,NITEM ,KMAX ,KN
      28 A(10) ,KXUUL ,NDS ,NITEM ,KMAX ,KN

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      INUM = IRRR(IR,J)
240  C(IRON,J)=C(IIRON,J)+SIGN=CHN
      CONTINUE
      GO TO 807
245  801  NP(J5)NP(J5)=NP(IR)
      NM(J5)NM(J5)=NM(IR)
      807  NP(J4)NP(J4)=NP(IR)
      NM(J4)NM(J4)=NM(IR)
250  806  NP(J3)NP(J3)=NP(IR)
      NM(J3)NM(J3)=NM(IR)
      NP(J2)NP(J2)=NP(IR)
      NM(J2)NM(J2)=NM(IR)
300  805  NP(J1)NP(J1)=NP(IR)
      NM(J1)NM(J1)=NM(IR)
      1 CONTINUE
      DU 807 J01,NUS
305  C PRODUCTION RATE IN MOLE/GM-CM
      ADD(J,K)=NP(J)-NM(J)/U(K)
      807 CONTINUE
      USUM=US(K)/L(K)
      DU 10 IN=1,NUS
310  USUM=USUM+PA(I,K)
      QX(IN)= ((I,K)*NMS(IN)/MU(I(K))+USUM*AAUT(I,K))
      L1 11 JH=1,NUS
      CM(IN,JH)=CM(I,K,JH)+USUM
      P (IN,L1,JH) L1(IN,JH)=1.0+CM(IN,JH)
315  11 CONTINUE
      IF (IDUGSH .NE. 0)
      *WRITE(6,100) K,IN,(CM(IN,JH),JH=1,NUS),QX(IN)
      100 FORMAT(1X,215,1PBE12,3)
      10 CONTINUE
      CALL SLOW(QX,LP,NDS)
      DU 12 IN=1,NUS
320  IF (IDUGSH .NE. 0)
      *WRITE(6,100) K,IN,(CM(IN,JH),JH=1,NUS),QX(IN)
      C2(IN,K)= QX(IN)
325  C CALCULATE NEW MASS FRACTION OF SPECIES
      12 CONTINUE
      RETURN
      END

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      CMH  202
      CMH  203
      CMH  204
      CMH  205
      CMH  206
      CMH  207
      CMH  208
      CMH  209
      CMH  250
      CMH  251
      CMH  252
      CMH  253
      CMH  254
      CMH  255
      CMH  256
      CMH  257
      CMH  258
      CMH  259
      CMH  260
      CMH  261
      CMH  262
      CMH  263
      CMH  264
      CMH  265
      CMH  266
      CMH  267
      CMH  268
      CMH  269
      CMH  270
      CMH  271
      CMH  272
      CMH  273
      CMH  274
      CMH  275
      CMH  276
      CMH  277
      CMH  278
      CMH  279
      CMH  280
      CMH  281

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      SUBROUTINE CPUTIN(JUENT,IRINE,NUS)
      .....
      DIMENSION JUENT (25),IZU(5)
      COMMON/CHEN1/ ZI(5), IRR(40),INT(40),HC(40,5),INRR(40,5),
3  1 AV,CM(20,20),F1(20),NP(20),
      2 NM(20),ADD(20,40),CS(20),QX(20)
      LUM=UN/XYZ/1P2,XYZ
      JAYZ=4
      JXYZ=0
      AV=0.03E21
      DU 2 IN=1,IRINE
10  HEAD (5,100) (1ZU(J),JH=5),INR(1),INT(1),INC(1,NR),NR=1,5)
      *WRITE(6,101) 1, (1ZU(J),JH=5),INR(1),INT(1),INC(1,NR),NR=1,5)
      DU 5 JH=5
15  INRR (1,JH)
      DU 3 L=1,NDS
      10 (1ZU(J)=IDENI(L)) 3,4,5
      4 INRR(1,JH)
      3 CONTINUE
      5 CONTINUE
      HC(1,1)=NC(1,1)+AV
      2 CONTINUE
100 FORMAT (1A,3X,A0,10X,A0,3X,A0,3X,A0,9 12,11,1A,2,F4,1,F9,1)
101 FORMAT (2X,12,2V,A0,3X,A0,10X,A0,3X,4, 3X,A0,9X,12,11,1A,2,
25 1 F4,1,F9,1)
      RETURN
      END

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      CPUTIN 2
      CPUTIN 3
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      CPUTIN 28

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      SUBROUTINE UNAG
      COMMON A(25,7) ,AA(40) ,ALFA(25,2),ALPHAH ,ALPHAH
1  ,ALU ,DELTA ,OMEGA ,C(11(25) ,C(25,40)
2  ,C(12(25) ,C2(25,40) ,LABAH(25) ,CUBAH(25) ,CP
3  ,CPS(25) ,CPSH ,CSH(25) ,LSH(1(25) ,CSTREH(25)
4  ,XSTIR ,DPIH(25,25),DIFF(25) ,DEFF(25) ,DEFLA
5  ,D11 ,DLS(40) ,DLS ,DLS(40)
6  ,D1H(25,25) ,D12 ,DLY(40) ,D15 ,DUDY(40)
7  ,D14 ,DPH(10,10) ,DPCU ,
10  ,DPELON ,DXTA(50) ,DSTLP ,DPMAX ,DGHAD
      ,D11 ,D(40) ,DMM ,DMD ,DMM(25)
4  ,DMPH(25) ,DMPH(25) ,DILUSI ,DIDUNI ,DIDENI(25)
1  ,DPMH ,DXTA(50) ,DPLAG ,DIND
2  ,DPLC ,DLMUCK ,D11PI ,DPU
15  3 ,D11PI ,D ,DAYS ,DAYS ,DAYS2
4  ,D1U ,DPMAX ,D ,DPLANE ,DMA
5  ,DUP ,D ,D ,D ,D
6  ,DASH ,D1U ,DPMAX ,DUD ,D ,D
7  ,D12 ,DUS ,D ,D ,DASH
20  8 ,DADUAL ,DND ,D11EN ,DMMAX ,DNN
      LUMUN NUCASE ,OMEGA(25) ,D11 ,D(40) ,D12
1  ,D2(40) ,DPM(50) ,DPMAR ,DPMAR
2  ,DMS ,D15 ,DPM(40) ,DPM(50) ,DPM(50)
3  ,DPM(50) ,DPM(50) ,DPM(50) ,DPM(50)
25  4 ,DPM(40) ,DPM(50) ,DPM(50) ,DPM(50)
5  ,DPM ,DPM ,DPM(50) ,DPM(50)
6  ,D11 ,D(40) ,DXTM1 ,DXTM2
7  ,D(50) ,D11 ,D(40) ,D(50) ,D(50)

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      UNAG 2
      UNAG 3
      UNAG 4
      UNAG 5
      UNAG 6
      UNAG 7
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      UNAG 24
      UNAG 25
      UNAG 26
      UNAG 27
      UNAG 28

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Line	Code	Text	Column
75	20	DU 20 101,403	FLUX 25
	20	2J(1,1)03,0	FLUX 25
	20	DO 10 10000	FLUX 26
	1	----- 36 TOP OF AVERAGE FLUX QUANTITIES NEEDED -----	FLUX 27
	30	IF (MATH)0000	FLUX 28
	30	IF (MATH)0000	FLUX 29
	30	IF (MATH)0000	FLUX 30
	30	IF (MATH)0000	FLUX 31
	30	IF (MATH)0000	FLUX 32
	30	IF (MATH)0000	FLUX 33
	30	IF (MATH)0000	FLUX 34
	30	IF (MATH)0000	FLUX 35
	30	IF (MATH)0000	FLUX 36
	30	IF (MATH)0000	FLUX 37
	30	IF (MATH)0000	FLUX 38
	30	IF (MATH)0000	FLUX 39
	30	IF (MATH)0000	FLUX 40
	30	IF (MATH)0000	FLUX 41
	30	IF (MATH)0000	FLUX 42
	30	IF (MATH)0000	FLUX 43
	30	IF (MATH)0000	FLUX 44
	30	IF (MATH)0000	FLUX 45
	30	IF (MATH)0000	FLUX 46
	30	IF (MATH)0000	FLUX 47
	30	IF (MATH)0000	FLUX 48
	30	IF (MATH)0000	FLUX 49
	30	IF (MATH)0000	FLUX 50
	30	IF (MATH)0000	FLUX 51
	30	IF (MATH)0000	FLUX 52
	30	IF (MATH)0000	FLUX 53
	30	IF (MATH)0000	FLUX 54
	30	IF (MATH)0000	FLUX 55
	30	IF (MATH)0000	FLUX 56
	30	IF (MATH)0000	FLUX 57
	30	IF (MATH)0000	FLUX 58
	30	IF (MATH)0000	FLUX 59
	30	IF (MATH)0000	FLUX 60
	30	IF (MATH)0000	FLUX 61
	30	IF (MATH)0000	FLUX 62
	30	IF (MATH)0000	FLUX 63
	30	IF (MATH)0000	FLUX 64
	30	IF (MATH)0000	FLUX 65
	30	IF (MATH)0000	FLUX 66
	30	IF (MATH)0000	FLUX 67
	30	IF (MATH)0000	FLUX 68
	30	IF (MATH)0000	FLUX 69
	30	IF (MATH)0000	FLUX 70
	30	IF (MATH)0000	FLUX 71
	30	IF (MATH)0000	FLUX 72
	30	IF (MATH)0000	FLUX 73
	30	IF (MATH)0000	FLUX 74
	30	IF (MATH)0000	FLUX 75
	30	IF (MATH)0000	FLUX 76
	30	IF (MATH)0000	FLUX 77
	30	IF (MATH)0000	FLUX 78
	30	IF (MATH)0000	FLUX 79
	30	IF (MATH)0000	FLUX 80
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	30	IF (MATH)0000	FLUX 82
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	30	IF (MATH)0000	FLUX 84
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	30	IF (MATH)0000	FLUX 88
	30	IF (MATH)0000	FLUX 89
	30	IF (MATH)0000	FLUX 90
	30	IF (MATH)0000	FLUX 91
	30	IF (MATH)0000	FLUX 92
	30	IF (MATH)0000	FLUX 93
	30	IF (MATH)0000	FLUX 94
	30	IF (MATH)0000	FLUX 95
	30	IF (MATH)0000	FLUX 96
	30	IF (MATH)0000	FLUX 97
	30	IF (MATH)0000	FLUX 98
	30	IF (MATH)0000	FLUX 99
	30	IF (MATH)0000	FLUX 100
	30	IF (MATH)0000	FLUX 101
	30	IF (MATH)0000	FLUX 102
	30	IF (MATH)0000	FLUX 103
	30	IF (MATH)0000	FLUX 104
	30	IF (MATH)0000	FLUX 105
	30	IF (MATH)0000	FLUX 106
	30	IF (MATH)0000	FLUX 107
	30	IF (MATH)0000	FLUX 108
	30	IF (MATH)0000	FLUX 109
	30	IF (MATH)0000	FLUX 110
	30		

25	C	END	MUST BE A FIXED OR FLOATING POINT VARIABLE DESIGNATING A ONE-DIMENSIONAL ERASABLE ARRAY OF LENGTH AT LEAST EQUAL TO THE NUMBER OF ROWS IN MATRIX A. IT IS IN THIS AREA SINGE KEEPS A RECORD OF THE COLUMN PERMUTATIONS.	MATE7	25
	C		IS A FIXED POINT VARIABLE WHICH WILL BE ASSIGNED THE FIXED POINT CONSTANTS.	MATE7	26
	C		1 IF THE SOLUTION WAS SUCCESSFUL.	MATE7	27
30	C		2 IF THE MATRIX A WAS SINGULAR.	MATE7	28
	C		DIMENSIONS	MATE7	29
	C	A	LDIM = ICIM IF MM IS LESS THAN OR EQUAL TO LDIM	MATE7	30
	C		LDIM = MM IF MM IS GREATER THAN LDIM	MATE7	31
35	C	P	LDIM = MM	MATE7	32
	C	END	LDIM	MATE7	33
	C		NOTE	MATE7	34
	C		FOR MATRICES THE MM DIMENSION MUST BE THE SAME AS THAT GIVEN ABOVE. THE COLUMN DIMENSION MUST AT LEAST BE THAT GIVEN ABOVE.	MATE7	35
40	C	NNNN	FOR THE VECTOR THE DIMENSION MUST AT LEAST BE THAT GIVEN ABOVE	MATE7	36
	C		EXECUTION OF THIS ROUTINE DESTROYS THE ORIGINAL A AND N MATRICES	MATE7	37
45	C		AFTER A SUCCESSFUL EXIT FROM THIS SUBROUTINE, THE ANSWERS ON THE A MATRIX REPLACE THE A MATRIX.	MATE7	38
	C		THIS REPLACEMENT IS DONE ACCORDING TO THE SCHEME. A(I,J) IS REPLACED BY A(I,J).	MATE7	39
50	C	NNNN		MATE7	40
	C		SET INITIAL CONSTANTS	MATE7	41
	C	NNNN		MATE7	42
	C		SPECIAL CONSIDERATION WHEN THE ORDER OF MATRIX A IS 1.	MATE7	43
55	C		IF (N,NE,1) GO TO 5	MATE7	44
	C		IF (A(1,1),EW,0) GO TO 110	MATE7	45
	C		MUL(NA(1,1))	MATE7	46
	C		DU 1 INT,M	MATE7	47
	C			MATE7	48
60	C		1 A(I,1)MM(1,1)/MUL	MATE7	49
	C		DETDETMUL	MATE7	50
	C		DETMM	MATE7	51
	C		5 AM(NA)=1	MATE7	52
	C		AM(NA)=1	MATE7	53
	C		INITIALIZE DETERMINANT	MATE7	54
65	C		DETMM=1	MATE7	55
	C		INITIALIZE COLUMN INDICATORS	MATE7	56
	C		DU 5 INT,M	MATE7	57
	C		5 IND(I)=1	MATE7	58
	C		PPU(I) INTRIGULARIZATION TO PUT UPPER TRIANGLE	MATE7	59
70	C		DU 70 INT,NAI	MATE7	60
	C		MM(NAI)	MATE7	61
	C		MM(NAI)	MATE7	62
	C		MM(NAI)	MATE7	63
	C		SEARCH FOR PIVOTAL ELEMENT	MATE7	64
75	C		MM(NAI,NAI)	MATE7	65
	C		DU 10 INT,NA	MATE7	66
	C		DU 10 INT,NA	MATE7	67
	C		IF (MM(NAI,NAI),ABS(A(I,J))) GO TO 10	MATE7	68
	C		MM(NAI,NAI)	MATE7	69
	C		MM(NAI)	MATE7	70
	C		MM(NAI)	MATE7	71
	C		MM(NAI)	MATE7	72
	C		MM(NAI)	MATE7	73
80	C		MM(NAI)	MATE7	74
	C		MM(NAI)	MATE7	75
	C		MM(NAI)	MATE7	76
	C		MM(NAI)	MATE7	77
	C		MM(NAI)	MATE7	78
	C		MM(NAI)	MATE7	79
	C		MM(NAI)	MATE7	80
85	C		MM(NAI)	MATE7	81
	C		MM(NAI)	MATE7	82
	C		MM(NAI)	MATE7	83
	C		MM(NAI)	MATE7	84
	C		MM(NAI)	MATE7	85
	C		MM(NAI)	MATE7	86
	C		MM(NAI)	MATE7	87
	C		MM(NAI)	MATE7	88
	C		MM(NAI)	MATE7	89
	C		MM(NAI)	MATE7	90
90	C		MM(NAI)	MATE7	91
	C		MM(NAI)	MATE7	92
	C		MM(NAI)	MATE7	93
	C		MM(NAI)	MATE7	94
	C		MM(NAI)	MATE7	95
95	C		MM(NAI)	MATE7	96
	C		MM(NAI)	MATE7	97
	C		MM(NAI)	MATE7	98
	C		MM(NAI)	MATE7	99
	C		MM(NAI)	MATE7	100
100	C		MM(NAI)	MATE7	101
	C		MM(NAI)	MATE7	102
	C		MM(NAI)	MATE7	103
	C		MM(NAI)	MATE7	104
	C		MM(NAI)	MATE7	105
105	C		MM(NAI)	MATE7	106
	C		MM(NAI)	MATE7	107
	C		MM(NAI)	MATE7	108
	C		MM(NAI)	MATE7	109
	C		MM(NAI)	MATE7	110
110	C		MM(NAI)	MATE7	111
	C		MM(NAI)	MATE7	112
	C		MM(NAI)	MATE7	113
	C		MM(NAI)	MATE7	114
	C		MM(NAI)	MATE7	115
115	C		MM(NAI)	MATE7	116
	C		MM(NAI)	MATE7	117
	C		MM(NAI)	MATE7	118
	C		MM(NAI)	MATE7	119
120	C		MM(NAI)	MATE7	120
	C		MM(NAI)	MATE7	121
	C		MM(NAI)	MATE7	122
	C		MM(NAI)	MATE7	123
	C		MM(NAI)	MATE7	124
	C		MM(NAI)	MATE7	125
	C		MM(NAI)	MATE7	126
	C		MM(NAI)	MATE7	127
	C		MM(NAI)	MATE7	128
	C		MM(NAI)	MATE7	129
	C		MM(NAI)	MATE7	130
	C		MM(NAI)	MATE7	131
	C		MM(NAI)	MATE7	132
	C		MM(NAI)	MATE7	133
	C		MM(NAI)	MATE7	134
	C		MM(NAI)	MATE7	135
	C		MM(NAI)	MATE7	136
	C		MM(NAI)	MATE7	137
	C		MM(NAI)	MATE7	138
	C		MM(NAI)	MATE7	139
	C		MM(NAI)	MATE7	140
	C		MM(NAI)	MATE7	141
	C		MM(NAI)	MATE7	142
	C		MM(NAI)	MATE7	143
	C		MM(NAI)	MATE7	144
	C		MM(NAI)	MATE7	145
	C		MM(NAI)	MATE7	146
	C		MM(NAI)	MATE7	147
	C		MM(NAI)	MATE7	148
	C		MM(NAI)	MATE7	149
	C		MM(NAI)	MATE7	150
	C		MM(NAI)	MATE7	151
	C		MM(NAI)	MATE7	152
	C		MM(NAI)	MATE7	153
	C		MM(NAI)	MATE7	154
	C		MM(NAI)	MATE7	155
	C		MM(NAI)	MATE7	156
	C		MM(NAI)	MATE7	157
	C		MM(NAI)	MATE7	158
	C		MM(NAI)	MATE7	159
	C		MM(NAI)	MATE7	160
	C		MM(NAI)	MATE7	161
	C		MM(NAI)	MATE7	162
	C		MM(NAI)	MATE7	163
	C		MM(NAI)	MATE7	164
	C		MM(NAI)	MATE7	165
	C		MM(NAI)	MATE7	166
	C		MM(NAI)	MATE7	167
	C		MM(NAI)	MATE7	168
	C		MM(NAI)	MATE7	169
	C		MM(NAI)	MATE7	170
	C		MM(NAI)	MATE7	171
	C		MM(NAI)	MATE7	172
	C		MM(NAI)	MATE7	173
	C		MM(NAI)	MATE7	174
	C		MM(NAI)	MATE7	175
	C		MM(NAI)	MATE7	176
	C		MM(NAI)	MATE7	177
	C		MM(NAI)	MATE7	178
	C		MM(NAI)	MATE7	179
	C		MM(NAI)	MATE7	180
	C		MM(NAI)	MATE7	181
	C		MM(NAI)	MATE7	182
	C		MM(NAI)	MATE7	183
	C		MM(NAI)	MATE7	184
	C		MM(NAI)	MATE7	185
	C		MM(NAI)	MATE7	186
	C		MM(NAI)	MATE7	187
	C		MM(NAI)	MATE7	188
	C		MM(NAI)	MATE7	189
	C		MM(NAI)	MATE7	190
	C		MM(NAI)	MATE7	191
	C		MM(NAI)	MATE7	192
	C		MM(NAI)	MATE7	193
	C		MM(NAI)	MATE7	194
	C		MM(NAI)	MATE7	195
	C		MM(NAI)	MATE7	196
	C		MM(NAI)	MATE7	197
	C		MM(NAI)	MATE7	198
	C		MM(NAI)	MATE7	199
	C		MM(NAI)	MATE7	200

25	C	IND	MUST BE A FIXED OR FLOATING POINT VARIABLE DESIGNATING	MATE7	25
	C		A ONE-DIMENSIONAL ERASABLE ARRAY OF LENGTH AT LEAST	MATE7	26
	C		EQUAL TO THE NUMBER OF ROWS IN MATRIX A. IT IS IN THIS	MATE7	27
	C		AREA SAVED KEYS A RECORD OF THE COLUMN PERMUTATIONS.	MATE7	28
	C	NOGU	IS A FIXED POINT VARIABLE WHICH WILL BE ASSIGNED THE	MATE7	29
	C		FIXED POINT CONSTANTS.	MATE7	30
30	C		1 IF THE SOLUTION WAS SUCCESSFUL.	MATE7	31
	C		3 IF THE MATRIX A WAS SINGULAR.	MATE7	32
	C		DIMENSIONS	MATE7	33
	C	A	IDIM X IDIM IF NM IS LESS THAN OR EQUAL TO IDIM	MATE7	34
	C		IDIM X NM IF NM IS GREATER THAN IDIM	MATE7	35
35	C	B	IDIM X NM	MATE7	36
	C	IND	IDIM	MATE7	37
	C		NOTE	MATE7	38
	C		FOR MATRICES THE MIN DIMENSION MUST BE THE SAME AS	MATE7	39
	C		THAT GIVEN ABOVE. THE COLUMN DIMENSION MUST AT LEAST	MATE7	40
40	C		BE THAT GIVEN ABOVE.	MATE7	41
	C	NRHM	FOR THE VECTOR THE DIMENSION MUST AT LEAST BE THAT	MATE7	42
	C		GIVEN ABOVE.	MATE7	43
	C		EXECUTION OF THIS ROUTINE DESTROYS THE ORIGINAL A AND	MATE7	44
45	C		B MATRICES.	MATE7	45
	C		AFTER A SUCCESSFUL EXIT FROM THIS SUBROUTINE, THE	MATE7	46
	C		ANSWERS ON THE B MATRIX REPLACE THE A MATRIX.	MATE7	47
	C		THIS REPLACEMENT IS DONE ACCORDING TO THE SCHEME:	MATE7	48
	C		A(I,J) IS REPLACED BY B(I,J).	MATE7	49
50	C	NRHM		MATE7	50
	C		SET INITIAL CONSTANTS	MATE7	51
	C	NOGU		MATE7	52
	C		SPECIAL CONSIDERATION WHEN THE ORDER OF MATRIX A IS 1.	MATE7	53
55	C		IF (A(1,1).EQ.0.) GO TO 110	MATE7	54
	C		IF (A(1,1).EQ.0.) GO TO 110	MATE7	55
	C		MULDW(A(1,1))	MATE7	56
	C		DO 1 1,1,M	MATE7	57
	C			MATE7	58
	C			MATE7	59
60	C		1 A(I,1)B(I,1)/MULD	MATE7	60
	C		DET=DET*MULD	MATE7	61
	C		RETURN	MATE7	62
	C		5 NM=NR-1	MATE7	63
	C		NP1=NM+1	MATE7	64
	C		INITIALIZE DETERMINANT	MATE7	65
65	C		DET=MULD	MATE7	66
	C		INITIALIZE COLUMN INDICATORS	MATE7	67
	C		DO 5 1,1,M	MATE7	68
	C		5 IND(I)=I	MATE7	69
	C		BEGIN TRIANGULARIZATION TO GET UPPER TRIANGLE	MATE7	70
	C		DO 70 NR,1,M	MATE7	71
70	C		NP1=NR+1	MATE7	72
	C		NRH	MATE7	73
	C		PCRA	MATE7	74
	C		SEARCH FOR PIVOTAL ELEMENT	MATE7	75
75	C		HIGABAS(A(NR,N))	MATE7	76
	C		DO 10 1,N,N	MATE7	77
	C		DO 10 J,N,N	MATE7	78
	C		IF (HIGABAS(A(I,J))) GO TO 10	MATE7	79
	C		HIGABAS(A(I,J))	MATE7	80
80	C		NRH	MATE7	81
	C		NRH	MATE7	82
	C		10 CONTINUE	MATE7	83
	C		TEST FOR SINGULAR MATRIX	MATE7	84
	C		IF (HIGABAS(EQ.0.)) GO TO 110	MATE7	85
85	C		UPDATE DETERMINANT	MATE7	86
	C		DET=DET*(A(NR,N))	MATE7	87
	C		INTERCHANGE ROWS	MATE7	88
	C		IF (NRH, NR, N) GO TO 30	MATE7	89
	C		DO 20 1,N,N	MATE7	90
90	C		MULDW(A(I,1))	MATE7	91
	C		A(I,1)B(A(NR,1))	MATE7	92
	C		20 A(NR,1)MULD	MATE7	93
	C		INTERCHANGE ELEMENTS OF RIGHT HAND SIDES	MATE7	94
	C		DO 25 1,N,N	MATE7	95
95	C		MULDW(A(N,1))	MATE7	96
	C		B(N,1)B(A(NR,1))	MATE7	97
	C		25 A(NR,1)MULD	MATE7	98
	C		CHANGE SIGN OF DETERMINANT DUE TO ROW INTERCHANGE	MATE7	99
	C		DET=DET*-1	MATE7	100
100	C		INTERCHANGE COLUMNS	MATE7	101
	C		30 IF (C, NR, N) GO TO 35	MATE7	102
	C		DO 40 1,N,N	MATE7	103
	C		MULDW(A(J,1))	MATE7	104
	C		A(J,1)B(A(I,1))	MATE7	105
105	C		40 A(I,1)MULD	MATE7	106
	C		INTERCHANGE COLUMN INDICATORS	MATE7	107
	C		IND(I)=IND(N)	MATE7	108
	C		IND(N)=IND(I)	MATE7	109
110	C		CHANGE SIGN OF DETERMINANT DUE TO COLUMN INTERCHANGE	MATE7	110
	C		DET=DET*-1	MATE7	111
	C		DIVIDE REDUCED EQUATION-NR BY LEADING ELEMENT	MATE7	112
	C		55 DO 60 1,N,N	MATE7	113
	C		60 A(N,1)B(A(NR,1)/A(NR,N))	MATE7	114
	C		110 65 L=1,N	MATE7	115
115	C		65 B(N,1)B(A(NR,1)/A(NR,N))	MATE7	116
	C		REDUCE MATRIX AND RIGHT HAND SIDES	MATE7	117
	C		70 L=1,N	MATE7	118
	C		65 J=1,N	MATE7	119

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MATE7	147

UNTHIG 2
UNTHOG 3
UNTHOG 4

PHNUY	2
A	2
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A	48
A	49
A	50
PHNUY	0
PHNUY	5
PHNUY	6
PHNUY	7
PHNUY	8
PHNUY	9
PHNUY	10
PHNUY	11
PHNUY	12
PHNUY	13
PHNUY	14
PHNUY	15

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100 1X72410
101 JAY240
102 KXKX0
103 0005,J81,NPG
104 NBLJENBL(J)
105 NBLJENBL(J)
106 UNUSP(J)=UNUSP(J)+ULS(NBLJ)+V2(NBLJ,J)/V2(NBLJ,J)
107 UNUSP(J)=Y2(NBLJ)+UNUSP(J)
108 0012 IEM+1,NMAX
109 IF (IEM+1).GT.100 GO TO 15
110 12 CONTINUE
111 IF (IEM+1).GT.100
112 15 CONTINUE
113 IF (IEM+1).GT.100 UNUSP(J)=UNUSP(J)+Y2(IEM+1)
114 IF (IEM+1).GT.100 NMAX=IEM
115 0014 IEM+1,IEM
116 IF (IEM+1).GT.100 GO TO 17
117 0016 IEM+1,IEM
118 17 V2(LA,J)=V2(LA+1,J)
119 V2(LA,J)=V2(LA+1,J)
120 IP2(LA,J)=IP2(LA+1,J)
121 KMP2(LA,J)=KMP2(LA+1,J)
122 18 CONTINUE
123 VLA,J)=V2(LA,J)
124 KLA,J)=V2(LA,J)
125 IP(LA,J)=IP2(LA,J)
126 KMP(LA,J)=KMP2(LA,J)
127 19 CONTINUE
128 NBL(J)=IEM
129 5 CONTINUE
130 KMIT(6,102)=S2(IEM+1,J),NPG
131 102 FORMAT(1M,33H BOUNDARY OF PARTICLE PHASE AT SE,11125,1P811,3)
132 103 FORMAT(1M,33H BOUNDARY OF PARTICLE PHASE AT SE,11125,1P811,3)
133 RETURN
134 END

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1 SUBROUTINE PPUTIN
2 COMMON /A(25,7)
3 /A(25,7)
4 /A(25,7)
5 /A(25,7)
6 /A(25,7)
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56 /A(25,7)
57 /A(25,7)
58 /A(25,7)
59 /A(25,7)
60 /A(25,7)

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1. SUPP, SUPPV, SUPPE
DIMENSION V1(10), V2(10), TPI(10), HNP(10), DENG(10)
NAMELIST/NAME1/ FFF, FFG, UMI, CL, CS, IPS, HNS, AT, HTHAN, SIG, EP, MPG,
65 1. AC, KPA, AP, A1, V1, TPI, HNP
1. V2
1. J1, J2
HEAD (5,100) FFF, FFG, CL, CS, HTHAN, AT
70 HEAD (5,100) HNS, SIG, EP, IPS, UMI
HEAD (5,100) HNP, AC
HEAD (5,100) (HNP(J), J1, MPG)
DO 25 J1, MPG
25 HNP(J) HNP(J) = 1
HNP(J) HNP(J)
75 IF (KPG, LE, 1) GO TO 245
DO 241 J2, MPG
IF (KMA, LT, HNP(J)) KMA HNP(J)
241 CONTINUE
245 CONTINUE
HEAD (5,100) (HNP(J), J1, MPG)
IF (KAC, LE, 1) GO TO 2
HEAD (5,100) (HNP(J), J1, MPG)
HEAD (5,100) (V1(J), J1, MPG)
HEAD (5,100) (TPI(J), J1, MPG)
85 HEAD (5,100) (HNP(J), J1, MPG)
HEAD (5,100) (DENG(J), J1, MPG)
DO 5 J1, MPG
KMA HNP(J)
DO 4 A1, KPA
4 K(J) HNP(J)
V(K, J) HNP(J) V1(J) / (K - 1)
IF (K, J) HNP(J)
HNP(K, J) HNP(J)
DENG(K, J) DENG(J)
95 V(K, J) HNP(J)
V2(K, J) HNP(J)
V2(K, J) HNP(J)
HNP(K, J) HNP(J)
4. CONTINUE
100 HNP(J) HNP(J)
HNP(J) HNP(J)
5. CONTINUE
DO 21 J1, MPG
KMA HNP(J)
105 DO 21 K1, KPA
105 CONTINUE
21 CONTINUE
DO 11 3
1. CONTINUE
DO 1 K1, KPA
HEAD (5,100) (K, J), J1, MPG
HEAD (5,100) (V(K, J), J1, MPG)
HEAD (5,100) (TPI(K, J), J1, MPG)
HEAD (5,100) (HNP(K, J), J1, MPG)
115 HEAD (5,100) (DENG(K, J), J1, MPG)
HEAD (5,100) (DENG(K, J), J1, MPG)
DO 6 J1, MPG
62 (K, J) HNP(J)
V(K, J) HNP(J)
120 V2(K, J) HNP(J)
HNP(K, J) HNP(J)
6. CONTINUE
1. CONTINUE
HEAD (5,100) (HNP(J), J1, MPG)
HEAD (5,100) (V(K, J), J1, MPG)
3. CONTINUE
WRITE (6, NAME1)
DO 7 J1, MPG
VULP(J) 1.333333 * V1 * (HNP(J) + 3)
130 V1P(J) HNP(J) HNS
HNP(J) HNP(J) HTHAN
DO 12 J1, MPG
IF (KMA HNP(J)) 12, 12, 11
11 KMA HNP(J) + 1
DO 10 KMA HNP(J)
V(K, J) HNP(J)
V(K, J) HNP(J)
140 V(K, J) HNP(J)
HNP(K, J) HNP(J)
V2(K, J) HNP(J)
V2(K, J) HNP(J)
HNP(K, J) HNP(J)
145 HNP(K, J) HNP(J)
HNP(K, J) HNP(J)
10. CONTINUE
12. CONTINUE
DO 22 K1, KPA
WRITE (6, 101) K
150 WRITE (6, 102) (V(K, J), J1, MPG)
WRITE (6, 103) (V1P(J), J1, MPG)
WRITE (6, 105) (V(K, J), J1, MPG)
WRITE (6, 106) (HNP(K, J), J1, MPG)
22 CONTINUE
155 100. FLOW RATE (10, 3)
101. FLOW RATE (10, 15)
102. FLOW RATE (10, 15)
103. FLOW RATE (10, 15)
104. FLOW RATE (10, 15)
105. FLOW RATE (10, 15)
106. FLOW RATE (10, 15)
107. FLOW RATE (10, 15)
108. FLOW RATE (10, 15)
109. FLOW RATE (10, 15)
110. FLOW RATE (10, 15)
111. FLOW RATE (10, 15)
112. FLOW RATE (10, 15)
113. FLOW RATE (10, 15)
114. FLOW RATE (10, 15)
115. FLOW RATE (10, 15)
116. FLOW RATE (10, 15)

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1 SUBROUTINE PAHT
COMMON A(25,7)
1 .AAT(40) .ALFA(25,25) .ALPHAM .ALPHAP .PANT 2
2 .ATOL .BETAP .BVIK .C11(25) .C(25,40) .A 3
3 .C12(25) .C2(25,40) .CABAN(25) .CBBAN(25) .CP .A 4
4 .CPS(25) .CPM .CCH(25) .CSH1(25) .CSINEM(25) .A 5
5 .CSTHM .C2TH(25,25) .DEFF(25) .DEFF(25) .DELTA .A 6
6 .D11 .DELB(40) .DELB .DELSU .DELS(40) .A 7
7 .DIM(25,25) .D12 .DELY(40) .D13 .DPOY(40) .A 8
8 .D14 .DPHDS(40) .EPCIN .A 9
9 .EPBLON .EXTNA(50) .E1EP .PHAX .CHAD .A 10
10 .H11 .H(40) .H .HJ .HMH .A 11
11 .H2PW(25) .H3PM(25) .ICNBT .ICOUNT .ICENT(25) .A 12
12 .IEMWON .IEXTRA(50) .IPLAG .IKINO .A 13
13 .IPTUC .ISHOCK .I1YPE .IPD .A 14
14 .IRIFF .K .KAY .KAYS .KAY2 .A 15
15 .KLO .KPAK .K .A 16
16 .KUP .K .LL .LPLANE .MA .A 17
17 .KABM .KDO1 .KMAX .MU .A 18
18 .KUZ .KUS .K .M2 .M2M .A 19
19 .KABUND .KDS .K1TH .KMAX .M .A 20
20 .K2(40) .KUEGA(25) .P11 .P(40) .P12 .A 21
21 .P15 .P11(40) .P11(50) .P11(50) .A 22
22 .P15 .P15 .P11(50) .P11(50) .A 23
23 .P15 .P15 .P11(50) .P11(50) .A 24
24 .P15 .P15 .P11(50) .P11(50) .A 25
25 .P15 .P15 .P11(50) .P11(50) .A 26
26 .P15 .P15 .P11(50) .P11(50) .A 27
27 .P15 .P15 .P11(50) .P11(50) .A 28
28 .P15 .P15 .P11(50) .P11(50) .A 29
29 .P15 .P15 .P11(50) .P11(50) .A 30
30 .P15 .P15 .P11(50) .P11(50) .A 31
31 .P15 .P15 .P11(50) .P11(50) .A 32
32 .P15 .P15 .P11(50) .P11(50) .A 33
33 .P15 .P15 .P11(50) .P11(50) .A 34
34 .P15 .P15 .P11(50) .P11(50) .A 35
35 .P15 .P15 .P11(50) .P11(50) .A 36
36 .P15 .P15 .P11(50) .P11(50) .A 37
37 .P15 .P15 .P11(50) .P11(50) .A 38
38 .P15 .P15 .P11(50) .P11(50) .A 39
39 .P15 .P15 .P11(50) .P11(50) .A 40
40 .P15 .P15 .P11(50) .P11(50) .A 41
41 .P15 .P15 .P11(50) .P11(50) .A 42
42 .P15 .P15 .P11(50) .P11(50) .A 43
43 .P15 .P15 .P11(50) .P11(50) .A 44
44 .P15 .P15 .P11(50) .P11(50) .A 45
45 .P15 .P15 .P11(50) .P11(50) .A 46
46 .P15 .P15 .P11(50) .P11(50) .A 47
47 .P15 .P15 .P11(50) .P11(50) .A 48
48 .P15 .P15 .P11(50) .P11(50) .A 49
49 .P15 .P15 .P11(50) .P11(50) .A 50
50 .P15 .P15 .P11(50) .P11(50) .A 51
51 .P15 .P15 .P11(50) .P11(50) .A 52
52 .P15 .P15 .P11(50) .P11(50) .A 53
53 .P15 .P15 .P11(50) .P11(50) .A 54
54 .P15 .P15 .P11(50) .P11(50) .A 55
55 .P15 .P15 .P11(50) .P11(50) .A 56
56 .P15 .P15 .P11(50) .P11(50) .A 57
57 .P15 .P15 .P11(50) .P11(50) .A 58
58 .P15 .P15 .P11(50) .P11(50) .A 59
59 .P15 .P15 .P11(50) .P11(50) .A 60
60 .P15 .P15 .P11(50) .P11(50) .A 61
61 .P15 .P15 .P11(50) .P11(50) .A 62
62 .P15 .P15 .P11(50) .P11(50) .A 63
63 .P15 .P15 .P11(50) .P11(50) .A 64
64 .P15 .P15 .P11(50) .P11(50) .A 65
65 .P15 .P15 .P11(50) .P11(50) .A 66
66 .P15 .P15 .P11(50) .P11(50) .A 67
67 .P15 .P15 .P11(50) .P11(50) .A 68
68 .P15 .P15 .P11(50) .P11(50) .A 69
69 .P15 .P15 .P11(50) .P11(50) .A 70
70 .P15 .P15 .P11(50) .P11(50) .A 71
71 .P15 .P15 .P11(50) .P11(50) .A 72
72 .P15 .P15 .P11(50) .P11(50) .A 73
73 .P15 .P15 .P11(50) .P11(50) .A 74
74 .P15 .P15 .P11(50) .P11(50) .A 75
75 .P15 .P15 .P11(50) .P11(50) .A 76
76 .P15 .P15 .P11(50) .P11(50) .A 77
77 .P15 .P15 .P11(50) .P11(50) .A 78
78 .P15 .P15 .P11(50) .P11(50) .A 79
79 .P15 .P15 .P11(50) .P11(50) .A 80
80 .P15 .P15 .P11(50) .P11(50) .A 81
81 .P15 .P15 .P11(50) .P11(50) .A 82
82 .P15 .P15 .P11(50) .P11(50) .A 83
83 .P15 .P15 .P11(50) .P11(50) .A 84
84 .P15 .P15 .P11(50) .P11(50) .A 85
85 .P15 .P15 .P11(50) .P11(50) .A 86
86 .P15 .P15 .P11(50) .P11(50) .A 87
87 .P15 .P15 .P11(50) .P11(50) .A 88
88 .P15 .P15 .P11(50) .P11(50) .A 89
89 .P15 .P15 .P11(50) .P11(50) .A 90
90 .P15 .P15 .P11(50) .P11(50) .A 91
91 .P15 .P15 .P11(50) .P11(50) .A 92
92 .P15 .P15 .P11(50) .P11(50) .A 93
93 .P15 .P15 .P11(50) .P11(50) .A 94
94 .P15 .P15 .P11(50) .P11(50) .A 95
95 .P15 .P15 .P11(50) .P11(50) .A 96
96 .P15 .P15 .P11(50) .P11(50) .A 97
97 .P15 .P15 .P11(50) .P11(50) .A 98
98 .P15 .P15 .P11(50) .P11(50) .A 99
99 .P15 .P15 .P11(50) .P11(50) .A 100

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100 C ENERGY EQUATION
      CDEL
      IF (TP(N,J).LE.TPBT) CC=CS
      IF (ICOND(N,J).EQ.0) GO TO 10
      XNUM=TP(J)+DEMP2(J) *DEL /MIN(J)/MJ
      XNUM=MAX(1,INT(XNUM)) *DEL *BIG(1P8+0,0)/N(N,J)*DEL/MJ
      DENG(N,J)=DENG(N,J)+XNUM*DEL
      IF (ABS(DENG(N,J)).GE.MC(J)) GO TO 15
      GO TO 12
105 15 CONTINUE
      ICOD(N,J)=0
      WRITE (6,102) NP(J),S,K,DENG(N,J)
110 10 CONTINUE
      XNUM=DEMP2(J)/MJ/(A,J)/CC
      XNUM=MAX(1,INT(XNUM)) *DEL *BIG(3,0)/(K(N,J) *NP(J)+XNUM*CC) * (TP(N,J)+0,0)/MJ
      TP2(N,J)=TP(N,J)+XNUM*OTON * (XNUM+XNUM)*DEL
      IF (C(TP2(N,J)-TPS)*(TP(N,J)-TPS).GE.0,0) GO TO 12
      ICOND(N,J)=1
      TP2(N,J)=TPS
      WRITE (6,103) NP(J),S,K
115 103 FORMAT (1H,1PE10,3,43H CM PARTICLE GROUP START SOLIDIFYING AT S,
120 1,OPP10,4,19H CM STREAMLINE NO.,15/)
      12 CONTINUE
      C CONTINUITY EQUATION
      XN1=(S*(N(N+1)+N(N)))/DELTA
      XN2=(S*(N2(N+1)+N2(N)))/DELTA
      XN10=XN1 *NM(N,J)+N(N,J)
      XN10=XN1 *NM(N,J)+N(N,J)
      XN2(N,J)=1.0/(XN1 *N2(N,J)+(XN10(1.0-DEL *THEN)
125 1 *ORRYD*DEL)
      XN1=XN1+DEL *
      GO TO 1
      3, V2(N,J)=0.0
      N2(N,J)=0.0
      TP2(N,J)=0.0
      XN2(N,J)=0.0
      DEP(N,J)=0.0
130 13 CONTINUE
      102 FORMAT (1H,1PE10,3,40H CM SIZE PARTICLE GROUP SOLIDIFIED AT S,
135 1,OPP10,4,19H CM STREAMLINE NO.,15/,3X,1PE10,3,3H CM /)
      XN2=
      RETURN
      END

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PART 52
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1 SUBROUTINE PUTOUT(10UT)
      COMMON A(25,7)
      ALFA(25,25),ALPHAM,ALPHAP,PUTOUT
      1 ATUL,MEIAP,MPIX,C11(25),C(25,40),A
      2 C12(25),C2(25,40),LADAN(25),CHDAN(25),CP
      3 CP5(25),CPBM,CNN(25),CSM1(25),CSTHEM(25),A
      4 INSTAN,OPTIM(25,25),DEFF(25),DEFF(25),DELTA
      5 D11,DELS(40),DELS,DELSU,DL8(40),A
      6 D12(25,25),D12,DELY(40),D13,OPDY(40),A
      7 D14,CPH108(40),PCUN,PHAS,CHAN,A
      8 EPBLUN,EXTHA(50),PSTEP,PHAS,CHAN,A
      9 PH1,N(40),PHN,MJ,HPM(25),A
      10 H2PM(25),HSPM(25),ICOUNT,ICOUNT,ICOUNT,A
      11 ITHUON,EXTHA(50),IFLAG,ININ,A
      12 IPTUC,ISMOK,ITYPE,IPD,A
      13 IUIFF,K,KAY,KAYS,KAY2,A
      14 KALD,KMAX,ALL,ALPLANE,MA,A
      15 KUP,KK,LL,MUO,MU,A
      16 KASH,MOUT,MNAX,MU,MN,A
      17 KUP,MUS,MH,MH2,MHSH,A
      18 KHIKND,KDS,INTER,MNAX,MN,A
      19 COMMON KUCASE,MEGA(25),P11,P(40),P12,P13,A
      20 P2(40),PH(50),PABAN,PHAN,A
      21 PHS,P15,PHI(40),PHI(50),PHI(50),A
      22 PH10M1,PMSTM,PHI,PHI(50),PHI,A
      23 PH12(40),PH108,PI,PR(25),PBM,A
      24 PSM1,PS1,PMSTM,PM(50),PM,A
      25 D11,U(40),UXTH1,UTIN2,A
      26 D12(50),H11,H(40),HN(50),A
      27 KMS,H13,HHAN(40),H14,H2(40),A
      28 KCUA,H15,HL(40),HESH,H14,A
      29 KNU(40),H17,HH12(40),HMS(25),HMSH,A
      30 KMSMUN,KHSHAN,HHSHAN,HU,BSH,A
      31 KSTH,KV,K19,KH(100),S,A
      32 S15,SH(50),SC(25),SH(50),A
      33 S15,SH(40),T11,T(40),T12,A
      34 T2(40),TANAR,THAN,T9(25),T14,A
      35 TAN(40),TATH1,TXTH2,TUL,TSM,A
      36 TSM1,TSTHEM,Ta(50),TAS,A
      37 T5,UI1,U(40),U12,U2(40),A
      38 C(PH1,UXTH1),UXTH2,U14,UNAN(40),USH,A
      39 USH1,USIFEM,UN(50),UN8,A
      40 X11,X(40),X12,X2(40),X15,A
      41 XH(50),XMS,XANAR(25),XNHAN(25),X15,A
      42 XH(25,40),XSH,XSTHEM,XN(50),A
      43 XMAK(40),Y11,Y(40),Y12,A
      44 Y2(40),YABAN,YUBAN,ZA(25),Z11(25),A
      45 ZJ(25,40),ZM,Z2X,Z2H,Z2H,A
      46 FX(2,40),FK(2,40),FPH1(2,40),FP(2,40),FT(2,40),A
      47 FU(2,40),INDL(2,40),INDH(2,40),FC(2,40,25),CARD,A
      48 K,K,KAYS(25),KAY2,KK,MOUT(40),PUTOUT
      49 K
      50 KAL KAY KAYS(25) KAY2 KK MOUT(40) PUTOUT

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[illegible]

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1120 SUM1(1)=SUM1(1)+POT(L)*C(1,L)
1150 IF (LL) 1125,1160,1160
1125 WRITE (6,1225) IDENT(1),C(1,L),X(1,L),X(1,L),SUM1(1),ZJ(1,L)
1225 FORMAT(1X,6HSPC1E,2X, A4,7X,2HCO,1P1E11,4,2X,2HXX,1P1E11,4,2X,
150 4 5HCO,1E11,4,3X,3HF10,
4 1P1E11,4,3X,3H230,1P1E11,4)
GO TO 1290
1160 WRITE (6,1225) IDENT(1),C(1,L),X(1,L),X(1,L),SUM1(1),ZJ(1,L)
1290 CONTINUE
155 450 IF (IP10 .EQ. 0) GO TO 457
ALPHA2(L)/PIE
ALPHA2(L)/PIE
122025
123025
124025
125025
126025
127025
160 160 00 J=1,ADS
IF (IDENT(J) .EQ. 101) 1710J
IF (IDENT(J) .EQ. 102) 1720J
IF (IDENT(J) .EQ. 103) 1730J
IF (IDENT(J) .EQ. 104) 1740J
IF (IDENT(J) .EQ. 105) 1750J
170 IF (IDENT(J) .EQ. 106) 1760J
IF (IDENT(J) .EQ. 107) 1770J
CONTINUE
175 ENA=0.733E22*P2(L)/12(L)+C2(127,L)
VET=0.215E5*SUM1(1(L))
SPHOC2(121,L)=C2,00E-23+VEE+2.40E-16)+C2(122,L)+0.7E-8/VEE+C2(123,
1L)+0.5,9/(VEE+0.2)+C2(124,L)+0.1,05/(VEE+VEE)+C2(125,L)+0.1,29E-23+VEE+
C2(126,L)+0.1,45E-23+VEE+0.9E-16)
ENUE=0.57E27*P2(L)/SGRT(T2(L))+0.00
SGM=7.14549E-04*E/E/ENUE
IF (IP10,0,1) GO TO 01
WRITE (6,1400) XL,AL,ENUE,ENUE,SGM
1400 FORMAT (2X,6H1E11,4,5X,6HXX,1P1E11,4,5X,6HXX,1P1E11,4,5X,
1 4HXX,1P1E11,4,5X,6HXX,1P1E11,4)
165 IF (IP10 .EQ. 1) OR (IP10 .EQ. 3)
1P1E11 (7,1000) XL,AL,ENUE,ENUE,SGM
1600 FORMAT (4E14,12/3E14,12)
IF (IP10 .EQ. 2) OR (IP10 .EQ. 4)
1P1E11 (8,1601) XL,AL,ENUE,ENUE,SGM
190 1601 FORMAT (7E10
01 WRITE (6,
12(L),P2(L),XS(122,L),XS(123,L),XS(121,L)
1500 FORMAT (5X,6HXX,1P1E11,4,5X,6HXX,1P1E11,4,5X,6HXX,1P1E11,4,5X,
1 2HXX,1P1E11,4,5X,6HXX,1P1E11,4,5X,6HXX,1P1E11,4,5X,6HXX,1P1E11,4)
IF (IP10 .EQ. 5) OR (IP10 .EQ. 3)
1P1E11 (7,1000) XL,AL,ENUE,ENUE,SGM
195 IF (IP10 .EQ. 6) OR (IP10 .EQ. 3)
1P1E11 (7,1000) XL,AL,ENUE,ENUE,SGM
457 IF (
500 LE
200 GO TO 250
1300 IF (NBOUND) 1320,1310,1320
1310 K=1
DELPHI=PMI(K)-PMISM
WRITE (6,1401) LL,X(12(K),X(2(K),PMI2(K),PMI1,DELPHI
1401
205 GO TO 600
1520 DELPHI=PMI(KMAX)-PMISM
WRITE (6,1401) LL,X(12(K),X(2(K),PMI2(K),PMI1,DELPHI
1401 FORMAT (1H0,5HSHOCK,2X,3ML2,14,2X,2HXX,12,2X,2HXX,1P1E11,4,2X,2HXX
210 2X,1P1E11,4,2X,2HXX,1P1E11,4,2X,2HXX,1P1E11,4,2X,2HXX,1P1E11,4)
600 RETURN
END

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1 SUBROUTINE PMOUT(RANK)
COMMON A(25,7)
1 A(10L) ,HFTAP ,BMIX ,C11(25) ,C(25,40)
2 A(12(25) ,C(25,40) ,CAHAN(25) ,CHAR(25) ,CH
3 A(13(25) ,C(25) ,CSH(25) ,C(25) ,CSTHE(25)
4 A(14(25) ,D(25,25) ,DEFF(25) ,DEFF(25) ,DELTA
5 A(15(25) ,DELS(40) ,DELS ,DELSU ,DELS(40)
6 A(16(25) ,D12 ,DELY(40) ,D13 ,DDPY(40)
7 A(17(25) ,DPHDS(40) ,EPCUN ,FMAX ,GHAD
8 A(18(25) ,FSTKA(40) ,FSTEP ,FMAX ,GHAD
9 A(19(25) ,FSTKA(40) ,FSTEP ,FMAX ,GHAD
10 A(20(25) ,FSTKA(40) ,FSTEP ,FMAX ,GHAD
11 A(21(25) ,FSTKA(40) ,FSTEP ,FMAX ,GHAD
12 A(22(25) ,FSTKA(40) ,FSTEP ,FMAX ,GHAD
13 A(23(25) ,FSTKA(40) ,FSTEP ,FMAX ,GHAD
14 A(24(25) ,FSTKA(40) ,FSTEP ,FMAX ,GHAD
15 A(25(25) ,FSTKA(40) ,FSTEP ,FMAX ,GHAD
16 A(26(25) ,FSTKA(40) ,FSTEP ,FMAX ,GHAD
17 A(27(25) ,FSTKA(40) ,FSTEP ,FMAX ,GHAD
18 A(28(25) ,FSTKA(40) ,FSTEP ,FMAX ,GHAD
19 A(29(25) ,FSTKA(40) ,FSTEP ,FMAX ,GHAD
20 A(30(25) ,FSTKA(40) ,FSTEP ,FMAX ,GHAD
21 A(31(25) ,FSTKA(40) ,FSTEP ,FMAX ,GHAD
22 A(32(25) ,FSTKA(40) ,FSTEP ,FMAX ,GHAD

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D-27

	GO TO 402	STEP	50
	401 CALL CHEM(KINE)	STEP	55
	402 CONTINUE	STEP	56
105	DO 403 I=1,NDS	STEP	57
	IF (C2(I,K)) 404,403,403	STEP	58
	403 CONTINUE	STEP	59
	GO TO 405	STEP	60
110	404 DELS=DELS+.5	STEP	61
	ICOUNT=ICOUNT+1	STEP	62
	KALAK=2	STEP	63
	GO TO 10000	STEP	64
	405 ZH=0.0	STEP	65
	DO 500 I=1,NDS	STEP	66
	500 ZH=ZH+C2(I,K)/K(I)	STEP	67
115	ZH=Z/2M	STEP	68
	ZH=Z/2M	STEP	69
	DO 600 J=1,N	STEP	70
	Z(J)=0.0	STEP	71
	ZAU(J)=0	STEP	72
120	DO 600 I=1,NDS	STEP	73
	ZAU(J)=ZAU(J)+A(I,J)*C(I,K)	STEP	74
	600 Z(J)=Z(J)+A(I,J)*C2(I,K)	STEP	75
	ITER=0	STEP	76
	700 US=US*V(K)	STEP	77
125	CP=0.0	STEP	78
	CPUL=0.0	STEP	79
	DO 702 J=1,N	STEP	80
	CPUL=CPUL+O+FLOAT(J-2)*ZAU(J)*T(K)+A(J-3)	STEP	81
	702 CP=CP+Z(J)+FLAT(J-2)+T(K)+A(J-3)	STEP	82
130	B1=(RHS+G+MOUT(K)+A(K))/R1AP+RAN)+P(K)	STEP	83
	R1=1-0.5*(K)+SLIP/R1AP	STEP	84
	R2=R1/(K)/(R1AP+RAN)	STEP	85
	706 R3=(RHS+G+MOUT(K)+A(K))/R1AP+RAN	STEP	86
	IF (IRUGSH,FL,0) GO TO 703	STEP	87
135	WRITE (6,2001) A(LD,M1,M2,M3	STEP	88
	2001 FORMAT (1H0.5M1,LD,1PIE11,4.5M B1,1PIE11,4.5M B2,1PIE11,4.5M	STEP	89
	A M3,1PIE11,4.5M B3)	STEP	90
	703 IF (MA(K)-1.002) 1400,1400,704	STEP	91
	704 IF (ICOUNT) 705,710,705	STEP	92
140	705 IF (MA(K)=5.0) 800,710,710	STEP	93
	C	STEP	94
	CONSTANT GA=4	STEP	95
	710 RSEN=JACPARS+H	STEP	96
	RSEN=0.0+JACPARS+H2=1.0	STEP	97
145	RSEN(K)=RSEN+2.0+JACPARS+H2=1.0	STEP	98
	RSEN=0.0+JACPARS+H2=1.0	STEP	99
	IF (IRUGSH,FL,0) WRITE (6,2001) 84,B5,H6,DISCH	STEP	100
	1 84,B5,H6,DISCH	STEP	101
	IF (RSEN) 730,750,750	STEP	102
150	730 IF (ICOUNT=9) 740,740,9400	STEP	103
	740 ICOUNT=ICOUNT+1	STEP	104
	DELS=DELS/2.0	STEP	105
	KALAK=2	STEP	106
	GO TO 10000	STEP	107
155	750 US=(RHS+G+MOUT(K)+A(K))/R1AP+RAN)+P(K)	STEP	108
	IF (IRUGSH,FL,0) WRITE (6,2001) US,T(K),P(K),U(K)	STEP	109
	GO TO 1200	STEP	110
	VARIABLE HEAT CAPACITY	STEP	111
	700 ITER=ITER+1	STEP	112
160	RSEN=0.0+JACPARS+H2=1.0	STEP	113
	RSEN=0.0+JACPARS+H2=1.0	STEP	114
	DO 900 I=1,N	STEP	115
	DO 900 I=1,N	STEP	116
165	DO 900 I=1,N	STEP	117
	DO 900 I=1,N	STEP	118
	DO 900 I=1,N	STEP	119
	DO 900 I=1,N	STEP	120
	DO 900 I=1,N	STEP	121
	DO 900 I=1,N	STEP	122
170	IF (RHS+G+MOUT(K)+A(K))/R1AP+RAN)+P(K)	STEP	123
	1000 IF (ITER=4) 1100,1010,1010	STEP	124
175	1010 IF (ICOUNT=9) 1020,1020,9500	STEP	125
	1020 ICOUNT=ICOUNT+1	STEP	126
	DELS=DELS/2.0	STEP	127
	GO TO 10000	STEP	128
	1100 US=US*V(K)	STEP	129
	GO TO 800	STEP	130
180	1200 U2(K)=0	STEP	131
	U2(K)=0	STEP	132
	U2(K)=0	STEP	133
	U2(K)=0	STEP	134
	U2(K)=0	STEP	135
	U2(K)=0	STEP	136
	U2(K)=0	STEP	137
	U2(K)=0	STEP	138
185	1300 FORMAT (2H) PRESSURE BECAME NEGATIVE)	STEP	139
	1300 (ALL MOUT(K)+A(K))/R1AP+RAN)+P(K)	STEP	140
	CALL EXIT	STEP	141
	1300 IF (RSEN) 1305,1305,1307	STEP	142
190	1305 IF (ICOUNT,LT,9) GO TO 1020	STEP	143
	WRITE (6,1306)	STEP	144
	1306 FORMAT (2H) TEMPERATURE BECAME NEGATIVE)	STEP	145
	GO TO 1305	STEP	146
	1307 FORMAT (2H) AND REYNOLDS NUMBERS	STEP	147
195	1307 FORMAT (2H) AND REYNOLDS NUMBERS	STEP	148
	IF (RSEN) 1309	STEP	149
	1309 FORMAT (2H) AND REYNOLDS NUMBERS	STEP	150
	1309 FORMAT (2H) AND REYNOLDS NUMBERS	STEP	151
	1309 FORMAT (2H) AND REYNOLDS NUMBERS	STEP	152

[illegible]

15	1	IEKRUH	IEEXTRA(50)	IFLAG	IRIND	A	13
	2	IPYUC	ISHOCK	ITYPE	IPD	A	14
	3	IDIFF	IA	KAY	KAYS	A	15
	4	ALU	KMAX		KAY2	A	16
	5	KUP	KA	LL	LPLANE	A	17
	6	ASH	KOUT	KMAX	MU	A	18
20	7	U2	MUS	MA	MW?	A	19
	8	ACGUL	KUS	NITER	KMAX	A	20
	9	ACASE	UNEGA(25)	P11	P(40)	A	21
	10	P2(40)		PE(50)	PAHAN	A	22
	11	PS	P15	PHI(40)	PHI(50)	A	23
25	12	PHISH1	PHSIRH		PHI(50)	A	24
	13	PHI2(40)	PHIS	P1	PH(25)	A	25
	14	PSH1	PSI	PSTHEM	PH(50)	A	26
	15	Q11	Q(40)	QXTR1	QXTR2	A	27
30	16	LA(50)	M11	M(40)	MH(50)	A	28
	17	MHS	M13	MHAP(40)	M14	A	29
	18	MCUA	M15	ME(40)	M16	A	30
	19	MHI(40)	M17	MH2(40)	MHS(25)	A	31
	20	MHSOM	MHAHAN	MHBAR	MU	A	32
	21	MSTHEM	MV	M19	M(100)	A	33
35	22		Sh(50)	SL(25)	SA(50)	A	34
	23	S13	SX(40)	S11	S(40)	A	35
	24	T2(40)	TAMH	TBBH	T(25)	A	36
	25	TAK(40)	TATK1	TXINZ	TUL	A	37
	26	TSM1	TSTHEM	TW(50)	TAS	A	38
40	27	TS	T11	TU(40)	TU2(40)	A	39
	28	UXTR1	UXTR2	UTA	UHAN(40)	A	40
	29	USH1	USTHEM	UH(50)	UAS	A	41
	30	X11	X(40)	X12	X2(40)	A	42
45	31	XU(50)	XK2	XAHAN(25)	XBBAN(25)	A	43
	32	X5(25,40)	XSH	XSTHEM	XN(50)	A	44
	33	ZMAR(40)		Y11	Y(40)	A	45
	34	Y2(40)	YAHAN	YBBAN	Y2(25)	A	46
	35	ZJ(25,40)	ZHA	S2X	X2SH	A	47
	36	FX(2,40)	FK(2,40)	FPHI(2,40)	FP(2,40)	A	48
50	37	FU(2,40)	INDL(2,40)	INDR(2,40)	FC(2,40,25)	A	49
	38	KA	K			A	50
	39	KAY	KAYS(25)	KAY2	KW	TRANS	51
	40	KX(40)	KU	MUS(25)	MU(25)	TRANS	52
	41	KX(25)	KW	MAH	MASH	TRANS	53
55	42	KAP				TRANS	54
	43	CUMMUN/MHELE/MHEE(40)				TRANS	55
	44	CUMMUN/TURBUL/ILE,THR,EDDYK,ITURB,DELMIX				TRANS	56
60	45	CUMMUN/INPUX/ITE,XG,YO,FHAC,KFL				TRANS	57
	46	CUMMUN/KYZ/XYZ,XYZ				TRANS	58
	47	CUMMUN/PEHGF/PEHGF				TRANS	59
	48	XYZZ1H				TRANS	60
	49	XYZZ2H				TRANS	61
	50	KAYZRAY				TRANS	62
	51	KYZRAY				TRANS	63
65	52	U2IM(1,2)=DIM(1,2)				TRANS	64
	53	IF (THUGSH,NE,0)				TRANS	65
	54	1 KXILE(6,700) KKK,LL,POOT(KKK),UKKK,KHU(KKK),KHU(KKK+1),IFLAG				TRANS	66
	55	IF (KKA=2) 100,100,450				TRANS	67
70	56	IF (L1,0) 10,400				TRANS	68
	57	IF (L1UKK=2) 301,302,400				TRANS	69
	58	1 KXAXKMU(2)U(2)				TRANS	70
	59	MUPINEMH(2)U(2)				TRANS	71
	60	GO 300 LE2,KMAX				TRANS	72
	61	TEST = KXILE(1)U(1)				TRANS	73
75	62	IF (TEST,01, PUMAX) GO TO 200				TRANS	74
	63	IF (TEST,01, PUMIN) GO TO 300				TRANS	75
	64	PUMIN = TEST				TRANS	76
	65	GO TO 300				TRANS	77
	66	200 PUMAX=TEST				TRANS	78
80	67	300 CUMIN=TEST				TRANS	79
	68	DELTA=TEST(KKK)				TRANS	80
	69	KUR (DELTA/FOOTK)+(PUMAX-PUMIN) * 0.047896				TRANS	81
	70	DELTA(1,1) PUMAX,PUMIN,DELTA,KU,KHU(KMAX),U(KMAX),KHAHAN				TRANS	82
	71	GO TO 450				TRANS	83
85	72	301 KXAXKAX=1				TRANS	84
	73	----- DEFORMING SIMPLANE -----				TRANS	85
	74	1 KXAXKAX(2)				TRANS	86
	75	KXSL2				TRANS	87
	76	GO KXU IF K2,KKM				TRANS	88
90	77	IF (ABS(TAX(I+1)),LL,ABS(TAXM)) GO TO 400				TRANS	89
	78	KXSL2=0				TRANS	90
	79	1 KXAXKAX(KXSL2)				TRANS	91
	80	KXSL2=0				TRANS	92
	81	KXSL2=0				TRANS	93
95	82	----- COMPUTE DEFORM AND MU -----				TRANS	94
	83	PERFORMPERF=ABS(TAXM)				TRANS	95
	84	1 KXAXKAX				TRANS	96
	85	IF K1,IF K2,KXSL				TRANS	97
	86	IF (ABS(TAX(I+1)),LL,PERF) KXAXIF				TRANS	98
100	87	LL=1				TRANS	99
	88	K1=KXAXKAX				TRANS	100
	89	DELTA=KXAXKAX(KKAX)+ABS(U(1)-U(KMAX))/EDDYK				TRANS	101
	90	IF (DELTA,LL,0,0) KXSL=0				TRANS	102
105	91	K1=KXAXKAX				TRANS	103
	92	1 KXAXKAX				TRANS	104
	93	IF K1,IF K2,KXSL				TRANS	105
	94	IF (ABS(TAX(I+1)),LL,PERF) KXAXIF				TRANS	106
110	95	LL=1				TRANS	107
	96	K1=KXAXKAX				TRANS	108
	97	DELTA=KXAXKAX(KKAX)+ABS(U(1)-U(KMAX))/EDDYK				TRANS	109
	98	IF (DELTA,LL,0,0) KXSL=0				TRANS	110
	99	K1=KXAXKAX				TRANS	111
	100	1 KXAXKAX				TRANS	112
	101	IF K1,IF K2,KXSL				TRANS	113
	102	IF (ABS(TAX(I+1)),LL,PERF) KXAXIF				TRANS	114
	103	LL=1				TRANS	115
	104	K1=KXAXKAX				TRANS	116
	105	DELTA=KXAXKAX(KKAX)+ABS(U(1)-U(KMAX))/EDDYK				TRANS	117
	106	IF (DELTA,LL,0,0) KXSL=0				TRANS	118
	107	K1=KXAXKAX				TRANS	119
	108	1 KXAXKAX				TRANS	120
	109	IF K1,IF K2,KXSL				TRANS	121
	110	IF (ABS(TAX(I+1)),LL,PERF) KXAXIF				TRANS	122
	111	LL=1				TRANS	123
	112	K1=KXAXKAX				TRANS	124
	113	DELTA=KXAXKAX(KKAX)+ABS(U(1)-U(KMAX))/EDDYK				TRANS	125
	114	IF (DELTA,LL,0,0) KXSL=0				TRANS	126
	115	K1=KXAXKAX				TRANS	127
	116	1 KXAXKAX				TRANS	128
	117	IF K1,IF K2,KXSL				TRANS	129
	118	IF (ABS(TAX(I+1)),LL,PERF) KXAXIF				TRANS	130
	119	LL=1				TRANS	131
	120	K1=KXAXKAX				TRANS	132
	121	DELTA=KXAXKAX(KKAX)+ABS(U(1)-U(KMAX))/EDDYK				TRANS	133
	122	IF (DELTA,LL,0,0) KXSL=0				TRANS	134
	123	K1=KXAXKAX				TRANS	135
	124	1 KXAXKAX				TRANS	136
	125	IF K1,IF K2,KXSL				TRANS	137
	126	IF (ABS(TAX(I+1)),LL,PERF) KXAXIF				TRANS	138
	127	LL=1				TRANS	139
	128	K1=KXAXKAX				TRANS	140
	129	DELTA=KXAXKAX(KKAX)+ABS(U(1)-U(KMAX))/EDDYK				TRANS	141
	130	IF (DELTA,LL,0,0) KXSL=0				TRANS	142
	131	K1=KXAXKAX				TRANS	143
	132	1 KXAXKAX				TRANS	144
	133	IF K1,IF K2,KXSL				TRANS	145
	134	IF (ABS(TAX(I+1)),LL,PERF) KXAXIF				TRANS	146
	135	LL=1				TRANS	147
	136	K1=KXAXKAX				TRANS	148
	137	DELTA=KXAXKAX(KKAX)+ABS(U(1)-U(KMAX))/EDDYK				TRANS	149
	138	IF (DELTA,LL,0,0) KXSL=0				TRANS	150
	139	K1=KXAXKAX				TRANS	151
	140	1 KXAXKAX				TRANS	152
	141	IF K1,IF K2,KXSL				TRANS	153
	142	IF (ABS(TAX(I+1)),LL,PERF) KXAXIF				TRANS	154
	143	LL=1				TRANS	155
	144	K1=KXAXKAX				TRANS	156
	145	DELTA=KXAXKAX(KKAX)+ABS(U(1)-U(KMAX))/EDDYK				TRANS	157
	146	IF (DELTA,LL,0,0) KXSL=0				TRANS	158
	147	K1=KXAXKAX				TRANS	159
	148	1 KXAXKAX				TRANS	160
	149	IF K1,IF K2,KXSL				TRANS	161
	150	IF (ABS(TAX(I+1)),LL,PERF) KXAXIF				TRANS	162
	151	LL=1				TRANS	163
	152	K1=KXAXKAX				TRANS	164
	153	DELTA=KXAXKAX(KKAX)+ABS(U(1)-U(KMAX))/EDDYK				TRANS	165
	154	IF (DELTA,LL,0,0) KXSL=0				TRANS	166
	155	K1=KXAXKAX				TRANS	167
	156	1 KXAXKAX				TRANS	168
	157	IF K1,IF K2,KXSL				TRANS	169
	158	IF (ABS(TAX(I+1)),LL,PERF) KXAXIF				TRANS	170
	159	LL=1				TRANS	171
	160	K1=KXAXKAX				TRANS	172
	161	DELTA=KXAXKAX(KKAX)+ABS(U(1)-U(KMAX))/EDDYK				TRANS	173
	162	IF (DELTA,LL,0,0) KXSL=0				TRANS	174
	163	K1=KXAXKAX				TRANS	175
	164	1 KXAXKAX				TRANS	176
	165	IF K1,IF K2,KXSL				TRANS	177
	166	IF (ABS(TAX(I+1)),LL,PERF) KXAXIF				TRANS	178
	167	LL=1				TRANS	179
	168	K1=KXAXKAX				TRANS	180
	169	DELTA=KXAXKAX(KKAX)+ABS(U(1)-U(KMAX))/EDDYK				TRANS	181
	170	IF (DELTA,LL,0,0) KXSL=0				TRANS	182
	171	K1=KXAXKAX				TRANS	183
	172	1 KXAXKAX				TRANS	184
	173	IF K1,IF K2,KXSL				TRANS	185
	174	IF (ABS(TAX(I+1)),LL,PERF) KXAXIF				TRANS	186
	175	LL=1				TRANS	187
	176	K1=KXAXKAX				TRANS	188
	177	DELTA=KXAXKAX(KKAX)+ABS(U(1)-U(KMAX))/EDDYK				TRANS	189
	178	IF (DELTA,LL,0,0) KXSL=0				TRANS	190
	179	K1=KXAXKAX				TRANS	191
	180	1 KXAXKAX				TRANS	192
	181	IF K1,IF K2,KXSL				TRANS	193
	182	IF (ABS(TAX(I+1)),LL,PERF) KXAXIF				TRANS	194
	183	LL=1				TRANS	195
	184	K1=KXAXKAX				TRANS	196
	185	DELTA=KXAXKAX(KKAX)+ABS(U(1)-U(KMAX))/EDDYK				TRANS	197
	186	IF (DELTA,LL,0,0) KXSL=0				TRANS	198
	187	K1=KXAXKAX				TRANS	199
	188	1 KXAXKAX				TRANS	200
	189	IF K1,IF K2,KXSL				TRANS	201
	190	IF (ABS(TAX(I+1)),LL,PERF) KXAXIF				TRANS	202
	191	LL=1				TRANS	203
	192	K1=KXAXKAX				TRANS	204
	193	DELTA=KXAXKAX(KKAX)+ABS(U(1)-U(KMAX))/EDDYK				TRANS	205
	194	IF (DELTA,LL,0,0) KXSL=0				TRANS	206
	195	K1=KXAXKAX				TRANS	207
	196	1 KXAXKAX				TRANS	208
	197	IF K1,IF K2,KXSL				TRANS	209
	198	IF (ABS(TAX(I+1)),LL,PERF) KXAXIF				TRANS	210
	199	LL=1				TRANS	211
	200	K1=KXAXKAX				TRANS	212
	201	DELTA=KXAXKAX(KKAX)+ABS(U(1)-U(KMAX))/EDDYK				TRANS	213
	202	IF (DELTA,LL,0,0) KXSL=0				TRANS	214
	203	K1=KXAXKAX				TRANS	215
	204	1 KXAXKAX				TRANS	216
	205	IF K1,IF K2,KXSL				TRANS	217
	206	IF (ABS(TAX(I+1)),LL,PERF) KXAXIF				TRANS	218
	207	LL=1				TRANS	219
	208	K1=KXAXKAX				TRANS	220
	209	DELTA=KXAXKAX(KKAX)+ABS(U(1)-U(KMAX))/EDDYK				TRANS	221


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      DO 500 J=1,NH
      500 CPS(I)=CPS(I)+FLOAT(J-2)*A(I,J)*TI*(J-3)
      TRANSF 66
      TRANSF 67

115      IF (LL .LT. 0) GO TO 550
      CPS(I)=CPS(I)+CAHAR(I)
      GO TO 600
      550 CPS(I)=CPS(I)+C(I,NH)
      600 CONTINUE
      120      KAYR(M)=U/TPK
      IF (IHUGSM.NE.0)
      1KH(I(6,702) CP,NAT,KHABAR
      IF (LL .LT. 0) GO TO 650
      LHM(I,2)=KAY+ILE/(RHABAR*CP)
      125      GO TO 10000
      650 UH(I,2)=KAY+ILE/(HML*(KAY)*CP)
      11 FHM(I(1M ,CHPUPAAA,1PE9,2,2X,CHPUMIN,19,2,2X,7HDELMIX,19,2,2X
      1 ,3HPU,19,2,2X,CHHMO,19,2,2X,2HPU,19,2,2X,7HMHABAR,19,2)
      700 FOM(I(1M ,CHHKKK,13,2X,3HLL,13,2X,5HDOT,1PE9,2,2X,2HPU,19,2,2X
      1 2X,7HMHOD(h),19,2,2X,CHRHU(h+1),19,2,2X,6HIFLAG,12)
      702 FHM(I(1M ,3HCP,1PE9,2,2X,CHKAY,19,2,2X,7HMHABAR,19,2)
      10000 KETUP,
      END
      TRANSF 68
      TRANSF 69
      TRANSF 70
      TRANSF 71
      TRANSF 72
      TRANSF 73
      TRANSF 74
      TRANSF 75
      TRANSF 76
      TRANSF 77
      TRANSF 78
      TRANSF 79
      TRANSF 80
      TRANSF 81
      TRANSF 82
      TRANSF 83
      TRANSF 84
      TRANSF 85
      TRANSF 86

1      SUBROUTINE VISCO
      THIS SUBROUTINE CALCULATES THE FLUX CONTRIBUTIONS TO THE
      C CONSERVATION EQUATIONS
      C COMMON A(25,7) ,AA(40) ,ALFA(25,25),ALPHAM ,ALPHAP ,ALPHA
      5      1 ,ATOL ,REIAP ,HMIX ,C11(25) ,C(25,40) ,A
      2 ,C12(25) ,C2(25,40) ,CAHAR(25) ,CBBAR(25) ,CP ,A
      3 ,CPS(25) ,CPSH ,CSH(25) ,CSH1(25) ,CSTHEM(25) ,A
      4 ,XSTHM ,U2TH(25,25),UEFF(25) ,DEFF(25) ,DELTA ,A
      5 ,U11 ,DELS(40) ,DELS ,DELS1 ,DELS1(40) ,A
      10      6 ,LHM(25,25) ,H12 ,DELT(40) ,U13 ,UPUT(40) ,A
      7 ,D14 ,LHMUS(40) ,LPCUM ,A
      8 ,LMSLH ,FALKA(50) ,FSTEP ,FMAX ,GHAD ,A
      9 ,H11 ,H(40) ,HM ,HMJ ,HMH(25) ,A
      15      10 ,H2PM(25) ,H3PM(25) ,ICONS1 ,ICUUM1 ,IDEN1(25) ,A
      1 ,LEKRIW ,IFXTHA(50) ,IFLAG ,INJAD ,A
      2 ,IPLOC ,ISHOCK ,ITYPE ,IPD ,A
      3 ,IDIFF ,K ,KAY ,KAYS ,KAY2 ,A
      4 ,RLU ,KMAX ,A
      5 ,RUP ,KA ,LL ,LMLAL ,MA ,A
      20      6 ,MASH ,MLU1 ,MMAX ,MH0 ,MU ,A
      7 ,MU2 ,MLS ,MA ,M2 ,MASH ,A
      8 ,MHUUM1 ,MDS ,NITEM ,NMAX ,NA ,A
      9 ,CUMIN ,UCASE ,UMEGA(25) ,P11 ,P(40) ,P12 ,A
      25      1 ,P2(40) ,PB(50) ,PABAR ,PBHAR ,A
      2 ,PMS ,F15 ,PHI(40) ,PHI(50) ,PHI ,A
      3 ,FMSH ,FMSHNM ,P1 ,PM(50) ,P1A ,A
      4 ,PM12(40) ,PHIS ,P1 ,PM(25) ,PM ,A
      5 ,PMS1 ,PMS1 ,PSTHM ,U2TH2 ,PM(50) ,A
      30      6 ,H11 ,Q(40) ,UXTH1 ,A
      7 ,H(50) ,H11 ,H(40) ,UXTH2 ,PB(50) ,A
      8 ,HNS ,H13 ,HBAK(40) ,H14 ,H2(40) ,A
      9 ,HCUA ,H15 ,H(40) ,HLSH ,H16 ,A
      40      10 ,HMH(40) ,H17 ,HMH(25) ,HNS(25) ,HMSH ,A
      1 ,HMSH ,KHABAR ,HMSH ,RU ,RSH ,A
      35      2 ,HSTHEM ,MV ,R19 ,Rn(100) ,S ,A
      3 ,SH(50) ,SC(25) ,SH(50) ,A
      4 ,S15 ,S(40) ,T11 ,T(40) ,T12 ,A
      5 ,T2(40) ,TAPAR ,TBAH ,T(25) ,T14 ,A
      6 ,T(40) ,T1K1 ,T1K2 ,TUL ,TSH ,A
      40      7 ,TSM1 ,TSTHEM ,T(50) ,TMS ,A
      8 ,T5 ,U11 ,U(40) ,U12 ,U2(40) ,A
      9 ,U2TH1 ,U2TH2 ,U14 ,UHAR(40) ,USH ,A
      1 ,USH1 ,USTHEM ,UH(50) ,UAS ,A
      45      2 ,U11 ,A(40) ,A12 ,A2(40) ,A
      3 ,XN(50) ,XNS ,XABAR(25) ,XMBAR(25) ,X15 ,A
      4 ,X2(25,40) ,XSM ,XSTHEM ,XNS(25) ,XN(50) ,A
      5 ,XMAK(40) ,Y11 ,Y(40) ,Y12 ,A
      6 ,Y2(40) ,YBAK ,YHBAK ,ZA(25) ,Z11(25) ,A
      50      7 ,Z(25,40) ,ZHA ,SZX ,H2SH ,X2SH ,A
      8 ,F1(2,40) ,F(2,40) ,FPM1(2,40) ,FP(2,40) ,F1(2,40) ,A
      9 ,F(2,40) ,FADL(2,40) ,FUR(2,40) ,FC(2,40,25),FAND1 ,A
      6 ,F ,A
      55      7HAE ,NAT ,NAYS(25) ,NAY2 ,NN ,MDUT(40) ,VISCO
      8 ,NA(40) ,NU ,MU0(25) ,MU2 ,VISCO
      9 ,NA(25) ,NU2 ,MASH ,MASH1 ,MASH ,VISCO
      10 ,NA ,A
      11 ,A ,A
      12 ,A ,A
      13 ,A ,A
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	400	CUMMHBAK(K-1))=DELTA	VISCO	25
	60	IL 1200	VISCO	26
75	1000	IF (LELTA) 400,1100,900	VISCO	27
	1100	LELTA=1.0	VISCO	28
	C	HEAT TRANSFER	VISCO	29
	1200	HNSPUM=(HBAK(K))=DELTA*(A(K)*DLS(K)-DUMH*(A(K-1)*DLS(K-1)))	VISCO	30
		AAPI=2.0=DELTA	VISCO	31
80	C	HEAT TRANSFER	VISCO	32
	IF (K-2)	1300,1300,1600	VISCO	33
	1300	DSSPIN=HBAK(K)=DELTA*(A(K)*UBAK(K)+UBAK(K+1))/2.0=DLS(K)-DUMH	VISCO	34
		AA(K)=S+DLS(K-1)	VISCO	35
	60	IL 1500	VISCO	36
	1400	DSSPIN=HBAK(K)=DELTA*(A(K)*UBAK(K)+UBAK(K+1))/2.0=DLS(K)-DUMH	VISCO	37
		AA(K)=1+(UBAK(K)+UBAK(K+1))/2.0=DLS(K-1)	VISCO	38
85	C	HEAT TRANSFER BY DIFFUSION	VISCO	39
	1500	DIFFLX=0.0	VISCO	40
	IL 1700	I=1,NHS	VISCO	41
		HSP(I)=H2PM(I)	VISCO	42
90		H2PM(I)=HPI(I)	VISCO	43
		HPI(I)=0.0	VISCO	44
	1550	IF (K+1-NHSA) 1550,1650,1650	VISCO	45
	1550	IF (IFLAG) 1570,1580,1570	VISCO	46
95	1580	HPI(I)=CPS(I)+I(I+1)	VISCO	47
	60	IL 1850	VISCO	48
	1570	HPI(I)=LPS(I)+.5*(I(K+1)+I(K+2))	VISCO	49
	1650	DIFFLX=DIFFLX+.5*(HBAK(K)+DELTA*(ZJ(I,K)*DLS(K)+(HPI(I)+H2PM(I)))	VISCO	50
		A=.5*(DUMH*(ZJ(I,K-1)*DLS(K-1)+(H2PM(I)+H3PM(I)))	VISCO	51
	C	SPECIES TRANSFER	VISCO	52
100	1700	HNS(I)=HBAK(K)=DELTA*(ZJ(I,K)*DLS(K)-DUMH*(ZJ(I,K-1)*DLS(K-1))+PI	VISCO	53
		AA(2.0)=DELTA	VISCO	54
	C	HEAT TRANSFER	VISCO	55
		HNSPUM=(DIFFLX+DSSPIN/HJ+HBAK(K)+DELTA*(A(K)*DLS(K)-DUMH*(A(K-1)*DLS	VISCO	56
		K-1))+API=2.0=DELTA	VISCO	57
105		HDEL(K)=ABS(HNSPUM)+HDEL(K)	VISCO	58
	IF (IHUGSM,10.0)	GO TO 10000	VISCO	59
		WRITE (6,2001)	VISCO	60
	2001	FUNWAT (100,7HNSPUM)	VISCO	61
110		DO 2100 I=1,NHS	VISCO	62
		WRITE (6,2002) HNS(I)	VISCO	63
	2100	CONTINUE	VISCO	64
	2002	FUNWAT (3X,1PIE11.4)	VISCO	65
		WRITE (6,2003) HNS(0),DIFFLX,DSSPIN,HNSPUM	VISCO	66
	2003	FUNWAT (100,7HNSPUM,1PIE11.4,HNS DIFFLX,1PIE11.4,9H DSSPIN,1PI	VISCO	67
115		PIE11.4,HNS HNSPUM,1PIE11.4)	VISCO	68
	10000	RETURN	VISCO	69
	END		VISCO	70
1		SUBROUTINE SLUP(X,N)	SLUP	2
		THIS PROGRAM FINDS THE SOLUTIONS TO A SET OF N SIMULTANEOUS LINEAR	SLUP	3
		EQUATIONS BY USING THE GAUSS-JORDAN REDUCTION ALGORITHM WITH THE	SLUP	4
		DIAGONAL PIVOT STRATEGY	SLUP	5
5		DIAGONAL PIVOT STRATEGY	SLUP	6
	DO 9 K=1,N		SLUP	7
	IF (ABS(A(K,K)) .GT. 1.E-10) GO TO 5		SLUP	8
	WRITE (6,101) K,A(K,K)		SLUP	9
	GO TO 101,K		SLUP	10
10		WRITE (6,100)(A(I,K),J=1,N),J=1,N)	SLUP	11
	10	CONTINUE	SLUP	12
	100	FUNWAT(1X, 10F12.3)	SLUP	13
	101	FORWAT(22H INVERSE = S=ALL PIVOT ,15, F12.3)	SLUP	14
		STOP	SLUP	15
15		5 KPI=K+1	SLUP	16
	DO 8 J=KPI, N		SLUP	17
	A(I,J)=A(I,J)/A(I,K)		SLUP	18
	X(K)=X(K)/A(K,K)		SLUP	19
	A(K,K)=1.0		SLUP	20
20	DO 9 I=1,K		SLUP	21
	IF (1.E-6 .LT. ABS(A(I,K)+A(I,K)*X(K))) GO TO 4		SLUP	22
	DO 8 J=KPI,N		SLUP	23
	A(I,J)=A(I,J)-A(I,K)*A(K,J)		SLUP	24
	X(I)=X(I)-A(I,K)*X(K)		SLUP	25
25		A(I,K)=0.	SLUP	26
	4	CONTINUE	SLUP	27
	99	RETURN	SLUP	28
	END		SLUP	29
1		SUBROUTINE MURKLE	MURKLE	2
	COMMON A(25,7)	ALFA(25,25),ALPHAM	MURKLE	3
	1	ALGOL	HEIAP	4
	2	CL(25)	CL(25,40)	5
5	3	CPS(25)	CONAR(25)	6
	4	CSM(25)	CSM(25)	7
	5	CSM(25)	CSM(25)	8
	6	CSM(25)	CSM(25)	9
	7	CSM(25)	CSM(25)	10
10	8	CSM(25)	CSM(25)	11
	9	CSM(25)	CSM(25)	12
	10	CSM(25)	CSM(25)	13
	11	CSM(25)	CSM(25)	14
	12	CSM(25)	CSM(25)	15
15	13	CSM(25)	CSM(25)	16
	14	CSM(25)	CSM(25)	17
	15	CSM(25)	CSM(25)	18
	16	CSM(25)	CSM(25)	19
	17	CSM(25)	CSM(25)	20
	18	CSM(25)	CSM(25)	21
	19	CSM(25)	CSM(25)	22
	20	CSM(25)	CSM(25)	23
	21	CSM(25)	CSM(25)	24
	22	CSM(25)	CSM(25)	25
	23	CSM(25)	CSM(25)	26
	24	CSM(25)	CSM(25)	27
	25	CSM(25)	CSM(25)	28
	26	CSM(25)	CSM(25)	29
	27	CSM(25)	CSM(25)	30
	28	CSM(25)	CSM(25)	31
	29	CSM(25)	CSM(25)	32
	30	CSM(25)	CSM(25)	33
	31	CSM(25)	CSM(25)	34
	32	CSM(25)	CSM(25)	35
	33	CSM(25)	CSM(25)	36
	34	CSM(25)	CSM(25)	37
	35	CSM(25)	CSM(25)	38
	36	CSM(25)	CSM(25)	39
	37	CSM(25)	CSM(25)	40
	38	CSM(25)	CSM(25)	41
	39	CSM(25)	CSM(25)	42
	40	CSM(25)	CSM(25)	43
	41	CSM(25)	CSM(25)	44
	42	CSM(25)	CSM(25)	45
	43	CSM(25)	CSM(25)	46
	44	CSM(25)	CSM(25)	47
	45	CSM(25)	CSM(25)	48
	46	CSM(25)	CSM(25)	49
	47	CSM(25)	CSM(25)	50
	48	CSM(25)	CSM(25)	51
	49	CSM(25)	CSM(25)	52
	50	CSM(25)	CSM(25)	53
	51	CSM(25)	CSM(25)	54
	52	CSM(25)	CSM(25)	55
	53	CSM(25)	CSM(25)	56
	54	CSM(25)	CSM(25)	57
	55	CSM(25)	CSM(25)	58
	56	CSM(25)	CSM(25)	59
	57	CSM(25)	CSM(25)	60
	58	CSM(25)	CSM(25)	61
	59	CSM(25)	CSM(25)	62
	60	CSM(25)	CSM(25)	63
	61	CSM(25)	CSM(25)	64
	62	CSM(25)	CSM(25)	65
	63	CSM(25)	CSM(25)	66
	64	CSM(25)	CSM(25)	67
	65	CSM(25)	CSM(25)	68
	66	CSM(25)	CSM(25)	69
	67	CSM(25)	CSM(25)	70
	68	CSM(25)	CSM(25)	71
	69	CSM(25)	CSM(25)	72
	70	CSM(25)	CSM(25)	73
	71	CSM(25)	CSM(25)	74
	72	CSM(25)	CSM(25)	75
	73	CSM(25)	CSM(25)	76
	74	CSM(25)	CSM(25)	77
	75	CSM(25)	CSM(25)	78
	76	CSM(25)	CSM(25)	79
	77	CSM(25)	CSM(25)	80
	78	CSM(25)	CSM(25)	81
	79	CSM(25)	CSM(25)	82
	80	CSM(25)	CSM(25)	83
	81	CSM(25)	CSM(25)	84
	82	CSM(25)	CSM(25)	85
	83	CSM(25)	CSM(25)	86
	84	CSM(25)	CSM(25)	87
	85	CSM(25)	CSM(25)	88
	86	CSM(25)	CSM(25)	89
	87	CSM(25)	CSM(25)	90
	88	CSM(25)	CSM(25)	91
	89	CSM(25)	CSM(25)	92
	90	CSM(25)	CSM(25)	93
	91	CSM(25)	CSM(25)	94
	92	CSM(25)	CSM(25)	95
	93	CSM(25)	CSM(25)	96
	94	CSM(25)	CSM(25)	97
	95	CSM(25)	CSM(25)	98
	96	CSM(25)	CSM(25)	99
	97	CSM(25)	CSM(25)	100

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20      8      ,NBOUND      ,NDS      ,NITEM      ,NMAX      ,NN      A      20
COMMON NUCASE      ,OMEGA(25)      ,P11      ,P(40)      ,P12      A      21
1      ,P2(40)      ,P15      ,PB(50)      ,PABAR      ,PBBAR      A      22
2      ,PBS      ,P15      ,PBI(40)      ,PBI(50)      ,PBI(50)      A      23
3      ,PMISH1      ,PMISHM      ,P1      ,PM(25)      ,PM(50)      A      24
4      ,PM12(40)      ,PM13      ,P1      ,PM(25)      ,PM(50)      A      25
5      ,PM1      ,P1      ,PSTREM      ,PM(50)      ,PM(50)      A      26
6      ,U11      ,Q(40)      ,QXTR1      ,QXTR2      A      27
7      ,G(50)      ,R11      ,R(40)      ,R(50)      A      28
8      ,RBS      ,R13      ,RBAR(40)      ,R14      ,R2(40)      A      29
9      ,RCUN      ,R15      ,RE(40)      ,RESM      ,R16      A      30
10     ,RMU(40)      ,R17      ,RMU2(40)      ,RMS(25)      ,RMSEN      A      31
11     ,RMSMUM      ,RHABAR      ,RMU      ,RMS      A      32
12     ,RSTHEM      ,RV      ,R19      ,RM(100)      ,S      A      33
13     ,S(50)      ,S(50)      ,SC(25)      ,SM(50)      A      34
14     ,S13      ,S(40)      ,T11      ,T(40)      ,T12      A      35
15     ,T2(40)      ,TABAR      ,TUBAR      ,T(25)      ,T14      A      36
16     ,TAA(40)      ,TXTR1      ,TXTR2      ,TUL      ,TSM      A      37
17     ,TSM1      ,TSTHEM      ,Tm(50)      ,Tm3      A      38
18     ,TS      ,U11      ,U(40)      ,U12      ,U2(40)      A      39
40     COMMON UXTR1      ,UXTR2      ,U14      ,UBAR(40)      ,USH      A      40
1      ,USH1      ,USTREM      ,U14      ,U(50)      ,US      A      41
2      ,X11      ,X(40)      ,X12      ,XBAR(25)      ,XBAR(25)      A      42
3      ,XB(50)      ,XBS      ,XSTHEM      ,X15      A      43
4      ,XS(25,40)      ,XSH      ,Y(40)      ,YM(50)      A      44
5      ,ZMAX(40)      ,ZM      ,Y11      ,Y12      A      45
6      ,Y2(40)      ,YABAR      ,YBAM      ,ZA(25)      ,Z11(25)      A      46
7      ,ZJ(25,40)      ,ZM      ,Z2X      ,Z2M      A      47
8      ,FX(2,40)      ,FX(2,40)      ,FPHI(2,40)      ,FP(2,40)      ,FT(2,40)      A      48
9      ,FU(2,40)      ,INDL(2,40)      ,INDH(2,40)      ,FC(2,40,25)      ,CANOI      A      49
50     ,H      ,H      A      50
COMMON/ALL/ TALL
COMMON/OUT/ REJ1,THETA1,CF1,UF1,CFCCF1,M12,U1SM,THETAC(2),DELT1,
DELTA2,CF2,UMALL,TAUMAL,ETA(140),TETC(140),UBUFI(140),YDFI(140)
DIMENSION I(140),J(140),K(140)
55     REAL KAT      ,KAT5(25)      ,KAT2      ,KN      ,MUOT(40)      ,
      ,MA(40)      ,MU      ,MUS(25)      ,MUO(25)      ,MU2      ,
      ,MA(25)      ,MA2      ,MASH      ,MASH1      ,MASHM      ,
      ,MAP      ,
      ,REAL MULHEF      ,
      ,IF (LL) 90,90,91      ,
90     UMALL=0.0      ,
      ,TAUMAL=0.0      ,
      ,UO TU 500      ,
91     CONTINUE      ,
      ,MULHEF=91.0-UB      ,
      ,TET2(KMAX)      ,
      ,UEMU2(KMAX)      ,
      ,MUOT=MMU2(KMAX)      ,
      ,PERP2(KMAX)      ,
      ,XEX2(KMAX)      ,
      ,PAUM2(KMAX)      ,
      ,SEXS(KMAX)      ,
      ,ANE=1.2      ,
      ,AMUE=MUFRETSUNT(TE/1034.)      ,
      ,AMUE=MUFRETSUNT(TALL/1034.)      ,
      ,REX1=UBASEMUE/AMUE      ,
      ,THETA1=.036*SE/(REX1**2)      ,
      ,DELT1=10.286*THETA1      ,
      ,C....CALC SKIN FRICTION X-S EU., FLAT PLATE, ZERO PRES GRAD      ,
      ,RETH1=RETHETA1*AMUE      ,
      ,CF1=1./17.0*(ALU610*(RETH1)**2+25.11*ALU610*(RETH1)**0.012)      ,
      ,C....CALC FRICTION VELOCITY      ,
      ,KCR2=1      ,
      ,C.....MEPLAGE ZHAK(1) WITH CONST MOLECULAR MGT FUN NON      ,
      ,NMU=TE*MMOL/TALL      ,
      ,UF1=(.5*NMOL*UE*CF1/UMALL)**.5      ,
      ,C....GENERAL TABLE OF TET VS U/UF      ,
      ,CONSTRAUE=4.973*U=UM      ,
      ,TET=TE*ALU61      ,
      ,ACR=(TET/UF-1.)      ,
      ,CCT=ALL/TE      ,
      ,RETH1=TE/UC      ,
      ,C....FIND RESULTS FOR UETAD=1      ,
      ,T1=50      ,
      ,T2=140      ,
      ,T3=1101      ,
      ,DO 100 J=1,11      ,
      ,TIA(J)=J/(11*10.)      ,
      ,DO 101 J=13,12      ,
100     TIA(J)=TIA(11)*(J-11)/100.      ,
      ,DETA1=ETA(1)      ,
      ,DETA2=ETA(13)-ETA(11)      ,
      ,DO 102 J=1,12      ,
102     TETC(J)=1./((ACR*ETA(J)+ETA(J)+8*(ETA(J)**ANE)+CL))**.5      ,
105     C....CALC (MUS CF)/M(CP CP      ,
      ,SUM1=0      ,
      ,DO 110 J=1,11      ,
110     SUM1=SUM1+TETC(J)*DETA1      ,
      ,DO 111 J=13,12      ,
111     SUM1=SUM1+TETC(J)*DETA2      ,
      ,CF1=SUM1/SUM1      ,
      ,CF1=CF1*CF1      ,
      ,C....FALL MEAL TRANSFER AND SHEAR STRESS      ,
      ,STAR=35*CF1      ,

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115 UNALL=STANARDUE+UE+CP*(110TE-TALL)
    TAUNAL=,SANKHUE+UE+CP*
C....RETURN POINT FROM FIRST PASS
    IF (IFLAG) 112,500,112
120 C....CALC U INCOMP/U FRICT INCOMP
    112 UBUF1(1)=SQRT(2./CFC)*TEIC(1)*DELTA1
    DO 120 J=2,11
    120 UBUF1(J)=UBUF1(J-1)+SQRT(2./CFC)*TEIC(J)*DELTA1
    DO 121 J=13,12
    121 UBUF1(J)=UBUF1(J-1)+SQRT(2./CFC)*TEIC(J)*DELTA2
125 C....CALL COMRESPONDING Y/DELTA, INCOMP
    FACT=1./EXP((UBUF1(12)-5.05)/2.5))
    DO 132 J=1,12
    IF (UBUF1(J).LT.5.0) GO TO 130
    IF (UBUF1(J).LT.13.96) GO TO 131
    YDEL1(J)=EXP((UBUF1(J)-5.05)/2.5)*FACT
    GO TO 132
    130 YDEL1(J)=UBUF1(J)*FACT
    GO TO 132
    131 YDEL1(J)=EXP((UBUF1(J)+3.05)/5.0)*FACT
135 132 CONTINUE
C....NOW HAVE Y/DELTA INCOMP = Y/DELTA COMP VS U/UE & TE/T*COMP
C....CALL THEIA/DELTA & DISP/DELTA BY TRAPEZOIDAL INTEGRATION
C....IF U/UE(1)=U/UE; (1-U/UE) & Y/DELTA
140 SUMD2=0.0
    SUMD1=0.0
    I=12-1
    DO 140 J=1,12
    D1INT(J)=1.-EIA(J)
    D2INT(J)=D1INT(J)+EIA(J)
    DO 141 J=1,12
    141 SUMD1=SUMD1+(YDEL1(J+1)-YDEL1(J))*S*(D1INT(J)+D1INT(J+1))
    SUMD2=SUMD2+(YDEL1(J+1)-YDEL1(J))*S*(D2INT(J)+D2INT(J+1))
    M12=SUMD1/SUMD2
150 C....CALL IMETAL FROM MOMENTUM INTEGRAL
    IF (LL.EQ.0) GO TO 142
    IMETAC(1)=IMETAL(2)
    GO TO 143
    142 IMETAC(1)=1
    143 DELX=SQRT((X2(KMAX)-X(KMAX))**2+(R2(KMAX)-R(KMAX))**2)
    UOXX=(UE-U(KMAX))/DELX
    UOXX=(HMOU-KM(KMAX))/DELX
    UOXX=(HMOU-KM(KMAX))/DELX
    AD=(M12+2.)/UE+DUDX+DHIJDX/RHDE+ORDX/RAD
    IMETAC(2)=(LFC/2.+DELX*IMETAL(1))/(1.+AD*DELX)
    DELTAC=IMETAC(2)/SUMD2
    DISP=SUMD1+DELTAC
    DO 150 J=1,12
    150 TEIC(J)=1./TEIC(J)*TEIC(J)
165 500 RETURN
    END
1 SUBROUTINE STABLE
C THIS SUBROUTINE DETERMINES STABLE STEPPING DISTANCE AND PUNCHES
C OUTPUT DATA WHEN CALLED FOR
5 C
    C(1)=A(25,1) ,AA(40) ,ALFA(25,25),ALPHAN ,ALPHAP
    1 ,ATUL ,METAP ,MMIX ,C11(25) ,C(25,40)
    2 ,C12(25) ,C2(25,40) ,CAHAK(25) ,CUHAN(25) ,CP
    3 ,CPS(25) ,CPSH ,CSH(25) ,CSH1(25) ,CSTHEM(25)
    4 ,CSTIR ,C21M(25,25),DEFF(25) ,DELTA
    10 5 ,U11 ,DELS(40) ,DELS ,DELSU ,OLS(40)
    6 ,U1M(25,25) ,U12 ,DELY(40) ,U13 ,UPDT(40)
    7 ,U14 ,UPHIDS(40) ,UPCIN ,
    8 ,EPSLCH ,EXTRA(50) ,FSTEP ,FMAX ,GMAX
    15 9 ,M11 ,M(40) ,MH ,MJ ,HPM(25)
    4 ,M2PM(25) ,M3PM(25) ,ICONS ,ICOUNT ,IDENI(25)
    1 ,IEHON ,EXTRA(50) ,IFLAG ,IKIND
    2 ,IFRIC ,ISMCN ,ITYPE ,IPD
    3 ,IUFF ,K ,KAY ,KAY2
    20 4 ,KLU ,KMAX ,LL ,LPLANE ,MA
    5 ,KUP ,K ,KMA ,MU0 ,MU
    6 ,KASH ,KDU1 ,KMA ,KPP ,KASH
    7 ,K02 ,KUS ,K ,KAY ,KAY2
    25 8 ,KMH(40) ,KLS ,K11 ,P11 ,P(40)
    1 ,KULASH ,KLEGA(25) ,P(50) ,PABAN ,PBBAN
    2 ,PHS ,P15 ,PHI(40) ,PHIM(50) ,PIR
    3 ,PHISH ,PHSTRM ,PI ,PHIM(50) ,PSH
    4 ,PHI2(40) ,PHIHS ,PI ,PHI(25) ,PSH
    30 5 ,PSM ,PS1 ,PSTHEM ,UATW2
    6 ,U11 ,U(40) ,UATR1 ,UATW2
    7 ,UR(50) ,K11 ,K(40) ,R14 ,R2(40)
    8 ,RBS ,R13 ,RBAR(40) ,R14 ,R2(40)
    9 ,RCUN ,R15 ,RE(40) ,RESH ,R16
    4 ,RH0(40) ,R17 ,RH02(40) ,RHSE(25) ,RHSEF
    35 1 ,RSHM ,KMAHAK ,RHOBAN ,RU ,RSM
    2 ,RSTHEM ,K ,K19 ,R19 ,S
    3 ,S13 ,SH(50) ,SC(25) ,S(50)
    4 ,S13 ,SX(40) ,I11 ,I(40)
    5 ,IP(40) ,TABAR ,TBBAR ,I0(25) ,I14
    40 6 ,TAA(40) ,TATW1 ,TATR2 ,TUL ,TSM
    7 ,TSM ,TSTHEM ,T(50) ,TAS
    8 ,T3 ,U11 ,U(40) ,U12 ,U2(40)
    9 ,UATW1 ,UATR2 ,U14 ,UBAN(40) ,USH
    45 1 ,USH ,USTHEM ,U14 ,UBAN(40) ,USH
    2 ,U11 ,U(40) ,U12 ,U2(40) ,U15
    3 ,U(50) ,US ,UABAK(25) ,UABAK(25) ,U15
    4 ,U(25,40) ,USM ,ASTHEM ,U(50)
    5 ,U(40) ,Y11 ,Y(40) ,Y12
    6 ,T(40) ,YABAN ,YBAN ,Z(25) ,Z11(25)
    50 7 ,ZJ(25,40) ,ZNA ,S2X ,K2SH ,X2SH
    8 ,FX(2,40) ,FPH(2,40) ,FPH(2,40) ,F1(2,40)

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	9	,FU(2,40)	,INDL(2,40)	,INDR(2,40)	,FC(2,40,25),CARD1	A	49
	10	,H	,H			A	50
55	11	REAL KAY	,KAYS(25)	,KAY2	,KX	,MDDT(40)	STABLE 7
	12	,MA(40)	,PU	,MUS(25)	,MU0(25)	,MU2	STABLE 8
	13	,MA(25)	,MA2	,MASH	,MASH1	,MASH	STABLE 9
	14	,MAP					STABLE 10
		COMMON/PI1/AMDMV(8), AMQWP(8), ENEP1(8), HEP(40,8), V(40,8), n(40,8),				STABLE	11
		1V2(40,8), K2(40,8), TRDDY(8), TP(40,8), TP2(40,8), KP(8),				STABLE	12
60		2KMP(40,8), RMP2(40,8), ENEP2(8), HBL(8), UM(40)				STABLE	13
		3,VULP(8), WTP(8), HC(8), OENG(40,8), ICOND(40,8)				STABLE	14
		LC(MV(N/PI2/FFF,FFC,GAVA,UM1,PU,NPG,KMX,DNDSP(8)				STABLE	15
		CUBON/PI3/CL,CS,IFS,KHSS,MT,MKAN,SIG,EP,IPART,ININE,IN(40)				STABLE	16
		1,SUMP,SUMPV,SUMPE				STABLE	17
65		COMMON/IBUGS1/IBUGS8				STABLE	18
		DIMENSION INDEX2(120)				STABLE	19
		COMMON/INIL/XLX,IKI,HEX				STABLE	20
		COMMON/XYZ/XYZ,ZXYZ				STABLE	21
		DIMENSION CA(30)				STABLE	22
70		1XYZ=20				STABLE	23
		ZXYZ=0				STABLE	24
		IF (LL) 10,10,46				STABLE	25
		1AUE=ING				STABLE	26
75		10 IF X1MA(6)=0				STABLE	27
		1N=0				STABLE	28
		DO 20 I=1,NMAX				STABLE	29
		20 INDEX2(I)=1				STABLE	30
		NMAX=NMAX+1				STABLE	31
		GO 30 1=MAX,120				STABLE	32
80		30 IF INDEX2(I)=0				STABLE	33
		40 K=2				STABLE	34
		1=2				STABLE	35
		DELS=1.0E10				STABLE	36
		45 IF (EXTRA(5)=LPLANE) 50,50,90				STABLE	37
85		50 IF (INDEX2(1)) 75,90,90				STABLE	38
		75 I=I+1				STABLE	39
		GO TO 50				STABLE	40
		VISCOUS STABILITY CRITERION				STABLE	41
90		90 UEL1=DELY(K)*NE(K)/2.0				STABLE	42
		IF (IBUGS8,NE,0) WRITE (6,95) DEL1				STABLE	43
		95 FORMAT (6H DEL1=,E11.4)				STABLE	44
		INITIAL STABILITY CRITERION				STABLE	45
		IF (MA(K)-1.0=EPSLON) 100,100,600				STABLE	46
		100 WRITE (6,200) K				STABLE	47
95		200 FORMAT (25HIFLOW IS SUBSONIC IN TUBE,15)				STABLE	48
		CALL PUMP(A(1,1),N,1)				STABLE	49
		CALL EXIT				STABLE	50
		600 UEL2=5*DELY(K)*(MA(K)**2-1.0)**(.5)				STABLE	51
		500 FORMAT (6H UEL2=,E11.4)				STABLE	52
100		IF (IBUGS8,NE,0) WRITE (6,500) UEL2				STABLE	53
		IF (UEL1,EO,0.0) GO TO 590				STABLE	54
		DELS(K)=ALPHA/(1.0/DEL1+1.0/UEL2)				STABLE	55
		GO TO 595				STABLE	56
		590 DELS(K)=ALPHA*(UEL2				STABLE	57
105		595 CONTINUE				STABLE	58
		COMBINING SMALL TUBES				STABLE	59
		IF (LL) 608,602,602				STABLE	60
		602 FLELL				STABLE	61
		FPLANE=LPLANE				STABLE	62
110		FMULT=FPLANE				STABLE	63
		USTEP=STEP				STABLE	64
		IF (EXTRA(4),LT,1.E-5) GO TO 604				STABLE	65
		STEP=GSTEP=(USTEP-EXTRA(3))*(1,FMULT)/EXTRA(4)				STABLE	66
		GO TO 606				STABLE	67
115		604 STEP=GSTEP				STABLE	68
		606 L=K				STABLE	69
		CALL COMB(L)				STABLE	70
		STEP=GSTEP				STABLE	71
		IF (L=K) 610,604,620				STABLE	72
120		610 K=L				STABLE	73
		IF (EXTRA(5)=LPLANE) 630,630,90				STABLE	74
		620 K=L-1				STABLE	75
		I=I+1				STABLE	76
		IF (EXTRA(5)=LPLANE) 630,630,90				STABLE	77
125		630 J=1				STABLE	78
		640 J=J+1				STABLE	79
		IF (INDEX2(J)) 640,650,650				STABLE	80
		650 INDEX2(J)=1				STABLE	81
		EXTRA(K)=EXTRA(6)+1				STABLE	82
130		GO TO 50				STABLE	83
		ANALYSE LIMITATION				STABLE	84
		600 ULNAX(SIN(.5*(PM1(K)+PM1(K-1)))/(1.5*(M(K)+M(K-1)))*DELTA+(PHI(K)-				STABLE	85
		*PM1(K-1))/ELLY(K))*DELSS(K)				STABLE	86
		IF (IBUGS8,NE,0) WRITE (6,695) ULNAX,ATUL				STABLE	87
135		695 FORMAT (6H ULNAX=,E11.4,7H ATUL=,E11.4)				STABLE	88
		IF (ABS(ULNAX)-ATUL) MU0,600,700				STABLE	89
		700 DELSS(K)=DELSS(K)+ATUL/ABS(ULNAX)				STABLE	90
		600 IF (DELSS(K)-DELS) 900,1000,1000				STABLE	91
		900 DELSS(DELSS(K))				STABLE	92
140		EXTRA(7)=K				STABLE	93
		1000 IF (EXTRA(5)=LPLANE) 1025,1025,1050				STABLE	94
		1025 INDEX2(1)=K				STABLE	95
		EXTRA(K)=1				STABLE	96
		I=I+1				STABLE	97
145		1050 K=K+1				STABLE	98
		IF (K=NMAX) 65,65,1055				STABLE	99
		1055 IF (NMAX=32) 1057,1057,1060				STABLE	100

	REAL(5,400) IDEFAT(1),MUO(1),FO(1),OMEGA(1),PHI(1),SL(1),MA(1)	PUTIN	100
	PHI(1,400) IDEFAT(1),MUO(1),FO(1),OMEGA(1),PHI(1),SL(1),MA(1)	PUTIN	101
150	5502 PHI(1,400) IDEFAT(1),MUO(1),FO(1),OMEGA(1),PHI(1),SL(1),MA(1)	PUTIN	102
	DO 5501 J=1,400	PUTIN	103
	AM(J)=AM(1,JA)	PUTIN	104
	5501 CONTINUE	PUTIN	105
	A(1,1)=AM(5)+1,0E+6/MA(1)	PUTIN	106
	A(1,2)=AM(6)/MA(1)+1000,0	PUTIN	107
155	A(1,3)=AM(11)/MA(1)	PUTIN	108
	A(1,4)=AB(2)/12,0E+6/MA(1)	PUTIN	109
	A(1,5)=AM(3)/15,0E+6/MA(1)	PUTIN	110
	A(1,6)=AM(4)/10,0E+6/MA(1)	PUTIN	111
	LSI(1)=CSI(1)/A(1,1)	PUTIN	112
160	550 CONTINUE	PUTIN	113
	DO 560 J=1,400	PUTIN	114
	DO 560 I=1,400	PUTIN	115
	IF(A(1,1)) 330,351,330	PUTIN	116
	351 IF(A(1,1)) 330,340,330	PUTIN	117
165	330 IDEFAT=1	PUTIN	118
	560 CONTINUE	PUTIN	119
	400 FOMFAT(AQ,R),SELE,0,1E8,2)	PUTIN	120
	DELTA=0	PUTIN	121
	IF(1/AM,1E,0) GO TO 450	PUTIN	122
170	REAL(4,400) IF,1PH,1DUTY,PERMG	PUTIN	123
	PHI(1,400) IF,1PH,1DUTY,PERMG	PUTIN	124
	425 FOMFAT(12,5)	PUTIN	125
	IVISC=1	PUTIN	126
	DO 11 450	PUTIN	127
175	450 DO 450 I=1,400	PUTIN	128
	IF(MUO(1)) 500,600,500	PUTIN	129
	500 IVISC=1	PUTIN	130
	600 CONTINUE	PUTIN	131
	IF(1/AM,1E,0,AK,1PANT,HE,0) IVISC=0	PUTIN	132
180	450 PHIT(0,0)=20	PUTIN	133
	MUPT(0,0)=20	PUTIN	134
	C INNEW STREAMLINE POSITION	PUTIN	135
	1100 A=1	PUTIN	136
	IF(1PH,FO,C) REAL(5,200) X(K),H(K),PHI(K)	PUTIN	137
185	PHIT(0,0)=20	PUTIN	138
	IF(1PH,AL,0) REAL(9,200) X(K),H(K),PHI(K)	PUTIN	139
	PHIT(0,0)=20 K,X(K),H(K),PHI(K)	PUTIN	140
	FORX(K)	PUTIN	141
	X(K)=0,0	PUTIN	142
190	C INNEW STREAMLINE POSITION AND STREAMTUBE PROPER	PUTIN	143
	452	PUTIN	144
	1200 CONTINUE	PUTIN	145
	IF(1PH,1E,0) REAL(5,200) X(K),H(K),PHI(K),P(K),I(K),U(K)	PUTIN	146
	IF(1PH,NE,0) REAL(9,200) X(K),H(K),PHI(K),P(K),I(K),U(K)	PUTIN	147
195	PHIT(0,0)=20 K,X(K),H(K),PHI(K),P(K),I(K),U(K)	PUTIN	148
	H=SLKT((X(K)-X(K-1))**2+(H(K)-H(K-1))**2)	PUTIN	149
	IF(ABS(PHI(K)-PHI(K-1))=1,0E+06) 1300,1300,1400	PUTIN	150
	1300 DELTY(K)=H	PUTIN	151
	GO TO 1500	PUTIN	152
200	1400 DELTY(K)=PHI(K)-PHI(K-1))/H/(2.0+SIN(.5*(PHI(K)-PHI(K-1))))	PUTIN	153
	1500 AA(K)=P(K)-H(K-1))/DELTA/DELTY(K)	PUTIN	154
	X(K)=X(K-1)+DELTY(K)	PUTIN	155
	C SIN(AM) (H=PHI(1) (SPCETS PASS INALITION)	PUTIN	156
	IF(1PH,1E,0) REAL(5,1000) C(1,K),I=1,400	PUTIN	157
205	IF(1PH,NE,0) REAL(9,1000) C(1,K),I=1,400	PUTIN	158
	1000 FOMFAT(10,3)	PUTIN	159
	C SLIP CALCULATIONS	PUTIN	160
	1400 Z=0,0	PUTIN	161
	DO 2000 I=1,400	PUTIN	162
210	2000 Z=Z+PHI(1,1)/PHI(1)	PUTIN	163
	Z=Z+U/Z	PUTIN	164
	Z=Z+U/Z	PUTIN	165
	DO 2100 I=1,400	PUTIN	166
	2100 AS(1,K)=C(1,K)/Z	PUTIN	167
215	2000 PHIT(K)=Z*PHI(K)/PHI(1)	PUTIN	168
	PHIT(K)=PHI(K)/PHI(1)	PUTIN	169
	PHIT(K)=PHI(K)/PHI(1)	PUTIN	170
	PHIT(K)=PHI(K)/PHI(1)	PUTIN	171
	PHIT(K)=PHI(K)/PHI(1)	PUTIN	172
220	IF 2000 I=1,400	PUTIN	173
	IF 2000 I=1,400	PUTIN	174
	IF(1,1)Z=0,0	PUTIN	175
	2000 CONTINUE	PUTIN	176
	2000 CONTINUE	PUTIN	177
225	IF(IVISC) 2400,2402,2406	PUTIN	178
	2402 IF 2400 I=1,400	PUTIN	179
	UPS(1)=0	PUTIN	180
	IF 2400 I=1,400	PUTIN	181
	2400 UPS(1)=UPS(1)+C(1,1)/PHI(1)	PUTIN	182
230	GO TO 2400	PUTIN	183
	2400 INLET	PUTIN	184
	ADL=0	PUTIN	185
	(ALL INLETSP(1,K),P(K),K,PUL1,1810)	PUTIN	186
	2400 (P=0,0	PUTIN	187
235	IF 2400 I=1,400	PUTIN	188
	2401 INLETSP(1,K)=C(1,K)	PUTIN	189
	IF (P=0) 2430,2430,2420	PUTIN	190
	2420 INLETSP(1,K)=C(1,K)	PUTIN	191
	2430 IF 2450 I=1,400	PUTIN	192
240	2430 INLETSP(1,K)=C(1,K)	PUTIN	193
	IF 2450 I=1,400	PUTIN	194

```

IF (JH1(J),J)XN1(J)/MU=FMAX) 2450,2450,2440
2440 FMAX=JH1(J)XN1(J)/MU
2450 CUNTINUE
HE(N)=HND(N)+U(N)*DELT(N)/(MU*FMAX)
NA(N)=U(N)+SUH1((CP+ZM-HCON)/(CP+T(N)*KU))
GAMMA(CP/(CP+H(N)/ZM))
MU=4.0*GAMA/(4.0+GAMA-5.0)
H(N)=0
250 DO 2500 I=1,MUS
DO 2500 J=1,NH
2500 H(N)=H(N)+A(1,J)*C(1,K)+I(N)*A(J=2)
IF (K=FMAX) 2600,2700,2700
2600 K=K+1
DO TO 1200
2700 TORY(KMAX)
H1=H(KMAX)
C CALL CONDITIONS
IF (I=1,NE,0) GO TO 2801
K=11(6,621)
DO 2800 I=1,M+1
HEAD (5,290) XN(1),HN(1),PHIN(1),PM(1),SA(1)
2800 NH11(6,620) L=K(1),HN(1),PHIN(1),PM(1),SA(1)
GO TO 3700
285 HEAD (4,200) XN(1),HN(1),XN(2),RA(2)
3700 S=ASU,0
H1=H(ASU,0)
K=1
PHIN=PHIN(M)
PHIN=PHIN(N)
IF (I=PHIN) 3900,3900,3900
3900 PHIN(N)=PHIN(N)
3900 K=K+1
PHIN=PHIN(N)
IF (I=PHIN) 4300,4000,4300
4000 IF (K=FMAX) 4100,4200,4200
4100 PHIN(N)=XN1(A((K+1)-HN(N-1))/(XN(N-1)-XN(N-1)))
GO TO 4300
4200 PHIN(N)=XN1(A((K+1)-HN(N-1))/(XN(N-1)-XN(N-1)))
PHIN(N)=PHIN(N)-PHIN(N-1)
4300 IS=IS(N)
IF (IS= ) 4800,4200,4800
4800 FMAX(SIN(PHIN(N))-SIN(PHIN(N-1)))/(XN(N)-XN(N-1))
IF (ABS(FMAX/(XN(N)-XN(N-1)))>1,E=06) 4600,4600,4500
285 ILSFAX=PHIN(N)-PHIN(N-1)/FMAX

C
DO TO 4700
4600 ULSA=SUH1((XN(N)-XN(N-1))*2+(HN(N)-HN(N-1))*2)
4700 S=ASU+ULSA
S(N)=SAA
290 IF (K=FMAX) 3900,4900,4900
4900 IF (I=1,NE,0) GO TO 4901
NH11(6,623)
DO 5000 I=1,M+1
HEAD (5,200) XN(1),HN(1),PHIN(1),PM(1),SA(1)
295 NH11(6,620) XN(1),HN(1),PHIN(1),PM(1),SA(1)
GO TO 5700
4901 HEAD (4,200) XN(1),HN(1),XN(2),HB(2)
5700 S=ASU,0
C CALL CONDITIONS
K=1
PHIN=PHIN(N)
IF (I=PHIN) 5900,5900,5900
5900 PHIN(N)=PHIN(N)
5900 K=K+1
PHIN=PHIN(N)
IF (I=PHIN) 6300,6000,6300
6000 IF (K=FMAX) 6100,6200,6200
6100 PHIN(N)=XN1(A((K+1)-HN(N-1))/(XN(N-1)-XN(N-1)))
GO TO 6300
6200 PHIN(N)=XN1(A((K+1)-HN(N-1))/(XN(N-1)-XN(N-1)))
PHIN(N)=PHIN(N)-PHIN(N-1)
6300 IS=IS(N)
IF (IS= ) 6750,6400,6750
6400 FMAX(SIN(PHIN(N))-SIN(PHIN(N-1)))/(XN(N)-XN(N-1))
IF (ABS(FMAX/(XN(N)-XN(N-1)))>1,E=06) 6600,6600,6500
315 6500 ULSFAX=PHIN(N)-PHIN(N-1)/FMAX
GO TO 6700
6600 ULSA=SUH1((XN(N)-XN(N-1))*2+(HN(N)-HN(N-1))*2)
6700 S=ASU+ULSA
S(N)=SAA
320 IF (K=FMAX) 5900,6800,6800
6800 IF (I=1,NE,0) 6900,6850,6850
C CALL CONDITIONS
NH11(6,623) NH11(6,623) NH11(6,623) NH11(6,623)
NH11(6,623) NH11(6,623) NH11(6,623) NH11(6,623)
325 IF (I=1,NE,0) HEAD (5,200) XN,NSH,PSI,PSH,ISH,USH
IF (I=1,NE,0) HEAD (5,100) (CSH(1),I=1,MUS)
IF (I=1,NE,0) HEAD (4,200) XN,NSH,PSI,PSH,ISH,USH
NH11(6,624) XN,NSH,PSI,PSH,ISH,USH
IF (I=1,NE,0) HEAD (4,100) (CSH(1),I=1,MUS)
330 NH11(6,623)
NH11(6,623)
C CALL FLUX PROPERTIES AT FIRST SHOCK POINT
NH11(5,200) XSTREH,NSSTREH,PHSTREH,PSSTREH,ISSTREH,USSTREH
NH11(6,625) XSTREH,NSSTREH,PHSTREH,PSSTREH,ISSTREH,USSTREH
NH11(5,200) (CSH(1),I=1,MUS)
NH11(5,200) (CSH(1),I=1,MUS)
335 NH11(5,200) (CSH(1),I=1,MUS)
NH11(5,200) (CSH(1),I=1,MUS)
NH11(5,200) (CSH(1),I=1,MUS)
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NH11(5,200) (CSH(1),I=1,MUS)
NH11(5,200) (CSH(1),I=1,MUS)
NH11(5,200) (CSH(1),I=1,MUS)
NH11(5,200) (CSH(1),I=1,MUS)

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IF (DIM(I,J)XNH(I)/MU=FMAX) 2450,2450,2440
2440 FMAX=PH(I,J)XNH(I)/MU
2450 CONTINUE
NE(N)=HNO(N)+DELY(N)/(MU+FMAX)
NA(N)=NU(N)+SURT((CP+ZMH=HCON)/(CP+T(K)+NU))
GAMMA=CP/(CP+HCON/ZMH)
HUE 4.0+GAMMA/(4.0+HUE=5.0)
H(N)=0
250 DO 2500 I=1,NUS
   DO 2500 J=1,NH
2500 H(K)=H(N)+A(I,J)+C(I,K)+I(N)+J=2
   IF (H=FMAX) 2600,2700,2700
2600 H=H+1
   GO TO 1200
2700 YOUT(KMAX)
   H=H+1(KMAX)
C CALL CONDITIONS
IF (INT,NE,0) GO TO 2801
F=11E(10,621)
DO 2800 I=1,N+1
   READ(5,290) XN(I),HN(I),PHIN(I),PN(I),SN(I)
2800 WRITE(6,620) I,XN(I),HN(I),PHIN(I),PN(I),SN(I)
   GO TO 3700
285 READ(9,200) XN(1),HN(1),XN(2),HN(2)
3700 S=ASIN(0)
   H=H+ASIN(0)
   H=1
   S=SM=PM(N)
   PHIN=PHIN(N)
   IF (PHIN) 3900,3800,3900
3800 PHIN(N)=PHIN(N)
3900 H=H+1
   PHIN=PHIN(N)
   IF (PHIN) 4300,4000,4300
4000 IF (H=FMAX) 4100,4200,4200
4100 PHIN(N)=XN(I)+A(I,N+1)=HN(N+1)/(XN(N+1)+HN(N+1))
   GO TO 4300
4200 PHIN(N)=XN(I)+A(I,N+1)=HN(N+1)/(XN(N+1)+HN(N+1))
   PHIN(N)=PHIN(N)=PHIN(N+1)
280 4300 ISM=SN(N)
   IF (ISM) 4600,4400,4600
4400 F=ASIN(SIN(PHIN(N))=SIN(PHIN(N+1)))/(XN(N)=XN(N+1))
   IF (ANS(F)/F=1) XN(N+1)=1, F=0 4600,4600,4500
285 4500 L=LS=PHIN(N)=PHIN(N+1)/F
   GO TO 4700
4600 U=LS=ASIN((XN(N)=XN(N+1))+2+(HN(N)=HN(N+1))+2)
4700 S=ASIN(1)LS=1
   SN(N)=SNA
290 4800 IF (H=FMAX) 5900,4900,4900
4900 IF (INT,NE,0) GO TO 4901
   WRITE(6,623)
   GO 5000 I=1,N+1
   READ(5,290) XN(I),HN(I),PHIN(I),PN(I),SN(I)
295 5000 WRITE(6,624) I,XN(I),HN(I),PHIN(I),PN(I),SN(I)
   GO TO 5700
4901 READ(9,200) XN(1),HN(1),XN(2),HN(2)
5700 S=ASIN(0)
C CALL CONDITIONS
   H=1
   PHIN=PHIN(N)
   IF (PHIN) 5900,5800,5900
5800 PHIN(N)=PHIN(N)
5900 H=H+1
   PHIN=PHIN(N)
   IF (PHIN) 6300,6000,6300
6000 IF (H=FMAX) 6100,6200,6200
6100 PHIN(N)=XN(I)+A(I,N+1)=HN(N+1)/(XN(N+1)+HN(N+1))
   GO TO 6300
6200 PHIN(N)=XN(I)+A(I,N+1)=HN(N+1)/(XN(N+1)+HN(N+1))
   PHIN(N)=PHIN(N)=PHIN(N+1)
310 6300 ISM=SN(N)
   IF (ISM) 6750,6400,6750
6400 F=ASIN(SIN(PHIN(N))=SIN(PHIN(N+1)))/(XN(N)=XN(N+1))
   IF (ANS(F)/F=1) XN(N+1)=1, F=0 6600,6600,6500
315 6500 U=LS=PHIN(N)=PHIN(N+1)/F
   GO TO 6700
6600 U=LS=ASIN((XN(N)=XN(N+1))+2+(HN(N)=HN(N+1))+2)
6700 S=ASIN(1)LS=1
   SN(N)=SNA
320 6750 IF (H=FMAX) 5900,6800,6800
6800 IF (INT,NE,0) 6900,6850,6850
C CALL FLUIDS MECHANICAL FLUIDS
6850 GO TO 700
325 IF (INT,NE,0) READ(5,200) XSM,NSH,PSI,PSH,ISH,USH
   IF (INT,NE,0) READ(5,1000) (CSH(I),I=1,NUS)
   IF (INT,NE,0) READ(9,200) XSM,NSH,PSI,PSH,ISH,USH
   WRITE(6,624) XSM,NSH,PSI,PSH,ISH,USH
   IF (I=1,N) READ(9,1000) (LSH(I),I=1,NUS)
330 6900 READ(6,625)
   PH=PSH
C CALL FLUID PROPERTIES AT FIRST SHOCK POINT
6901 READ(5,200) XSTRE,NSSTRE,PHSTRE,PSSTRE,ISSTRE,USSTRE
   WRITE(6,625) XSTRE,NSSTRE,PHSTRE,PSSTRE,ISSTRE,USSTRE
   READ(5,6900) (LSSTRE(I),I=1,NUS)
335 6901 IF (I=1,N) 7000,6910,6910
   (7000) 7000 7000 7000 7000 7000
   7000 7000 7000 7000 7000
   IF (I=1,N) READ(5,200) XSM,NSM,PSM,PSM,ISM,USH

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